DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

A4CE
Revision 52
Textron Aviation

206	U206B	TP206D
P206	U206C	TP206E
P206A	U206D	TU206A
P206B	U206E	TU206B
P206C	U206F	TU206C
P206D	U206G	TU206D
P206E	TP206A	TU206E
U206	TP206B	TU206F
U206A	TP206C	TU206G
206H	T206H	

October 20, 2026

WARNING: Use of alcohol-based fuels can cause serious performance degradation and fuel system component damage and is therefore prohibited on Cessna airplanes.

TYPE CERTIFICATE DATA SHEET NO. A4CE

This data sheet which is part of Type Certificate No. A4CE prescribes conditions and limitations under which the product for which the type certificate was issued meets the airworthiness requirements of the Federal Aviation Regulations.

Type Certificate Holder — Textron Aviation Inc.
One Cessna Boulevard
Wichita, Kansas 67215

Type Certificate Holder Record — Cessna Aircraft Company transferred to Textron Aviation Inc. on July 29, 2015

I. Model 206, Super Skywagon, 6 PCL-SM (Normal Category), Approved July 19, 1963

Engine — Continental IO-520-A

*Fuel — 100/130 minimum grade aviation gasoline

*Engine Limits — For all operations, 2700 r.p.m. (285 b. hp.)

Propeller and Propeller Limits — Landplanes

1. McCauley Propeller
   Hub: D2A34C58/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 10.3°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
      (1) Woodward D210452
      (2) McCauley C290D2/T5 or C290D3/T5

Page No.	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20
Rev. No.	52	21	48	47	21	48	21	48	47	47	23	21	48	47	48	21	47	24	47	47
Page No.	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40
Rev. No.	32	33	24	27	27	27	27	27	27	27	30	30	28	40	44	46	40	42	46	50
Page No.	41	42	43	44	45	46	47	48	49											
Rev. No.	50	47	50	50	51	51	52	51	52											

I. Model 206 (cont'd)

Propeller and Propeller Limits (cont'd)

2. McCauley constant speed propeller installation:
(w/incorporation of Cessna Service Kit SK206-49)
Hub: D2A37C230
Blades: 90REB-8
Diameter: Not over 82 in., not under 80 in.
Pitch settings at 30 in. station:
Low pitch: 12.8°
High pitch: 28.3°
a. Propeller spinner: Cessna 0752004-2
b. Hydraulic governor:
(1) Woodward D210452
(2) McCauley C290D2/T5 or C290D3/T5

Floatplane
3. McCauley Propeller
Hub: D2A34C58/90AT-2
Diameter: Not over 88 in., not under 84 in.
Pitch settings at 36 in. station:
Low pitch: 9.0°
High pitch: 25.8°
a. Propeller spinner: Cessna spinner 0752004
b. Hydraulic governor:
(1) Woodward D210452
(2) McCauley C290D2/T5 or C290D3/T5

*Airspeed Limits (CAS)

Never exceed	210 m.p.h. (182 knots)
Maximum structural cruising	170 m.p.h. (148 knots)
Maneuvering	138 m.p.h. (120 knots)
Flaps extended	110 m.p.h. (96 knots)

C.G. Range

Landplane
(+33.0) to (+47.4) at 2250 lb. or less
(+40.5) to (+47.4) at 3300 lb.
Floatplane (Edo 582-3430)
(+34.0) to (+47.4) at 2600 lb. or less
(+38.5) to (+47.4) at 3500 lb.
Straight line variation between points given

Empty Wt. C.G. Range: None

*Maximum Weight
Landplane: 3300 lb.
Floatplane: 3500 lb.

No. of Seats: Six (Two at +36, Two at +69, 2 at +100)

Maximum Baggage: Reference weight and balance data

Fuel Capacity: 65 gal. (63.4 gal. usable); two 32.5 gal. tanks in wings at +48

Oil Capacity: 12 qt. at -19.4 (6 qt. usable)
(See NOTE 1 for data on system fuel and oil)

I. Model 206 (cont'd)

Control Surface Movements	Wing flaps (Land)	Up	0°	Down	40° +1°, -2°
	(Sea)	Up	0°	Down	30° +1°, -2°
	Aileron	Up	21° ±2°	Down	14° 30' ±2°
	Elevator	Up	26° 30' ±1°	Down	18° ±1°
	Elevator tab	Up	15° +1°, -0°	Down	25° +1°, -0°
	Rudder (Land)	Right	27° 13' ±1°	Left	27° 13' ±1°
	Rudder (Sea)	Right	24° 57' ±1°	Left	24° 57' ±1°
	(measured perpendicular to rudder hinge line)				
	Rudder (Land)	Right	24° ±1°	Left	24° ±1°
	Rudder (Sea)	Right	22° ±1°	Left	22° ±1°
	(measured parallel to 0.0.W.L.)				

Serial Nos. Eligible: 206-0001 thru 206-0275

II. Model U206, Super Skywagon, 6 PCL-SM (Normal Category), Approved October 8, 1964

Engine: Continental IO-520-A

*Fuel: 100/130 minimum grade aviation gasoline

*Engine Limits: For all operations, 2700 r.p.m. (285 b. hp.)

Propeller and Propeller Limits:

Landplane

1. McCauley Propeller
   Hub: D2A34C58/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 10.3°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
2. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: D2A37C230
   Blades: 90REB-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 12.8°
   High pitch: 28.3°
   a. Propeller spinner: Cessna 0752004-2
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5

Floatplane

1. McCauley Propeller
   Hub: D2A34C58/90AT-2
   Diameter: Not over 88 in., not under 84 in.
   Pitch settings at 36 in. station:
   Low pitch: 9.0°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5

II. Model U206 (cont'd)

*Airspeed Limits (CAS)	Never exceed	210 m.p.h. (182 knots)
	Maximum structural cruising	170 m.p.h. (148 knots)
	Maneuvering	138 m.p.h. (120 knots)
	Flaps extended	110 m.p.h. (96 knots)

C.G. Range

Landplane
(+33.0) to (+47.4) at 2250 lb. or less
(+40.5) to (+47.4) at 3300 lb.
Straight line variation between points given
Floatplane (Edo 582-3430)
(+34.0) to (+47.4) at 2600 lb. or less
(+38.5) to (+47.4) at 3500 lb.
Straight line variation between points given

Empty Wt. C.G. Range: None

*Maximum Weight: Landplane 3300 lb.; Floatplane 3500 lb.

No. of Seats: Six (Two at +36, Two at +69, 2 at +100)

Maximum Baggage: Reference weight and balance data

Fuel Capacity: 65 gal. (63.4 gal. usable); two 32.5 gal. tanks in wings at +48

Oil Capacity: 12 qt. at -19.4 (6 qt. usable)
See Note 1 for data on system fuel and oil

Control Surface Movements

Wing flaps	(Land)	Up	0°	Down	40° +1°, -2°
	(Sea)	Up	0°	Down	30° +1°, -2°
Aileron		Up	21° ±2°	Down	14° 30' ±2°
Elevator		Up	26° 30' ±1°	Down	18° ±1°
Elevator tab		Up	15° +1°, -0°	Down	25° +1°, -0°
Rudder (Land)		Right	27° 13' ±1°	Left	27° 13' ±1°
Rudder (Sea)		Right	24° 57' ±1°	Left	24° 57' ±1°
(measured perpendicular to rudder hinge line)					
Rudder (Land)		Right	24° ±1°	Left	24° ±1°
Rudder (Sea)		Right	22° ±1°	Left	22° ±1°
(measured parallel to 0.0.W.L.)					

Serial Nos. Eligible: U206-0276 thru 206-0437

III. Model P206, 6 PCL-SM (Normal Category), Approved October 8, 1964

Engine: Continental IO-520-A

*Fuel: 100/130 minimum grade aviation gasoline

*Engine Limits: For all operations, 2700 r.p.m. (285 b. hp.)

III. Model P206 (cont'd)

Propeller and Propeller Limits

Landplane

1. McCauley Propeller
   Hub: E2A34C64/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 10.3°,
   High pitch: 25.8°
   a. Propeller spinner: Cessna 1250411-4
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
2. McCauley Propeller
   Hub: E2A34C73/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 10.3°,
   High pitch: 25.8°
   a. Propeller spinner: Cessna spinner 1250415
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
3. McCauley Propeller
   Hub: D3A32C88/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 13.8°
   High pitch: 28.1°
   a. Propeller spinner: Cessna spinner 1250419
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
4. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49):
   Hub: E2A37C233
   Blades: 90REB-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 12.8°
   High pitch: 28.3°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5

Floatplane

1. McCauley Propeller
   Hub: D2A34C58/90AT-2
   Diameter: Not over 88 in., not under 84 in.
   Pitch settings at 36 in. station:
   Low pitch: 9.0°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5

III. Model P206 (cont'd)

*Airspeed Limits (CAS)	Never exceed	210 mph. (182 knots)
	Maximum structural cruising	170 mph. (148 knots)
	Maneuvering	138 mph. (120 knots)
	Flaps extended	110 mph. (96 knots)
C.G. Range	Landplane	
	(+33.0) to (+47.4) at 2250 lb. or less	
	(+40.5) to (+47.4) at 3300 lb.	
	Straight line variation between points given	
	Floatplane (Edo 582-3430)	
	(+34.0) to (+47.4) at 2600 lb. or less	
	(+38.5) to (+47.4) at 3500 lb.	
	Straight line variation between points given	
Empty wt. C.G. Range	None	
*Maximum Weight	Landplane	3300 lb.
	Floatplane	3500 lb.
No. of Seats	Six (Two at +36, 2 at +69, Two at +100)	
Maximum Baggage	Reference weight and balance data	
Fuel Capacity	65 gal. (63.4 gal. usable); two 32.5 gal. tanks in wings at +48	
Oil Capacity	12 qt. at -19.4 (6 qt. usable)	
	See Note 1 for data on system fuel and oil	

Control Surface Movements					
Wing flaps	(Land)	Up	0°	Down	40° +1°, -2°
	(Sea)	Up	0°	Down	30° +1°, -2°
Aileron		Up	21° ±2°	Down	14° 30' ±2°
Elevator		Up	26° 30' ±1°	Down	18° ±1°
Elevator tab		Up	15° +1°, -0°	Down	25° +1°, -0°
Rudder (Land)		Right	27° 13' ±1°	Left	27° 13' ±1°
Rudder (Sea)		Right	24° 57' ±1°	Left	24° 57' ±1°
(measured perpendicular to rudder hinge line)					
Rudder (Land)		Right	24° ±1°	Left	24° ±1°
Rudder (Sea)		Right	22° ±1°	Left	22° ±1°
(measured parallel to 0.0.W.L.)					

Serial Nos. Eligible	P206-0002 thru P206-0160

IV. Model U206A, 6 PCL-SM (Normal Category), Approved September 24, 1965

Engine	Continental IO-520-A
*Fuel	100/130 minimum grade aviation gasoline
*Engine Limits	For all operations, 2700 r.p.m. (285 b. hp.)

IV. Model U206A (cont'd)

Propeller and Propeller Limits

Landplane

1. McCauley Propeller
   Hub: D2A34C58/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 10.3°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
2. McCauley Propeller
   Hub: D3A32C79/82NK-2
   Diameter: not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 11.3°
   High pitch: 27.6°
   a. Propeller spinner: Cessna 1250420
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
3. McCauley Propeller
   Hub: D3A32C90/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 13.8°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250420
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
4. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: D2A37C230 with 90REB-8 blades
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 12.8°
   High pitch: 28.3°
   a. Propeller spinner: Cessna 0752004-2
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5

Floatplane

1. McCauley Propeller
   Hub: D2A34C58/90AT-2
   Diameter: Not over 88 in., not under 84 in.
   Pitch settings at 36 in. station:
   Low pitch: 9.0°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
2. McCauley Propeller
   Hub: D3A32C90/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 13.8°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor: McCauley C290D2/T5 or C290D3/T5

IV. Model U206A (cont'd)

*Airspeed Limits (CAS)

Never exceed	210 mph. (182 knots)
Maximum structural cruising	170 mph. (148 knots)
Maneuvering	
3300 lb. skiplane	138 mph. (120 knots)
3500 lb. floatplane	138 mph. (120 knots)
3600 lb. landplane	144 mph. (125 knots)
Flaps extended	110 mph. (96 knots)

C.G. Range

Landplane
(+33.0) to (+47.4) at 2500 lb. or less
(+42.5) to (+47.4) at 3600 lb.
Straight line variation between points given
Floatplane (Edo 582-3430)
(+34.0) to (+47.4) at 2600 lb. or less
(+38.5) to (+47.4) at 3500 lb.
Straight line variation between points given
Skiplane (FluiDyne R210)
(+33.0) to (+47.4) at 2250 lb. or less
(+40.5) to (+47.4) at 3300 lb.
Straight line variation between points given

Empty wt. C.G. Range: None

*Maximum Weight

Landplane	3600 lb.
Floatplane	3500 lb.
Skiplane	3300 lb.

No. of Seats: 6 (2 at +36, 2 at +69, 2 at +100)

Maximum Baggage: Reference weight and balance data

Fuel Capacity: 65 gal. (63.0 gal. usable); two 32.5 gal. tanks in wings at +48

Oil Capacity: 12 qt. at -19.4 (6 qt. usable)
See Note 1 for data on system fuel and oil

Control Surface Movements

Wing flaps	(Land)	Up	0°	Down	40° +1°, -2°
	(Sea)	Up	0°	Down	30° +1°, -2°
Aileron		Up	21° ±2°	Down	14° 30' ±2°
Elevator		Up	26° 30' ±1°	Down	18° ±1°
Elevator tab		Up	15° +1°, -0°	Down	25° +1°, -0°
Rudder (Land)		Right	27° 13' ±1°	Left	27° 13' ±1°
Rudder (Sea)		Right	24° 57' ±1°	Left	24° 57' ±1°
(measured perpendicular to rudder hinge line)					
Rudder (Land)		Right	24° ±1°	Left	24° ±1°
Rudder (Sea)		Right	22° ±1°	Left	22° ±1°
(measured parallel to 0.0.W.L.)					

Serial Nos. Eligible: U206-0438 thru U206-0656

V. Model P206A, 6 PCL-SM (Normal Category), Approved September 24, 1965
Model P206B, 6 PCLM (Normal Category), Approved August 3, 1966

Engine: Continental IO-520-A

*Fuel: 100/130 minimum grade aviation gasoline

*Engine Limits: For all operations, 2700 rpm. (285 b.hp.)

Propeller and Propeller Limits:

Landplane (Model P206A only)

1. McCauley Propeller
   Hub: E2A34C73/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 10.3°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
2. McCauley Propeller
   Hub: D2A34C58/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 10.3°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
3. McCauley Propeller
   Hub: D3A32C77/82NK-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 11.3°
   High pitch: 27.6°
   a. Propeller spinner: Cessna 1250419-2
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
4. McCauley Propeller
   Hub: D3A32C79/82NK-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 11.3°
   High pitch: 27.6°
   a. Propeller spinner: Cessna 1250420
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
5. McCauley Propeller
   Hub: D3A32C88/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 13.8°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250419-2
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5

V. <u>Model P206A, Model P206B</u> (cont'd)

Propeller and Propeller Limits (cont'd)

6. McCauley Propeller
   Hub: D3A32C90/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 13.8°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250420
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
7. McCauley constant speed propeller installation with incorporation of Cessna Service Kit SK206-49)
   Hub: E2A37C233
   Blades: 90REB-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 12.8°
   High pitch: 28.3°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
8. McCauley constant speed propeller installation with incorporation of Cessna Service Kit SK206-49)
   Hub: D2A37C230
   Blades: 90REB-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 12.8°
   High pitch: 28.3°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5

<u>Floatplane</u>

1. McCauley Propeller
   Hub: D2A34C58/90AT-2
   Diameter: Not over 88 in., not under 84 in.
   Pitch settings at 36 in. station:
   Low pitch: 9.0°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5

<u>Landplane</u> (Model P206B only)

1. McCauley Propeller
   Hub: E2A34C73/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 10.3°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5

V. <u>Model P206A, Model P206B</u> (cont'd)

Propeller and Propeller Limits (cont'd)	2. McCauley Propeller Hub: D3A32C88/82NC-2 Diameter: Not over 80 in., not under 78 in. Pitch settings at 30 in. station: Low pitch: 13.8° High pitch: 28.1° a. Propeller spinner: Cessna 1250419-2 b. Hydraulic governor: (1) Woodward D210452 (2) McCauley C290D2/T5 or C290D3/T5 3. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49) Hub: E2A37C233 Blades: 90REB-8 Diameter: Not over 82 in., not under 80 in. Pitch settings at 30 in. station: Low pitch: 12.8° High pitch: 28.3° a. Propeller spinner: Cessna 1250415 b. Hydraulic governor: (1) Woodward D210452 (2) McCauley C290D2/T5 or C290D3/T5
*Airspeed Limits (CAS)	Never exceed 210 mph. (182 knots) Maximum structural cruising 170 mph. (148 knots) Maneuvering 3300 lb. skiplane 138 mph. (120 knots) 3500 lb. floatplane 138 mph. (120 knots) 3600 lb. landplane 144 mph. (125 knots) Flaps extended 110 mph. (96 knots)
C.G. Range	<u>Landplane</u> (+33.0) to (+47.4) at 2500 lb. or less (+42.5) to (+47.4) at 3600 lb. Straight line variation between points given <u>Floatplane</u> (Edo 582-3430) (+34.0) to (+47.4) at 2600 lb. or less (+38.5) to (+47.4) at 3500 lb. Straight line variation between points given <u>Skiplane</u> (FluiDyne R210) (+33.0) to (+47.4) at 2250 lb. or less (+40.5) to (+47.4) at 3300 lb. Straight line variation between points given
Empty wt. C.G. Range	None
*Maximum Weight	Landplane 3600 lb. Floatplane 3500 lb. (Model P206A only) Skiplane 3300 lb.
No. of Seats	Six (Two at +36, 2 at +69, Two at +100)
Maximum Baggage	Reference weight and balance data
Fuel Capacity	65 gal. (63.0 gal. usable); two 32.5 gal. tanks in wings at +48

V. Model P206A, Model P206B (cont'd)

Oil Capacity	12 qt. at -19.4 (6 qt. usable) See Note 1 for data on system fuel and oil				
Control Surface Movements	Wing flaps (Land)	Up	0°	Down	40° +1°, -2°
	(Sea)	Up	0°	Down	30° +1°, -2°
	Aileron	Up	21° ±2°	Down	14° 30' ±2°
	Elevator	Up	26° 30' ±1°	Down	18° ±1°
	Elevator tab	Up	15° +1°, -0°	Down	25° +1°, -0°
	Rudder (Land)	Right	27° 13' ±1°	Left	27° 13' ±1°
	Rudder (Sea)	Right	24° 57' ±1°	Left	24° 57' ±1°
	(measured perpendicular to rudder hinge line)				
	Rudder (Land)	Right	24° ±1°	Left	24° ±1°
	Rudder (Sea)	Right	22° ±1°	Left	22° ±1°
	(measured parallel to 0.0.W.L.)				
Serial Nos. Eligible	Model P206A: P206-0161 thru P206-0306 Model P206B: P206-0307 thru P206-0419				

VI. Model TU206A, 6 PCL-SM (Normal Category), Approved December 20, 1965
Model TU206B, 6 PCL-SM (Normal Category), Approved August 3, 1966

Engine: Continental TSIO-520-C

*Fuel: 100/130 minimum grade aviation gasoline

*Engine Limits: For all operations, 2700 rpm. (285 b.hp.)

Propeller and Propeller Limits:

Landplane

1. McCauley Propeller
   Hub: D2A34C78/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 11.8°
   High pitch: 32.0°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
2. McCauley Propeller
   Hub: D3A32C79/82NK-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 13.2°
   High pitch: 32.5°
   a. Propeller spinner: Cessna 1250420
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
3. McCauley Propeller
   Hub: D3A32C90/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.0°
   High pitch: 33.0°
   a. Propeller spinner: Cessna 1250420
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

VI. Model TU206A, Model TU206B (cont'd)

Propeller and Propeller Limits (cont'd)

4. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: D2A37C231
   Blades: 90REB-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.3°
   High pitch: 34.5°
   a. Propeller spinner: Cessna 0752004-2
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

Floatplane

1. McCauley Propeller
   Hub: D2A34C58/90AT-2
   Diameter: Not over 88 in., not under 86 in.
   Pitch settings at 36 in. station:
   Low pitch: 9.0°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
2. McCauley Propeller
   Hub: D3A32C90/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.0°
   High pitch: 33.0°
   a. Propeller spinner: Cessna spinner 1250909
   b. Hydraulic governor: McCauley C290D2/T2 or C290D4/T2

*Airspeed Limits (CAS)

Never exceed	210 m.p.h. (182 knots)
Maximum structural cruising	170 m.p.h. (148 knots)
Maneuvering	
3300 lb. skiplane	138 m.p.h. (120 knots)
3600 lb. floatplane	138 m.p.h. (120 knots)
3600 lb. landplane	144 m.p.h. (125 knots)
Flaps extended	110 m.p.h. (96 knots)

C.G. Range

Landplane
(+33.0) to (+47.4) at 2500 lb. or less
(+42.5) to (+47.4) at 3600 lb.
Straight line variation between points given
Floatplane (Edo 582-3430)
(+34.0) to (+47.4) at 2600 lb. or less
(+39.0) to (+47.4) at 3600 lb.
Straight line variation between points given
Skiplane (FluiDyne R210)
(+33.0) to (+47.4) at 2250 lb. or less
(+40.5) to (+47.4) at 3300 lb.
Straight line variation between points given

Empty Wt. C.G. Range

None

VI. Model TU206A, Model TU206B (cont'd)

*Maximum Weight	Landplane	3600 lb.			
	Floatplane	3600 lb.			
	Skiplane	3300 lb.			
No. of Seats	6 (2 at +36, 2 at +69, 2 at +100)				
Maximum Baggage	Reference weight and balance data				
Fuel Capacity	65 gal. (63.0 gal. usable); two 32.5 gal. tanks in wings at +48				
Oil Capacity	12 qt. at -19.4 (6 qt. usable) See NOTE 1 for data on system fuel and oil				
Control Surface Movements	Wing flaps (Land)	Up	0°	Down	40° +1°, -2°
	(Sea)	Up	0°	Down	30° +1°, -2°
	Aileron	Up	21° ±2°	Down	14° 30' ±2°
	Elevator	Up	26° 30' ±1°	Down	18° ±1°
	Elevator tab	Up	15° +1°, -0°	Down	25° +1°, -0°
	Rudder (Land)	Right	27° 13' ±1°	Left	27° 13' ±1°
	Rudder (Sea)	Right	24° 57' ±1°	Left	24° 57' ±1°
	(measured perpendicular to rudder hinge line)				
	Rudder (Land)	Right	24° ±1°	Left	24° ±1°
	Rudder (Sea)	Right	22° ±1°	Left	22° ±1°
	(measured parallel to 0.0.W.L.)				
Serial Nos. Eligible	Model TU206A: U206-0487 thru U206-0656 Model TU206B: U206-0657 thru U206-0914				

VII. Model TP206A, 6 PCL-SM (Normal Category), Approved December 20, 1965
Model TP206B, 6 PCLM (Normal Category), Approved August 3, 1966

Engine: Continental TSIO-520-C

*Fuel: 100/130 minimum grade aviation gasoline

*Engine Limits: For all operations, 2700 r.p.m. (285 b. hp.)

Propeller and Propeller Limits:

Landplane (Model TP206A only)

1. McCauley Propeller
   Hub: E2A34C70/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 11.8°
   High pitch: 32.0°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T4
2. McCauley Propeller
   Hub: D2A34C78/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 11.8°
   High pitch: 32.0°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

VII. Model TP206A, TP206B (cont'd)

Propeller and Propeller Limits (cont'd)

3. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: D3A32C77/82NK-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 13.2°
   High pitch: 32.5°
   a. Propeller spinner: Cessna 1250419
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
4. McCauley Propeller
   Hub: D3A32C79/82NK-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 13.2°
   High pitch: 32.5°
   a. Propeller spinner: Cessna 1250420
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
5. McCauley Propeller
   Hub: D3A32C88/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.0°
   High pitch: 33.0°
   a. Propeller spinner: Cessna 1250419
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
6. McCauley Propeller
   Hub: D3A32C90/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.0°
   High pitch: 33.0°
   a. Propeller spinner: Cessna 1250420
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
7. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: E2A37C234 with 90REC-8 blades
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.3°
   High pitch: 34.5°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

VII. Model TP206A, TP206B (cont'd)

Propeller and Propeller Limits (cont'd)

8. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: D2A37C231
   Blades: 90REB-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.3°
   High pitch: 34.5°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

Floatplane (Model TP206A only)

1. McCauley Propeller
   Hub: D2A34C58/90AT-2
   Diameter: Not over 88 in., not under 86 in.
   Pitch settings at 36 in. station:
   Low pitch: 9.0°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 0752004
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

Landplane (Model TP206B only)

1. McCauley Propeller
   Hub: E2A34C70/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 11.8°
   High pitch: 32.0°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
2. McCauley Propeller
   Hub: D3A32C88/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.0°
   High pitch: 33.0°
   a. Propeller spinner: Cessna 1250419
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
3. McCauley Propeller
   Hub: D3A32C77/82NK-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 13.2°
   High pitch: 32.5°
   a. Propeller spinner: Cessna 1250419
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

VII. Model TP206A, TP206B (cont'd)

Propeller and Propeller Limits (cont'd)

4. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: E2A37C234 with 90REC-8 blades
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.3°
   High pitch: 34.5°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

***Airspeed Limits (CAS)**

Never exceed	210 m.p.h. (182 knots)
Maximum structural cruising	170 m.p.h. (148 knots)
Maneuvering	
3300 lb. skiplane	138 m.p.h. (120 knots)
3600 lb. floatplane	138 m.p.h. (120 knots)
3600 lb. landplane	144 m.p.h. (125 knots)
Flaps extended	110 m.p.h. (96 knots)

C.G. Range

Landplane
(+33.0) to (+47.4) at 2500 lb. or less
(+42.5) to (+47.4) at 3600 lb.
Straight line variation between points given
Floatplane (Edo 582-3430)
(+34.0) to (+47.4) at 2600 lb. or less
(+39.0) to (+47.4) at 3600 lb.
Straight line variation between points given
Skiplane (FluiDyne R210)
(+33.0) to (+47.4) at 2250 lb. or less
(+40.5) to (+47.4) at 3300 lb.
Straight line variation between points given

Empty Wt. C.G. Range

None

***Maximum Weight**

Landplane	3600 lb.
Floatplane	3600 lb. (Model TP206A only)
Skiplane	3300 lb.

No. of Seats

Six (Two at +36, 2 at +69, Two at +100)

Maximum Baggage

Reference weight and balance data

Fuel Capacity

65 gal. (63.0 gal. usable) two 32.5 gal. tanks in wings at +48

Oil Capacity

12 qt. at -19.4 (6 qt. usable)
See NOTE 1 for data on system fuel and oil

Control Surface Movement

Wing flaps (Land)	Up	0°	Down	40° +1°, -2°
(Sea)	Up	0°	Down	30° +1°, -2°
Aileron	Up	21° ±2°	Down	14° 30' ±1°
Elevator	Up	26° 30' ±1°	Down	18° ±1°
Elevator tab	Up	15° +1°, -0°	Down	25° +1°, -0°
Rudder (Land)	Right	27° 13' ±1°	Left	27° 13' ±1°
Rudder (Sea)	Right	24° 57' ±1°	Left	24° 57' ±1°
(measured perpendicular to rudder hinge line)				
Rudder (Land)	Right	24° ±1°	Left	24° ±1°
Rudder (Sea)	Right	22° ±1°	Left	22° ±1°
(measured parallel to 0.0.W.L.)				

VII. Model TP206A, TP206B (cont'd)

Serial Nos. Eligible	Model TP206A:	P206-0191 thru P206-0306
	Model TP206B:	P206-0307 thru P206-0419

VIII. Model U206B, 6 PCL-SM (Normal Category), Approved August 3, 1966

Engine: Continental IO-520-F

*Fuel: 100/130 minimum grade aviation gasoline

*Engine Limits: Takeoff (5 min.) at 2850 r.p.m. (300 hp.)
For all other operations, 2700 r.p.m. (285 hp.)

Propeller and Propeller Limits:

Landplane

1. McCauley Propeller
   Hub: D2A34C58/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 9.5°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 1250909-3
   b. Hydraulic governor:
   (1) Woodward 210452
   (2) McCauley C290D2/T4 or C290D4/T4
2. McCauley Propeller
   Hub: D3A32C90/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 11.5°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250909-8
   b. Hydraulic governor:
   (1) Woodward 210462
   (2) McCauley C290D2/T4 or C290D4/T4
3. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: D2A37C230
   Blades: 90REB-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 12.0°
   High pitch: 28.3°
   a. Propeller spinner: Cessna 1250909-K200
   b. Hydraulic governor:
   (1) Woodward 210462
   (2) McCauley C290D2/T4 or C290D4/T4

Floatplane

1. McCauley Propeller
   Hub: D2A34C58/90AT-4
   Diameter: Not over 86 in., not under 84 in.
   Pitch settings at 36 in. station:
   Low pitch: 11.5°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250909-3
   b. Hydraulic governor:
   (1) Woodward 210462
   (2) McCauley C290D2/T4 or C290D4/T4

VIII. Model U206B (cont'd)

Propeller and Propeller Limits	2. McCauley Propeller Hub: D3A32C90/82NC-2 Diameter: Not over 80 in., not under 78 in. Pitch settings at 30 in. station: Low pitch: 11.5° High pitch: 28.1° a. Propeller spinner: Cessna 1250909 b. Hydraulic governor: McCauley C290D2/T4 or C290D4/T4
*Airspeed Limits (CAS)	Never exceed 210 mph. (182 knots) Maximum structural cruising 170 mph. (148 knots) Maneuvering 3300 lb. skiplane 138 mph. (120 knots) 3500 lb. floatplane 138 mph. (120 knots) 3600 lb. landplane 144 mph. (125 knots) Flaps extended 110 mph. (96 knots)
C.G. Range	Landplane (+33.0) to (+47.4) at 2500 lb. or less (+42.5) to (+47.4) at 3600 lb. Straight line variation between points given Floatplane (Edo 582-3430) (+34.0) to (+47.4) at 2600 lb. or less (+38.5) to (+47.4) at 3500 lb. Straight line variation between points given Skiplane (FluiDyne R210) (+33.0) to (+47.4) at 2250 lb. or less (+40.5) to (+47.4) at 3300 lb. Straight line variation between points given
Empty wt. C.G. Range	None
*Maximum Weight	Landplane 3600 lb. Floatplane 3500 lb. Skiplane 3300 lb.
No. of Seats	Six (Two at +36, 2 at +69, Two at +100)
Maximum Baggage	Reference weight and balance data
Fuel Capacity	65 gal. (two 32.5 gal. tanks in wings at +48; 63.0 gal. usable)
Oil Capacity	12 qt. at -19.4 (6 qt. usable) See Note 1 for data on system fuel and oil

Control Surface Movements

Wing flaps	(Land)	Up	0°	Down	40° +1°, -2°
	(Sea)	Up	0°	Down	30° +1°, -2°
Aileron		Up	21° ±2°	Down	14° 30' ±2°
Elevator		Up	26° 30' ±1°	Down	18° ±1°
Elevator tab		Up	15° +1°, -0°	Down	25° +1°, -0°
Rudder (Land)		Right	27° 13' ±1°	Left	27° 13' ±1°
Rudder (Sea)		Right	24° 57' ±1°	Left	24° 57' ±1°
(measured perpendicular to rudder hinge line)					
Rudder (Land)		Right	24° ±1°	Left	24° ±1°
Rudder (Sea)		Right	22° ±1°	Left	22° ±1°
(measured parallel to 0.0.W.L.)					

Serial Nos. Eligible	U206-0657 thru U206-0914

IX. Model P206C/TP206C, 6 PCLM (Normal Category), Approved July 20, 1967
Model P206D/TP206D, 6 PCLM (Normal Category), Approved September 18, 1968
Model P206E/TP206E, 6 PCLM (Normal Category), Approved July 28, 1969

Models P206C, P206D, P206E

Engine: Continental IO-520-A

*Fuel: 100/130 minimum grade aviation gasoline

*Engine Limits: For all operations, 2700 rpm. (285 b.hp.)

Propeller and Propeller Limits:

1. McCauley Propeller
   Hub: E2A34C73/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 10.3°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
2. McCauley Propeller
   Hub: D3A32C88/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 13.8°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250419-2
   b. Hydraulic governor:
   (1) Woodward D210452
   (2) McCauley C290D2/T5 or C290D3/T5
3. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: E2A37C233
   Blades: 90REB-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 12.8°
   High pitch: 28.3°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T5 or C290D3/T5

Models TP206C, TP206D, TP206E

Engine: Continental TSIO-520-C

*Fuel: 100/130 minimum grade aviation gasoline

*Engine Limits: For all operations, 2700 rpm. (285 b.hp.)

IX. Models P206C/TP206C, P206D/TP206D, P206E/TP206E (cont'd)

Models TP206C, TP206D, TP206E (cont'd)

Propeller and Propeller Limits

1. McCauley Propeller
   Hub: E2A34C70/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 11.8°
   High pitch: 32.0°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
2. McCauley Propeller
   Hub: D3A32C88/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.0°
   High pitch: 33.0°
   a. Propeller spinner: Cessna 1250419
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2
3. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: E2A37C234
   Blades: 90REB-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.3°
   High pitch: 34.5°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

Models P206C/TP206C, P206D/TP206D, P206E/TP206E

*Airspeed Limits (CAS)	Never exceed	210 mph. (182 knots)
	Maximum structural cruising	170 mph. (148 knots)
	Maneuvering	
	3300 lb. skiplane	138 mph. (120 knots)
	3600 lb. landplane	144 mph. (125 knots)
	Flaps extended	110 mph. (96 knots)

C.G. Range

Landplane
(+33.0) to (+49.7) at 2500 lb. or less
(+42.5) to (+49.7) at 3600 lb.
Straight line variation between points given
Skiplane (FluiDyne R210)
(+33.0) to (+47.4) at 2250 lb. or less
(+40.5) to (+47.4) at 3300 lb.
Straight line variation between points given

Empty wt. C.G. Range: None

IX. Models P206C/TP206C, P206D/TP206D, P206E/TP206E (cont'd)

Models P206C/TP206C, P206D/TP206D, P206E/TP206E (cont'd)

*Maximum Weight	Landplane 3600 lb. Skiplane 3300 lb.				
No. of Seats	Six (Two at +36, 2 at +69, Two at +100)				
Maximum Baggage	Reference weight and balance data				
Fuel Capacity	65 gal. (63.0 gal. usable) two 32.5 gal. tanks in wings at +48				
Oil Capacity	12 qt. at -19.4 (6 qt. usable) See Note 1 for data on system fuel and oil				
Control Surface Movements	Wing flaps	Up	0°	Down	40° +1°, -2°
	Aileron	Up	21° ±2°	Down	14° 30' ±2°
	Elevator	Up	21° ±1°	Down	17° ±1°
	Elevator tab	Up	25° ±1°	Down	5° ±1°
	Rudder	Right	27° 13' ±1°	Left	27° 13' ±1°
	(measured perpendicular to rudder hinge line)				
	Rudder	Right	24° ±1°	Left	24° ±1°
	(measured parallel to 0.0.W.L.)				
Serial Nos. Eligible	P206C, TP206C: P206-0420 thru P206-0519 P206D, TP206D: P206-0520 thru P206-0603 P206E, TP206E: P206-0001, P20600604 thru P20600647				

X. Model U206C/TU206C, 6 PCL-SM (Normal Category), Approved July 20, 1967
Model U206D/TU206D, 6 PCL-SM (Normal Category), Approved September 18, 1968
Model U206E/TU206E, 6 PCL-SM (Normal Category), Approved July 28, 1969
Model U206F/TU206F, 6 PCL-SM (Normal Category), Approved October 26, 1971

Models U206C, U206D, U206E, U206F

Engine: Continental IO-520-F

*Fuel: 100/130 minimum grade aviation gasoline

*Engine Limits: Takeoff (5 min.) at 2850 rpm. (300 hp.)
For all other operations, 2700 rpm. (285 hp.)

Propeller and Propeller Limits:

Landplane

1. McCauley Propeller
   Hub: D2A34C58/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 9.5°
   High pitch: 25.8°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor:
      (1) Woodward 210462
      (2) McCauley C290D2/T4 or C290D4/T4
2. McCauley Propeller
   Hub: D3A32C90/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 11.5°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor:
      (1) Woodward 210462
      (2) McCauley C290D2/T4 or C290D4/T4

X. Models U206C/TU206C, U206D/TU206D, U206E/TU206E, U206F/TU206F (cont'd)

Models U206C, U206D, U206E, U206F (cont'd)

Propeller and Propeller Limits (cont'd)

3. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: D2A37C230
   Blades: 90REB-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 12.0°
   High pitch: 28.3°
   a. Propeller spinner: Cessna 1250909-K200
   b. Hydraulic governor:
   (1) Woodward 210462
   (2) McCauley C290D2/T4 or C290D4/T4

Floatplane

1. McCauley Propeller
   Hub: D2A34C58/90AT-4
   Diameter: Not over 86 in., not under 84 in.
   Pitch settings at 36 in. station:
   Low pitch: 8.0°
   High pitch: 25.0°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor:
   (1) Woodward 210462
   (2) McCauley C290D2/T4 or C290D4/T4
2. McCauley Propeller
   Hub: D3A32C90/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 11.5°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor: McCauley C290D2/T4 or C290D4/T4

Models TU206C, TU206D, TU206E, TU206F

Engine: Continental TSIO-520-C

*Fuel: 100/130 minimum grade aviation gasoline

*Engine Limits: For all operations, 2700 rpm. (285 b.hp.)

Propeller and Propeller Limits

Landplane (S/N U206-0915 thru U2062199)

1. McCauley Propeller
   Hub: D2A34C78/90AT-8 or /90AT-8.5
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 11.8°
   High pitch: 32.0°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
2. McCauley Propeller
   Hub: D3A32C79/82NK-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 13.2°
   High pitch: 32.5°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

X. Models U206C/TU206C, U206D/TU206D, U206E/TU206E, U206F/TU206F (cont'd)

Models TU206C, TU206D, TU206E, TU206F (cont'd)

Propeller and Propeller Limits (cont'd)

3. McCauley Propeller
   Hub: D3A32C90/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.0°
   High pitch: 33.0°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
4. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: D2A37C231
   Blades: 90REB-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.3°
   High pitch: 34.5°
   a. Propeller spinner: Cessna 1250909-K200
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

Floatplane (S/N U206-0915 thru U20602199)

1. McCauley Propeller
   Hub: D2A34C58/90AT-4
   Diameter: Not over 86 in., not under 84 in.
   Pitch settings at 36 in. station:
   Low pitch: 8.0°
   High pitch: 25.0°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
2. McCauley Propeller
   Hub: D3A32C90/82NC-2
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 11.5°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor: McCauley C290D2/T2 or C290D4/T2

Landplane (S/N U20602200 and On)

1. McCauley Propeller
   Hub: E2A34C70/90AT-8
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 36 in. station:
   Low pitch: 11.8°
   High pitch: 32.0°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

X. Models U206C/TU206C, U206D/TU206D, U206E/TU206E, U206F/TU206F (cont'd)

Models TU206C, TU206D, TU206E, TU206F (cont'd)

Propeller and Propeller Limits (cont'd)

2. McCauley Propeller
   Hub: D3A32C88/82NC-2
   Diameter: Not over 80 in., not under 78.0 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.0°
   High pitch: 33.0°
   a. Propeller spinner: Cessna 1250419
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2
3. McCauley constant speed propeller installation (w/incorporation of Cessna Service Kit SK206-49)
   Hub: E2A37C234 with 90REC-8 blades
   Diameter: Not over 82 in., not under 80 in.
   Pitch settings at 30 in. station:
   Low pitch: 14.3°
   High pitch: 34.5°
   a. Propeller spinner: Cessna 1250415
   b. Hydraulic governor:
   (1) Woodward G210452
   (2) McCauley C290D2/T2 or C290D4/T2

*Airspeed Limits (CAS)

U206-0915 thru U206-1444
U20601445 thru U20603020

Never exceed	210 mph. (182 knots)
Maximum structural cruising	170 mph. (148 knots)

Maneuvering: (S/N U206-0915 thru U206-1444 U20601445 thru U20601700)

3300 lb. skiplane	138 mph. (120 knots)
3500 lb. floatplane	138 mph. (120 knots)
3600 lb. floatplane	144 mph. (125 knots)
3600 lb. landplane	144 mph. (125 knots)

(S/N U20601701 thru U20602199)

3300 lb. skiplane	134 mph. (116 knots)
3500 lb. floatplane	138 mph. (120 knots)
3600 lb. floatplane	139 mph. (121 knots)
3600 lb. landplane	139 mph. (121 knots)

(S/N U20602200 thru U20603020)

3300 lb. skiplane	134 mph. (116 knots)
3500 lb. floatplane	138 mph. (120 knots)
3600 lb. landplane	139 mph. (121 knots)

Flaps extended (S/N U206-0915 thru U206-1444 U20601445 thru U20601700) 110 mph. (96 knots)
(S/N U20601701 thru U20603020) 120 mph. (104 knots)

*Aircraft Limits (IAS)

U20603021 and up

Never exceed	185 knots
Maximum structural cruising	151 knots
Maneuvering	
3300 lb. skiplane	120 knots
3500 lb. floatplane	123 knots
3600 lb. landplane	123 knots
Flaps extended	100 knots

X . Models U206C/TU206C, U206D/TU206D, U206E/TU206E, U206F/TU206F (cont'd)

Models TU206C, TU206D, TU206E, TU206F (cont'd)

C.G. Range	Landplane (+33.0) to (+49.7) at 2500 lb. or less (+42.5) to (+49.7) at 3600 lb. Straight line variation between points given Floatplane (EDO 582-3430) (+34.0) to (+47.4) at 2600 lb. or less (+38.5) to (+47.4) at 3500 lb. (U206C, U206D, U206E, U206F) (+39.0) to (+47.4) at 3600 lb. (TU206C, TU206D, TU206E, TU206F) Straight line variation between points given Skiplane (FluiDyne R210) (+33.0) to (+47.4) at 2250 lb. or less (+40.5) to (+47.4) at 3300 lb. Straight line variation between points given
Empty Wt. C.G. Range	None
*Maximum Weight	Landplane 3600 lb. Floatplane 3500 lb. (3600 lb. TU206C, TU206D, TU206E, TU206F) Skiplane 3300 lb.
No. of Seats	Six (Two at +36, 2 at +69, Two at +100) S/N U206-0915 thru U206-1444 U20601445 thru U20601587 Six (Two at +34 to +48, Two at +69 to +79, Two at +92 to +100) S/N U20601588 and On
Maximum Baggage	Reference weight and balance data
Fuel Capacity	65 gal. (63.0 gal. usable) two 32.5 gal. tanks in wings at +48 through S/N U20602126 61 gal. (59.0 gal. usable) two 30.5 gal. tanks in wings at +48 S/N U20602127 and on
Oil Capacity	12 qt. at -19.4 (6 qt. usable) See NOTE 1 for data on system fuel and oil
Control Surface Movements	Wing flaps Landplane Up 0° Down 40° +1°, -2° Floatplane Up 0° Down 30° +1°, -2° Aileron Up 21° ±2° Down 14° 30' ±2° Elevator Up 21° ±1° Down 17° ±1° Elevator tab Landplane Up 25° +1°, -0° Down 5° +1°, -0° Seaplane Up 25° +1°, -0° Down 5° +1°, -0° (S/N U206-0915 thru U206-1444 and U20601445 thru U20601700) Up 21° ±1° Down 9° 30' ±1° (S/N U20601701 and On) Rudder (Land) Right 27° 13' ±1° Left 27° 13' ±1° Rudder (Sea) Right 24° 57' ±1° Left 24° 57' ±1° (measured perpendicular to rudder hinge line) Rudder (Land) Right 24° ±1° Left 24° ±1° Rudder (Sea) Right 22° ±1° Left 22° ±1° (measured parallel to 0.0.W.L.)

X. Models U206C/TU206C, U206D/TU206D, U206E/TU206E, U206F/TU206F (cont'd)

Models TU206C, TU206D, TU206E, TU206F (cont'd)

Serial Nos. Eligible	U206C, TU206C:	U206-0915 thru U206-1234 (1968)
	U206D, TU206D:	U206-1235 thru U206-1444 (1969)
	U206E, TU206E:	U20601445 thru U20601587 (1970)
	U206E, TU206E:	U20601588 thru U20601700 (1971)
	U206F, TU206F:	U20601701 thru U20601874 (1972)
	U206F, TU206F:	U20601875 thru U20602199 (1973)
	U206F, TU206F:	U20602200 thru U20602579 (1974)
	U206F, TU206F:	U20602580 thru U20602588 and U20602590 thru U20603020 (1975)
	U206F, TU206F:	U20603021 thru U20603521 (1976)

XI. Model U206G/TU206G, 6 PCL-SM (Normal Category), Approved June 21, 1976
Model TU206G, S/N U20603522 thru U20604650, 6 PCLM only

Model U206G

Engine: Continental IO-520-F

*Fuel:
100/130 min. grade aviation gasoline
(S/N U20603522 thru U20604074)
100LL/100 min. aviation grade gasoline
(S/N U20602589 and U20604075 and On)

*Engine Limits: Takeoff (5 min.) at 2850 r.p.m. (300 hp.)
For all other operations, 2700 r.p.m. (285 hp.)

Propeller and Propeller Limits:

Landplane

1. McCauley Propeller
   Hub: 3A32C90/82NC-2 (S/N U20603522 thru U20604074)
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 11.5°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor:
   (1) Woodward 210462
   (2) McCauley C290D2/T4 or C290D4/T4
2. McCauley Propeller
   Hub: D3A34C404/80VA-0 (S/N U20602589 and U20604075 and On)
   Diameter: Not over 80 in., not under 78.5 in.
   Pitch settings at 30 in. station:
   Low pitch: 11.0°
   High pitch: 27.0°
   a. Propeller spinner: Cessna 1250419
   b. Hydraulic governor: McCauley C290D4/T4

Floatplane

1. McCauley Propeller
   Hub: D2A34C58/90AT-4 (S/N U20603522 thru U20604074)
   Diameter: Not over 86 in., not under 84 in.
   Pitch settings at 36 in. station:
   Low pitch: 8.0°
   High pitch: 25.0°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor:
   (1) Woodward 210462 or
   (2) McCauley C290D2/T4 or C290D4/T4

XI. Model U206G/TU206G (cont'd)

Model U206G (cont'd)

Propeller and Propeller Limits (cont'd)

2. McCauley Propeller
   Hub: D3A32C90/82NC-2 (S/N U20603522 thru U20604074)
   Diameter: Not over 80 in., not under 78 in.
   Pitch settings at 30 in. station:
   Low pitch: 11.5°
   High pitch: 28.1°
   a. Propeller spinner: Cessna 1250909
   b. Hydraulic governor: McCauley C290D2/T4 or C290D4/T4
3. McCauley Propeller
   Hub: D3A34C404/80VA-0 (S/N U20602589 and U20604075 and On)
   Diameter: Not over 80 in., not under 78.5 in.
   Pitch settings at 30 in. station:
   Low pitch: 11.0°
   High pitch: 27.0°
   a. Propeller spinner: Cessna 1250419
   b. Hydraulic governor: McCauley C290D4/T4

Model TU206G

Engine: Continental TSIO-520-M

*Fuel: 100/130 min. grade aviation gasoline (S/N U20603522 thru U20604074)
100LL/100 min. aviation grade gasoline (S/N U20604075 and On)

*Engine Limits: Takeoff (5 min.) at 2700 r.p.m., 36.5 in. Hg. mp., 310 hp.
For all operations 2600 r.p.m., 35 in. Hg. mp., 285 hp.

Propeller and Propeller Limits

1. McCauley Propeller
   Hub: D3A34C402/90DFA-10
   Diameter: Not over 80 in., not under 78.5 in.
   Pitch settings at 30 in. station:
   Low pitch: 12.4°
   High pitch: 28.5°
   Avoid continuous operation between 1850 and 2150 r.p.m. above 24 in. mp.
   a. Propeller spinner: Cessna 1250419
   b. Hydraulic governor: McCauley C290D4/T2

Model U206G/TU206G

*Airspeed Limits (IAS) (See NOTE 5)

U20603522 thru U20604074

Never exceed (U206G)	185 knots
(TU206G)	183 knots
Maximum structural cruising (U206G)	151 knots
(TU206G)	149 knots
Maneuvering:	
Landplane	120 knots
Floatplane (U206G)	120 knots
Flaps extended	100 knots

U20602589 and U20604075 and On

Never exceed	183 knots
Maximum structural cruising	149 knots
Maneuvering	120 knots
Flaps extended	100 knots

XI. Model U206G/TU206G (cont'd)

C.G. Range	U206G: Landplane (+33.0) to (+49.7) at 2500 lb. or less (+42.5) to (+49.7) at 3600 lb. Straight line variation between points given Floatplane (EDO 582-3430) (+34.0) to (+47.4) at 2600 lb. or less (+38.5) to (+47.4) at 3500 lb. Straight line variation between points given TU206G: Landplane (+33.0) to (+49.7) at 2500 lb. or less (+42.5) to (+49.7) at 3600 lb. Straight line variation between points given Floatplane (EDO 582-3430) (U20604361, U20604650 and On) (+34.0) to (+47.4) at 2600 lb. or less (+39.0) to (+47.4) at 3600 lb. Straight line variation between points given Amphibian (Edo 696-3500) (U20604361, U20604650 and On) (+34.0) to (+47.4) at 2600 lb. or less (+39.0) to (+47.4) at 3600 lb. Straight line variation between points given
*Maximum Weight	U206 Landplane 3600 lb. Floatplane 3500 lb. TU206 Landplane 3600 lb. Floatplane 3600 lb. Amphibian 3600 lb.
No. of Seats	Six (Two at +34 to 48, Two at +69 to +79, Two at +92 to +100)
Maximum Baggage	Reference weight and balance data
Fuel Capacity	(U20603522 thru U20604649) Standard: 61 gal. (59 gal. usable) two 30.5 gal. tanks in wings at +48 Optional: 80 gal. (76 gal. usable) two 40 gal. tanks in wings at +48 (U20602589, U20604650 and On) 92 gal. (88 gal. usable); two 46.0 gal. integral tanks in wings at +46.5
Oil Capacity	12 qt. at -19.4 (8 qt. usable) See NOTE 1 for data on system fuel and oil

XI. Model U206G/TU206G (cont'd)

Control Surface Movements	Wing flaps					
	Landplane	Up	0°	Down	40° +1°, -2°	
	Floatplane/Amphibian	Up	0°	Down	30° +1°, -2°	
	Aileron	Up	21° ±2°	Down	14° 30' ±2°	
	Elevator	Up	21° ±1°	Down	17° ±1°	
	Elevator tab					
	Landplane	Up	25° +1°, -0°	Down	5° +1°, -0°	
	Floatplane/Amphibian	Up	21° ±1°	Down	9° 30' ±1°	
	Rudder					
	Landplane	Right	27° 13' ±1°	Left	27° 13' ±1°	
	Floatplane/Amphibian	Right	24° 57' ±1°	Left	24° 57' ±1°	
	(measured perpendicular to rudder hinge line)					
	Rudder					
	Landplane	Right	24° ±1°	Left	24° ±1°	
	Floatplane/Amphibian	Right	22° ±1°	Left	22° ±1°	
	(measured parallel to 0.0.W.L.)					

Serial Nos. Eligible	U206G/TU206G:	676 and U20603522 thru U20604074 (1977)
	U206G/TU206G:	U20604075 thru U20604649 (1978)
	U206G/TU206G:	U20602589 and U20604650 thru U20605309 (1979)
	U206G/TU206G:	U20605310 thru U20605919 (1980)
	U206G/TU206G:	U20605920 thru U20606439 (1981)
	U206G/TU206G:	U20606440 thru U20606699 (1982)
	U206G/TU206G:	U20606700 thru U20606788 (1983)
	U206G/TU206G:	U20606789 thru U20606846 (1984)
	U206G/TU206G:	U20606847 thru U20606920 (1985)
	U206G/TU206G:	U20606921 thru U20607020 (1986)

Data Pertinent to Model Items I thru XI

Datum	Fuselage Sta. 0.0 (front face of firewall)
Leveling Means	Top of tailcone (thru S/N U20604074) Jig located nutplates and screws on left of tailcone (S/N U20602589 and U20604075 and On)

Certification Basis:
Part 3 of the Civil Air Regulations (CAR) effective May 15, 1956, as amended by 3-1 through 3-8. In addition, effective U20602589 and U20604650 and On, FAR 23.1559 effective March 1, 1978. Dual wheel amphibious float criteria Special Conditions dated January 14, 1969, and Amendment No. 1 dated February 20, 1969. FAR 36 and Amendments 1 through 6, S/N U20604075 and On. Effective S/N U20606847 and On, FAR 23.1545 effective December 1, 1978. Application for Type Certificate (TC) dated November 9, 1962. TC No. A4CE issued July 19, 1963, obtained by the manufacturer under delegation option procedures.

Equivalent Safety Items:	S/N U20602589 and U20603021 and On	
	Airspeed Indicator	CAR 3.757 (See NOTE 5)
	Operating Limitations	CAR 3.778(a)

Production Basis
Production Certificate No. 4. Delegation Option Manufacturer No. CE-1 authorized to Issue airworthiness certificates under delegation option provisions of Part 21 of the Federal Aviation Regulations.

Data Pertinent to Model Items I thru XI (cont'd)

Equipment
The basic required equipment as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the aircraft for certification. This equipment must include a current Airplane Flight Manual (AFM), effective S/N U20602589 and U20604650 and On.

In addition, the following item of equipment is required: Stall Warning Indicator, Cessna Dwg. 0511062-4.

NOTE 1. Current weight and balance report including list of equipment included in the certificated empty weight and loading instructions, when necessary, must be provided for each aircraft at the time of original certification. The certificated empty weight and corresponding center of gravity location must include unusable fuel of 10 lbs. at +46 on the 206, P206, and U206, unusable fuel of 12 lbs. at +46 on the 206 Series (A through G, as applicable to P, TP, U, and TU through S/N U20604649) and 24 lbs. at +48 on TU206 and U206 Series, S/N U20602589 and U20604650 and On, and undrainable oil of 0.0 lb. at -19.4 through S/N U20603020 and full oil of 22.5 lbs. at -19.4 for S/N U20602589 and U20603021 and On.

NOTE 2. The following placards must be displayed as indicated:

A. Applicable to Models 206, P206, and U206 Landplane and Floatplane
(1) In full view of the pilot:
(a) VFR flight only.
"This airplane must be operated as a normal category airplane in compliance with the operating limitations as stated in the form of placards, markings, and manuals. No acrobatic maneuvers including spins approved.

MAXIMUMS
Maneuvering 138 m.p.h. - CAS
Design weigh 3300 lb. (3500 lb.) (Note: ( ) indicates floatplane)
Flight maneuvering load factors:
Flaps up +3.8, -1.52 Flaps down +3.00 (+2.75)
Altitude loss in stall recovery 200 ft. (240 ft.)
Flap extension speed 110 mph. -CAS
Airplane is controllable in 20 knots (12 knots) crosswinds.
Known icing conditions to be avoided.
This airplane is certified for the following flight operations as of date of original airworthiness certification:
VFR-DAY-NIGHT"

(b) VFR or IFR flight
"This airplane must be operated as a normal category airplane in compliance with the operating limitations as stated in the form of placards, markings, and manuals. No acrobatic maneuvers including spins approved.

MAXIMUMS
Maneuvering 138 mph. - CAS
Design weight 3300 lb. (3500 lb.) (Note: ( ) indicates floatplane)
Flight maneuvering load factors:
Flaps up +3.8; -1.52 Flaps down +3.00 (+2.75)
Altitude loss in stall recovery 200 ft. (240 ft.)
Flap extension speed 110 m.p.h. - CAS
Airplane is controllable in 20 knot (12 knot) crosswinds.
Known icing conditions to be avoided.
This airplane is certified for the following flight operations as of date of original airworthiness certification:
VFR-IFR-DAY-NIGHT"

Data Pertinent to Model Items I thru XI (cont'd)

NOTE 2. (cont'd)

(2) On control lock: "Control lock - remove before starting engine."

(3) On fuel selector plate:
(Standard range tanks) "Both off. Left on 31.7 gal. Right on 31.7 gal. Use full rich mixture to switch tanks. Take off and land on fuller tank."
(Optional long-range tanks) "Both off. Left on 40.0 gal. Right on 40.0 gal. Use full rich mixture to switch tanks. Take off and land on fuller tank."

(4) On fuel tank filler cap:
(Standard range tanks) "Tank capacity 32.5 U.S. Gallons, 100/130."
(Optional long-range tanks) "Tank capacity 42.0 U.S. Gallons, 100/130."

(5) Above selector valve:
"Turn pump on 'HI' when switching from dry tank to a tank containing fuel."

(6) On cargo door:
"Refer to weight and balance data for baggage/cargo loading."

(7) In full view of the pilot:
(S/N 206-0138 thru 206-0275) (S/N P206-0002 thru P206-0160) (S/N U206-027 thru U206-0437)
"MAJOR FUEL FLOW FLUCTUATIONS/POWER SURGES
1. AUX FUEL PUMP ON ADJUST MIXTURE
2. SELECT OPPOSITE TANK
3. WHEN FUEL FLOW STEADY, RESUME NORMAL OPERATIONS
SEE PROCEDURE CARD D1189-13 FOR EXPANDED INSTRUCTIONS."

B. Applicable to following models thru S/N U20606439:

P206A	TP206A	U206A	TU206A
P206B	TP206B	U206B	TU206B
P206C	TP206C	U206C	TU206C
P206D	TP206D	U206D	TU206D
P206E	TP206E	U206E	TU206E
		U206F	TU206F
		U206G	TU206G

(1) In full view of the pilot:

(a) On the following models:

P206A	TP206A	U206A	TU206A
P206B	TP206B	U206B	TU206B
P206C	TP206C	U206C	TU206C
P206D	TP206D	U206D	TU206D
P206E	TP206E	U206E	TU206E

"This airplane must be operated as a normal category airplane in compliance with operating limitations as stated in the form of placards, markings, and manuals.
No acrobatic maneuvers, including spins, approved.

MAXIMUMS

Maneuvering speed 144 m.p.h. - CAS for 3600 lb.,
138 m.p.h. - CAS for 3300 - 3500 lb.

Design weight
Landplane 3600 lb.
Floatplane**
Skiplane 3300 lb.

Flight maneuvering load factors:
Flaps up +3.8, -1.52 Flaps down +2.6 all weights
Altitude loss in stall recovery 240 ft. all weights
Flap extension speed 110 m.p.h. - CAS 0° - 40°, 160 m.p.h. -CAS 0° - 10°
Airplane is controllable in crosswinds
Landplane 20 knots
Floatplane 12 knots
Skiplane 10 knots
Known icing conditions to be avoided.
This airplane is certified for the following flight operations as of date of original airworthiness certification.
VFR-IFR-DAY-NIGHT" (as applicable)

Note: **Indicates floatplane data applicable only to the following models with noted design weights:

Data Pertinent to Model Items I thru XI (cont'd)
NOTE 2. (cont'd)

(1) (a) (cont'd):

Model	Design Weight
P206A	3500 lb.
U206A	3500 lb.
U206B	3500 lb.
U206C	3500 lb.
U206D	3500 lb.
U206E	3500 lb.
TP206A	3600 lb.
TU206A	3600 lb.
TU206B	3600 lb.
TU206C	3600 lb.
TU206D	3600 lb.
TU206E	3600 lb.

(b) On the following models: U206F, TU206F (S/N U20601701 thru U20602199)
"This airplane must be operated as a normal category airplane in compliance with the operating limitations stated in the form of placards, markings, and manuals.

	MAXIMUMS			
	Landplane	Floatplane	Turbo Floatplane	Skiplane
Maneuvering speed (CAS)	139 mph (121 knots)	138 mph (120 knots)	139 mph (121 knots)	134 mph (116 knots)
Gross weight	3600 lb.	3500 lb.	3600 lb.	3300 lb.
Alt. loss in stall recovery	240 ft.	240 ft.	240 ft.	240 ft.
Demonstrated crosswind	20 knots	12 knots	12 knots	10 knots
Flight load factor	Flaps up +3.8, -1.52		Flaps down +2.0	

No acrobatic maneuvers, including spins, approved. Known icing conditions to be avoided. This airplane is certified for the following flight operations as of date of original airworthiness certificate:

DAY-NIGHT-VFR-IFR" (as applicable)

On the following models: U206F, TU206F (S/N U20602200 thru U20603020)
"This airplane must be operated as a normal category airplane in compliance with the operating limitations stated in the form of placards, markings, and manuals.

	MAXIMUMS		
	U206/TU206 Landplane	U206 Floatplane	U206/TU206 Skiplane
Maneuvering speed (CAS)	139 mph (121 knots)	138 mph (120 knots)	134 mph (116 knots)
Gross weight	3600 lb.	3500 lb.	3300 lb.
Alt. loss in stall recovery	240 ft.	240 ft.	240 ft.
Demonstrated crosswind	20 knots	12 knots	10 knots
Flight load factor	Flaps up +3.8, -1.52		Flaps down +2.0

No acrobatic maneuvers, including spins, approved. Known icing conditions to be avoided. This airplane is certified for the following flight operations as of date of original airworthiness certificate.

DAY-NIGHT-VFR-IFR" (as applicable)

Data Pertinent to Model Items I thru XI (cont'd)
NOTE 2. (cont'd)

(1) (b) (cont'd):
On the following models: U206F, TU206F (S/N U20603021 thru U20603521)
"This airplane must be operated as a normal category airplane in compliance with the operating limitations stated in the form of placards, markings, and manuals.

	MAXIMUMS		
	U206/TU206 Landplane	U206 Floatplane	U206/TU206 Skiplane
Maneuvering speed (CAS)	123 knots	123 knots	120 knots
Gross weight	3600 lb.	3500 lb.	3300 lb.
Demonstrated crosswind	20 knots	12 knots	10 knots
Flight load factor	Flaps up +3.8, -1.52	Flaps down +2.0	

No acrobatic maneuvers, including spins, approved. Altitude loss in a stall recovery -240 ft.
Flight into known icing conditions prohibited. This airplane is certified for the following flight operations as of date of original airworthiness certificate:

DAY-NIGHT-VFR-IFR" (as applicable)

On the following models: U206G, TU206G (S/N U20603522 thru U20604649)
"This airplane must be operated as a normal category airplane in compliance with the operating limitations stated in the form of placards, markings, and manuals.

	MAXIMUMS	
	U206/TU206 Landplane	U206 Floatplane
Maneuvering speed (CAS)	120 knots	120 knots
Gross weight	3600 lb.	3500 lb.
Flight load factor	Flaps up +3.8, -1.52	Flaps down +2.0

No acrobatic maneuvers, including spins, approved. Altitude loss in a stall recovery -240 ft.
Flight into known icing conditions prohibited. This airplane is certified for the following flight operations as of date of original airworthiness certificate.

DAY-NIGHT-VFR-IFR" (as applicable)

On the following models: U206G, TU206G (S/N U20602589 and U206046 thru U20606439)
"The markings and placards installed in this airplane contain operating limitations which must be complied with when operating this airplane in the Normal Category. Other operating limitations which must be complied with when operating this airplane in this category are contained in the Pilot's Operating Handbook and FAA Approved Airplane Flight Manual.
No acrobatic maneuvers, including spins, approved.
Flight into known icing conditions prohibited.

This airplane is certified for the following flight operations as of date of original airworthiness certificate:

DAY-NIGHT-VFR-IFR" (as applicable)

(2) On control lock through U20606439:
Control lock - remove before starting engine."

Data Pertinent to Model Items I thru XI (cont'd)

NOTE 2. (cont'd)

B. (3) On fuel selector plate

(a) Models P206A TP206A U206A TU206A
206B TP206B U206B TU206B
206C TP206C U206C TU206C
(Standard range tanks) "Both off. Left on 31.5 gal. Right on 31.5 gal.
Use full rich mixture to switch tanks. Take off and land on fuller tank."
(Optional long-range tanks)
"Both off. Left on 40.0 gal. Right on 40.0 gal.
Use full rich mixture to switch tanks. Take off and land on fuller tank."

(b) Models P206D, P206E, TP206D, TP206E, U206D, U206E, U206F, TU206D, TU206E, TU206F (thru S/N U20601986)
(Standard range tanks) "Off. Left on 31.5 gal. Right on 31.5 gal.
Use full rich mixture to switch tanks. Take off and land on fuller tank."
(Optional long-range tanks) "Off. Left on 40.0 gal. Right on 40.0 gal.
Use full rich mixture to switch tanks. Take off and land on fuller tank."

(c) Models U206F, TU206F (S/N U20601987 thru U20602126)
(Standard range tanks) "Off. Left on 31.5 gal. Right on 31.5 gal.
Take off and land on fuller tank."
(Optional long-range tanks) "Off. Left on 40.0 gal. Right on 40.0 gal.
Take off and land on fuller tank."

(d) Models U206F, TU206F, U206G, TU206G (S/N U20602127 thru U20604649)
(Standard range tanks) "Off. Left on 29.5 gal. Right on 29.5 gal.
Take off and land on fuller tank."
(Optional long-range tanks) "Off. Left on 38.0 gal. Right on 38.0 gal.
Take off and land on fuller tank."

Models U206G, TU206G (S/N U20602589 and U20604650 thru U20606439)
"Off. Left on 44.0 gal. Right on 44.0 gal. Take off and land on fuller tank."

B. (4) (a) On fuel tank filler cap or forward of fuel tank filler cap:
Models P206A TP206A U206A TU206A
P206B TP206B U206B TU206B
P206C TP206C U206C TU206C
P206D TP206D U206D TU206D
P206E TP206E U206E TU206E (through S/N U20601666)
(Standard range tanks) "Tank capacity 32.5 U.S. Gals., 100/130."
(Optional long-range tanks) "Tank capacity 42.0 U.S. Gals., 100/130."

(b) [1] Forward of fuel tank filler cap: Models U206E, TU206E, U206F, TU206F (S/N U20601667 thru U20602126)
(Standard range tanks) "Service this aircraft with 100/130 min. aviation grade gasoline - capacity 32.5 gal."
(Optional long-range tanks) "Service this aircraft with 100/130 min. aviation grade gasoline - capacity 42.0 gal."

[2] Forward of fuel tank filler cap: Models U206F, TU206F, U206G, TU206G (S/N U20602127 thru U20604074)
(Standard range tanks) "Service this airplane with 100/130 min. aviation grade gasoline - capacity 30.5 gal."
(Optional long-range tanks) "Service this airplane with 100/130 min. aviation grade gasoline - capacity 40.0 gal."

Data Pertinent to Model Items I thru XI (cont'd)
NOTE 2. (cont'd)

(b) [3] Forward of fuel tank filler cap: Models U206G, TU206G (S/N U20604075 thru U20604649)
(Standard range tanks) "Service this airplane with 100LL/100 min. aviation grade gasoline - capacity 30.5 gal."
(Optional long-range tanks) "Service this airplane with 100LL/100 min. aviation grade gasoline - capacity 40.0 gal."

[4] Forward of fuel tank filler cap: Models U206G, TU206G (S/N U20602589 and U20604650 thru U20603509)
"Fuel. 100LL/100 min. grade aviation gasoline
Capacity 46 U.S. gal., Capacity 32 U.S. gal. to bottom of filler collar."

[5] Forward of fuel tank filler cap: Models U206G, TU206G (S/N U20605310 thru U20606439)
"Fuel. 100LL/100 min. grade aviation gasoline
Capacity 46 U.S. gal., Capacity 34.5 U.S. gal. to bottom of filler collar."

(5) Above fuel selector valve:
(a) Models P206A TP206A U206A TU206A
P206B TP206B U206B TU206B
P206C TP206C U206C TU206C
"Turn pump on 'HI' when switching from dry tank to a tank containing fuel."

(b) Models P206D TP206D U206D TU206D
P206E TP206E U206E TU206E
U206F TU206F (thru S/N U20602199)
"When switching from dry tank turn pump 'On' momentarily."

Models U206F, TU206F, U206G, TU206G (S/N U20602200 thru U20606439)
"When switching from dry tank turn aux. fuel pump 'On' momentarily."

(6) On cargo door:
(a) "Refer to weight and balance data for baggage/cargo loading."
(Effective through S/N U20604649)
(b) "Baggage net 180 lb. maximum capacity refer to weight and balance data for baggage and cargo loading." (S/N U20604650 thru U20606439)

(7) Near manifold pressure gauge:
(a) Models P206A, U206A, P206B, P206D, P206C, P206E
"Climb fuel flow" Sea level 22 g.p.h.
4000 ft. 20 g.p.h.
8000 ft. 18 g.p.h.

(b) [1] Models TP206A TP206D TU206B TU206E
TP206B TP206E TU206C TU206F
TP206C TU206A TU206D
"Maximum allowable manifold pressure and climb fuel flow

Alt. Ft.	Man. Press. In. Hg.	Fuel Flow G.P.H.
S.L. to 19,000	32.5	28
20,000	31.5	26
22,000	29.5	24
24,000	27.5	22
26,000	25.5	20
28,000	23.5	19
30,000	21.5	18

75% power climb: 2500 r.p.m., 27.5 m.p., 20 g.p.h."

Data Pertinent to Model Items I thru XI (cont'd)

NOTE 2. (cont'd)

(b) [2] Model TU206G (S/N U20603522 thru U20604649)
"Maximum power setting and fuel flow
Takeoff (5 min. only): 2700 r.p.m., 36.5 in. m.p., 31 g.p.h.
Maximum continuous power: 2600 r.p.m., 35.0 in. m.p., 27 g.p.h.

Alt. Ft.	Man. Press. In. Hg.	Fuel Flow G.P.H.
S.L. to 17,000	35	27
18,000	34	26
20,000	32	24
22,000	30	22
24,000	28	20
26,000	26	18
28,000	24	17
30,000	22	16

Normal climb: 2500 r.p.m., 30.0 in. m.p., 22 g.p.h."

Model TU206G (S/N U20604650 thru U20606439)

Minimum Fuel Flow Max. Continuous Power: 2600 RPM									
Takeoff		SL-							
2700 r.p.m.	Alt.-Ft./1000	17	18	20	22	24	26	28	30
36.5 in. mp	HP in.Hg.	35	34	32	30	28	26	24	22
31 g.p.h.	Fuel flow GPH	27	26	24	22	20	18	17	16

(c) [1] Models U206B, U206C, U206D, U206E, U206F
"Fuel flow at full throttle

	2700 r.p.m.	2850 r.p.m.
S.L.	23 g.p.h.	24 g.p.h.
4000 ft.	21 g.p.h.	22 g.p.h.
8000 ft.	19 g.p.h.	20 g.p.h."

[2] Model U206G (S/N U20603522 thru U20604649)
"Maximum power settings and fuel flow
Takeoff (5 min. only): 2850 r.p.m.,
Maximum continuous pwr.: 2700 r.p.m.,
"Fuel flow at full throttle

	2700 r.p.m.	2850 r.p.m.
S.L.	23 g.p.h.	24 g.p.h.
4000 ft.	21 g.p.h.	22 g.p.h.
8000 ft.	19 g.p.h.	20 g.p.h.
12000 ft.	17 g.p.h.	18 g.p.h."

Model U206G (S/N U20602589 and U20604650 thru U20606439)
"Minimum fuel flows at full throttle

R.P.M.	S.L.	4000	8000	12000
2700	23 g.p.h.	21 g.p.h.	19 g.p.h.	17 g.p.h.
2850	24 g.p.h.	22 g.p.h.	20 g.p.h.	18 g.p.h."

(d) Model TU206G (S/N U20603522 thru U20606439)
"Avoid continuous operation between 1850 and 2150 r.p.m. above 24 in. mp."

(8) On instrument panel above fuel pump switch:

Models				
	U206A	P206A	TP206A	TU206A
	U206B	P206B	TP206B	TU206B
	U206C	P206C	TP206C	TU206C
	U206D	P206D	TP206D	TU206D

"Use 'HI' for emergency only"

Data Pertinent to Model Items I thru XI (cont'd)

NOTE 2. (cont'd)

(9) Near voltage light:

(a) Models P206E, U206E, U206F (effective thru S/N U20601874)
TP206E, TU206E, TU206F (effective thru S/N U20601874)
"Do not turn off alternator in flight except in emergency."

(b) Models U206F, U206G (S/N U20601875 thru U20604649)
TU206F, TU206G (S/N U20601875 thru U20604649)
"High Voltage"

Models U206G, TU206G (S/N U20602589 and U20604650 thru U20606439)
"Low Voltage"

(10) On the flap control indicator for the following models:

(a) (S/N U20601701 thru U20603020)
"(i) Up to 10° (Partial flap range with blue color code and 160 m.p.h. callout; also mechanical detent at 10°)
(ii) 10° to Full (Indices at these positions with white color code and 120 m.p.h. callout; also mechanical detent at 20°)"

(b) (S/N U20602589 and U20603021 thru U20606439)
"(i) Up to 10° (Partial flap range with blue color code and 140 knot callout; also mechanical detent at 10°)
(ii) 10° to Full (Indices at these positions with white color code and 100 knot callout; also mechanical detent at 20°)"

(11) In full view of the pilot:

(a) Models

P206A	U206A	TP206A	TU206A
P206B	U206B	TP206B	TU206B
P206C	U206C	TP206C	TU206C
P206D	U206D	TP206D	TU206D
P206E	U206E	TP206E	TU206E
	U206F		TU206F
	U206G		TU206G

(S/N P206-0001, P206-0161 thru P206-0603, P20600604 thru P20600647)
(S/N U206-0275 thru U206-1444, U20601445 thru U20604649)
"Major fuel flow fluctuations/power surges
1. Aux fuel pump on adjust mixture
2. Select opposite tank
3. When fuel flow steady, resume normal operations. See Procedure Card D1189-13 for expanded instructions."

(b) Model U206G (S/N U20602589 and U20604650 thru U20605309)
"Major fuel flow fluctuations/power surges
1. Aux fuel pump on, adjust mixture
2. Select opposite tank
3. When fuel flow steady, resume normal operations. See P.O.H. for expanded instructions."

(c) Model TU206G (S/N U20604650 through U20606439)
"Major fuel flow fluctuations/power surges
1. Aux fuel pump on, adjust mixture
2. Select opposite tank
3. When fuel flow steady, resume normal operations. See P.O.H. for expanded instructions."

Data Pertinent to Model Items I thru XI (cont'd)

NOTE 2. (cont'd)

C. Additional placards applicable to floatplane and amphibian only (thru S/N U20606439):

(1) On floor near water rudder retract hook:
"Water rudder always up except for water taxiing."

(2) On instrument panel under wing flap position selector handle:
"Floatplane maximum flaps 30°."

(3) (a) On fuel tank filler cap or forward of fuel tank filler cap:

Models				
	P206A	TP206A	U206A	TU206A
	P206B	TP206B	U206B	TU206B
	P206C	TP206C	U206C	TU206C
	P206D	TP206D	U206D	TU206D
	P206E	TP206E	U206E	TU206E (thru S/N U20601666)

(Standard range tanks) "Tank capacity 32.5 U.S. gals., 100/130."
(Optional long-range tanks) (Inboard fuel tank filler cap)
"Tank capacity 37.0 U.S. gals.,100/130."
(Outboard fuel tank filler cap) "Tank capacity 42.0 U.S. gals., 100/130."

(b) [1] Forward of fuel tank filler cap: Models U206E, TU206E, U206F, TU206F
(S/N U20601667 thru U20602126)
(Standard range tanks) "Service this airplane with 100/130 min.
aviation grade gasoline - capacity 32.5 gal."
(Optional long-range tanks) (Inboard fuel tank filler cap) "Service this airplane with 100/130 min. aviation grade gasoline - capacity 37.0 gal."
(Outboard fuel tank filler cap) "Service this airplane with 100/130 min. aviation grade gasoline - capacity 42.0 gal."

[2] Forward of fuel tank filler cap: Models U206F, TU206F, U206G
(S/N U20602127 thru U20604074)
(Standard range tanks) "Service this airplane with 100/130 min. aviation grade gasoline - capacity 30.5 gal."
(Optional long-range tanks) (Inboard fuel tank filler cap) "Service this airplane with 100/130 min. aviation grade gasoline - capacity 35.0 gal."
(Outboard fuel tank filler cap) "Service this airplane with 100/130 min. aviation grade gasoline - capacity 40.0 gal."

[3] Forward of fuel tank filler cap: Model U206G
(S/N U20604075 thru U20604649)
(Standard range tanks) "Service this airplane with 100LL/100 min. aviation grade gasoline - capacity 30.5 gal."
(Optional long-range tanks) (Inboard fuel tank filler cap) "Service this airplane with 100LL/100 min. aviation grade gasoline - capacity 35.0 gal."
(Outboard fuel tank filler cap) "Service this airplane with 100LL/100 aviation grade gasoline - capacity 40.0 gal."

[4] Forward of fuel tank filler cap: Model U206G, TU206G (S/N U20602589
and U20604650 thru U20605309)
"Fuel. 100LL/100 min. grade aviation gasoline.
Capacity 46 U.S. gal., Capacity 32 U.S. gal. to bottom of filler collar."

[5] Forward of fuel tank filler cap: Model U206G, TU206G (S/N U20605310
through U20606439)
"Fuel. 100LL/100 min. grade aviation gasoline.
Capacity 46 U.S. gal., Capacity 34.5 U.S. Gal. to bottom of filler collar."

Data Pertinent to Model Items I thru XI (cont'd)

NOTE 2. (cont'd)

(4) Outboard of inboard fuel tank filler cap (aircraft with long range tanks only):

Models				
	P206A	TP206A	U206A	TU206A
	P206B	TP206B	U206B	TU206B
	P206C	TP206C	U206C	TU206C
	P206D	TP206D	U206D	TU206D
	P206E	TP206E	U206E	TU206E
			U206F	TU206F
	U206G (effective thru S/N U20604649)			

"To fill tanks to maximum capacity, use outboard fillers."

(5) To the instrument panel near the pilot's control column:
"Avoid tail-low takeoffs and landings with floatplane stinger installed during operations as a landplane."

(6) To the instrument panel near the airspeed indicator:
(a) U206 Series, S/N U20602589, U20603021 thru U20606439
"Floatplane stall speeds are approx. 4 KIAS lower than indicator markings."
(b) TU206 Series, S/N U20604650 through U20606439
"Floatplane/Amphibian flaps up stall speed is approx. 5 KIAS lower than the indicator marking."

D. Applicable to Models 206, U206, and TU206 Series located on the forward cargo door:
(1) Thru S/N U20606091
"Emergency Exit Operation
(a) Open fwd cargo door as far as possible.
(b) Rotate red lever in rear cargo door fwd.
(c) Force rear cargo door full open."

(2) S/N U20606092 and On
"Emergency Exit Operation
(a) Rotate forward cargo door handle full forward then full aft.
(b) Open forward cargo door as far as possible.
(c) Rotate red lever in rear cargo door forward.
(d) Force rear cargo door full open."

E. Additional placards applicable to the skiplane only:
(1) To the instrument panel near the airspeed indicator: Models U206F, TU206F (S/N U20603021 thru U20603521)
"Skiplane flaps up and flaps down stall speeds are approx. 10 KIAS & 5 KIAS lwr., resp., then indicator markings."

F. Effective S/N U20606440 and On:
All placards required in the Pilot's Operating Handbook and FAA Approved Airplane Flight Manual must be installed in the appropriate locations.

NOTE 3. Reserved

NOTE 4. Cylinder head temperature probe location:

206		No. 1 Cylinder
P206	All Models	No. 6 Cylinder
U206	Serials U206-0276 thru U20602199	No. 1 Cylinder
U206	Serials U20602200 thru U20602579	No. 2 Cylinder
U206	Serials U20602580 and On	No. 3 Cylinder
TP206	All Models	No. 5 Cylinder
TP206	Serials U206-0457 thru U20602199	No. 5 Cylinder
TP206	Serials U20602200 thru U20603521	No. 1 Cylinder
TP206	Serials U20603522 and On	No. 5 Cylinder

Data Pertinent to Model Items I thru XI (cont'd)

NOTE 5. The marking of the airspeed indicator with IAS provides an equivalent level of safety to CAR 3.757 when the approved airspeed calibration data presented in Section V of the Pilots Operating Handbook listed below is available to the pilot.

U206F	Cessna P/N D1065-13	
TU206F	Cessna P/N D1066-13	
U206G	Cessna P/N D1090-13	S/N U20603522 thru U20604074
TU206G	Cessna P/N D1091-13	S/N U20603522 thru U20604074
U206G	Cessna P/N D1118-13	S/N U20604075 thru U20604649
TU206G	Cessna P/N D1119-13	S/N U20604075 thru U20604649
U206G	Cessna P/N D1147-13PH	S/N U20602589 and U20604650 thru U20605309
TU206G	Cessna P/N D1148-13PH	S/N U20604650 thru U20605309
U206G	Cessna P/N D1182-13PH	S/N U20605310 thru U20605919
TU206G	Cessna P/N D1183-13PH	S/N U20605310 thru U20605919
U206G	Cessna P/N D1203-13PH	S/N U20605920 thru U20606439
TU206G	Cessna P/N D1204-13PH	S/N U20605920 thru U20606439
U206G	Cessna P/N D1222-13PH	S/N U20606440 thru U20606699
TU206G	Cessna P/N D1223-13PH	S/N U20606440 thru U20606699
U206G	Cessna P/N D1240-13PH	S/N U20606700 thru U20606788
TU206G	Cessna P/N D1241-13PH	S/N U20606700 thru U20606788
U206G	Cessna P/N D1261-13PH	S/N U20606789 thru U20606846
TU206G	Cessna P/N D1262-13PH	S/N U20606789 thru U20606846

NOTE 6. 14-volt electrical system (206 series thru S/N U20604074)
28-volt electrical system (206 series, S/N U20602589 and U20604075 and On)

In addition to the above placards, the prescribed operating limitations indicated by an asterisk (*) under Sections I through XI of this data sheet must also be displayed by permanent markings.

XII. Model 206H, 6 PCLM (Normal Category), Approved November 26, 1997

Engine: Lycoming IO-540-AC1A5, Rated 300 Horsepower

Fuel: 100/100LL minimum grade aviation gasoline

Engine Limits: For all operations, 2700 RPM

Propeller and Propeller Limits:
1. McCauley Constant Speed Propeller
   Hub: B3D36C432/80VSA-1
   Diameter: Not over 79 in., not under 77.5 in.
   Pitch settings at 30 in. station:
   Low pitch: 12.6°
   High pitch: 30.0°
   a. Propeller spinner: Cessna 2150151
   b. Hydraulic governor: McCauley DC290D1/T37

Airspeed Limits:

Maneuvering	125 Knots IAS	(123 Knots CAS)
Max. Structural Cruising	149 Knots IAS	(147 Knots CAS)
Never Exceed	182 Knots IAS	(180 Knots CAS)
Flaps Extended	100 Knots IAS	(100 Knots CAS)

CG Range:
Normal Category:
Aft Limits: 49.7 inches aft of datum at 3600 lb. or less.
Forward Limits: Linear variation from 42.5 inches aft of datum at 3600 lb. to 33.0 inches aft of datum at 2500 lb.; 33.0 inches aft of datum at 2500 lb. or less.

Empty Wt. C.G. Range: None

XII. Model 206H (cont'd)

Reference Datum	Front Face of Firewall (Fuselage Station 0.0)		
MAC	58.8 inches; Leading edge of MAC 25.90 inches aft of datum		
Leveling Means	Left side of Tailcone at 151.85 inches and 180.25 inches aft of datum		
Maximum Weights (See NOTE 4)	Maximum Ramp:	3614 lb.	
	Maximum Takeoff:	3600 lb.	
	Maximum Landing:	3600 lb.	
No. of Seats	Six (Two at 34.0 to 48.0 inches aft of datum; Two at 69.0 to 79.0 inches aft of datum; Two at 98.0 inches aft of datum)		
Maximum Baggage	180 lb. (at 109.0 to 145.0 inches aft of datum)		
Fuel Capacity (Gal.)	(Units 20608001 thru 20608173) 92 gal. total; 88 gal. usable		
	(Units 20608174 and on) 92 gal. total; 87 gal. usable		
	(Two 46 gal. integral tanks in wings at 46.5 inches aft of datum) See NOTE 1 for data on unusable fuel.		
Oil Capacity (Qts.)	11.0 qts. at 12.8 inches forward of datum; 6 qts. usable		
Control Surface Movements	Wing Flaps:		Down 40° +1°, -2°
	Elevator Tab:	Up 25° +1°, -0°	Down 5° +1°, -0°
	Ailerons:	Up 21° ± 2°	Down 14°30' ± 2°
	Elevator: (Relative to stabilizer)	Up 21° ± 1°	Down 17° ± 1°
	Rudder:	Right: 24° ± 1° (Parallel to 0.00 W.L.)	Left: 24° ± 1°
		Right: 27°13' ± 1° (Perpendicular to hinge line)	Left: 27°13' ± 1°
Serial Nos. Eligible	20608001 and On		

Data Pertinent to Model 206H:

Certification Basis
Part 23 of the Federal Aviation Regulations effective February 1, 1965, as amended by 23-1 through 23-6, except as follows:

FAR 23.423, 23.611, 23.619, 23.623, 23.689, 23.775, 23.871, 23.1323, 23.1563 as amended by Amendment 23-7. 23.807 and 23.1524 as amended by Amendment 23-10. 23.507, 23.771, 23.853(a)(b)(c), 23.1365 as amended by Amendment 23-14. 23.951 as amended by Amendment 23-15. 23.607, 23.675, 23.685, 23.733, 23.787, 23.1309, 23.1322 as amended by Amendment 23-17. 23.1301 as amended by Amendment 23-20. 23.1353 and 23.1559 as amended by Amendment 23-21. 23.603, 23.605, 23.613, 23.1329, 23.1545 as amended by Amendment 23-23. 23.441 and 23.1549 as amended by Amendment 23-28. 23.1093 as amended by Amendment 23-29. 23.779 and 23.781 as amended by Amendment 23-33. 23.1, 23.51, 23.561 as amended by Amendment 23-34. 23.301, 23.331, 23.351, 23.427, 23.677, 23.701, 23.735, 23.831 as amended by Amendment 23-42. 23.961, 23.1107(b), 23.1143(g), 23.1147(b), 23.1303, 23.1357, 23.1361, 23.1385 as amended by Amendment 23-43. 23.562(a)(b)(2)(c)(1)(2)(3)(4) as amended by Amendment 23-44. 23.33, 23.53, 23.305, 23.321, 23.485, 23.621, 23.655, 23.731 as amended by Amendment 23-45.

FAR 36 dated December 1, 1969, as amended by Amendments 36-1 through 36-21.

Data Pertinent to Model 206H: (cont'd)
Certification Basis (cont'd)

Additions for the Garmin G1000 Integrated Cockpit System (ICS) Only:
FAR 23.303, 23.307, 23.601, 23.1163(a)(1)(2), 23.1367, 23.1381 as amended by Amendment 23-N/C. 23.1589 as amended by Amendment 23.13. 23.771(a) as amended by Amendment 23.14. 23.607 and (Electrical System) 23.1309(a)(1)(2)(c) as amended by Amendment 23-17. 23.1301, 23.1327, 23.1547(e) as amended by Amendment 23-20. 23.1501 and 23.1541(a)(1)(2)(b)(1)(2) as amended by Amendment 23-21. 23.603 and 23.605 as amended by Amendment 23-23. 23.1529 as amended by Amendment 23-26. 23.561(e), 23.1523, 23.1581(a)(2), 23.1583(a)(1)(2)(b)(h), 23.1585(a)(b)(d) as amended by Amendment 23-34. 23.301 as amended by Amendment 23-42. 23.1322; 23.1331 and 23.1357(a)(b)(c)(d) as amended by Amendment 23-43. 23.305, 23.773(a)(1)(2), 23.1525, 23.1549 as amended by Amendment 23-45. 23.1303(a)(b)(c)(f), 23.1309(a)(1)(i)(ii)(2)(b)(1)(2)(i)(ii)(3)(4)(i)(ii)(iii)(iv)(c)(1)(2)(iii)(c)(3)(d)(e)(f)(1), 23.1311, 23.1321(a)(c)(d)(e), 23.1323(a)(b)(1)(2)(c); 23.1329(g)(h); 23.1351(a)(1)(2)(i)(b)(1)(iii)(2)(3)(c)(4)(d)(1), 23.1353(a)(b)(c)(d)(e), 23.1359(c), 23.1361, 23.1365(a)(b)(d)(e)(f), 23.1431 (a)(b)(d)(e) as amended by Amendment 23-49. 23.1325(a)(b)(1)(2)(i)(3)(c)(d)(e), 23.1543(b)(c), 23.1545(a)(b)(1)(2)(3)(4), 23.1553, 23.1555(a)(b), 23.1563(a), 23.1567(a) as amended by Amendment 23.50. 23.777(a)(b), 23.955(a)(2), 23.1337(a)(1)(2)(b)(1)(c) as amended by Amendment 23.51.
23.1305(a)(1)(2)(3)(b)(2)(3)(i)(4)(i)(5)(6)(i) as amended by Amendment 23-52. 23.901(a)(b) as amended by Amendment 23-53.

Additions for the Garmin GFC-700 Automatic Flight Control System (AFCS) Only:
FAR 23.1335 as amended by Amendment 23-20. 23.1329 (a)(c)(d)(e)(f) as amended by Amendment 23-49.

Equivalent Safety Items:

(1)	Throttle Control	23.1143(g)	FAA letter No. 97-4, dated November 25, 1997
(2)	Mixture Control	23.1147(b)	FAA letter No. 97-4, dated November 25, 1997
(3)	Fuel Tank Sump	23.971	FAA letter No. ACE-02-03, dated January 3, 2002 (Units 20608174 and On)
(4)	Anti-collision Lights	23.1401(d)	FAA letter No. ACE-02-02, dated January 3, 2002 (Units 20608174 and On)
(5)	Aviation White Color Requirement	23.1397(c)	FAA letter No. ACE-07-12, dated November 29, 2007

Date of Application for Amended Type Certificate was October 25, 1996.
Type Certificate No. A4CE was amended November 26, 1997.

Special Conditions as follows:

No. 23-150-SC, Special Conditions: Cessna Aircraft Company, Model 206H, Installation of Electronic Flight Instrument System and the Protection of the System from High Intensity Radiated Fields (HIRF)

Production Basis (Model 206H)
Production Certificate No. PC-4 issued November 25, 1998. Applies to S/N 20608001 and On. Cessna is authorized to issue airworthiness certificates under the delegation provisions of Delegation Option Authorization No. CE-1 in accordance with Part 21 of the Federal Aviation Regulations.

Equipment
The basic required equipment as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the airplane for certification.

NOTE 1. Weight and Balance:

S/N 20608001 thru 20608173
The certificated basic empty weight and corresponding center of gravity location must include unusable fuel of 24 lbs. at 48 inches aft of datum, and full oil of 20.6 lb. at 12.8 inches forward of datum.

S/N 20608174 and On
The certificated basic empty weight and corresponding center of gravity location must include unusable fuel of 30 lbs. At 48 inches aft of datum, and full oil of 20.6 lb. At 12.8 inches forward of datum.

Data Pertinent to Model 206H: (cont'd)

NOTE 2. FAA Approved Airplane Flight Manual (AFM), or Pilot's Operating Handbook and FAA Approved Airplane Flight Manual (POH/AFM): Part number 206HPHUS00 or later FAA approved revisions are applicable to the Model 206H. The Airplane must be operated according to the appropriate AFM or POH/AFM. Required placards are included in the AFM or POH/AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 1200280, Airplane Assembly. A useful placarding reference is the Textron Aviation Illustrated Parts Catalogue (IPC). Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

FAA Approved AFM: Part Number 206HPHAUS-00 (or later FAA approved revisions) are applicable to the Model 206H equipped with Garmin G1000 Integrated Cockpit System. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 1200280, Airplane Assembly. A useful placarding reference is the Textron Aviation Illustrated Parts Catalogue (IPC). Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

FAA Approved AFM: Part Number 206HPHBUS-00 (or later FAA approved revisions) are applicable to the Model 206H equipped with Garmin G1000 Integrated Cockpit System and Garmin GFC-700 AFCS. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 1200280, Airplane Assembly. A useful placarding reference is the Textron Aviation Illustrated Parts Catalogue (IPC). Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

NOTE 3. The CHT probe must be installed on Head #3.

NOTE 4. Model 206H S/N 20608060 through 20608091 may differ structurally and are, therefore, not eligible for any weight increases above the approved maximum takeoff weight limit of 3,600 pounds unless compliance with Cessna Document SSP00-57-01 (latest revision) has been accomplished and documented with an appropriate logbook entry. Any exceptions must first be coordinated with the Central Certification Branch.

NOTE 5. Special Ferry Flight Authorization. Flight Standards District Offices (FSDO) are authorized to issue Special overweight ferry flight authorizations. This airplane is structurally satisfactory for ferry flight if maintained within the following limits:

(1) Takeoff weight must not exceed 130% of the maximum weight for Normal Category,
(2) The Never Exceed Airspeed (V_{NE}) and Max Structural Speed ($V_{C)}$ must be reduced by 30%.,
(3) Forward and aft center of gravity limits may not be exceeded,
(4) Structural load factors of 2.5 g. to -1.0 g. may not be exceeded.

Requirements for any additional oil should be established in accordance with Advisory Circular AC23.1011-1. Increased stall speeds and reduced climb performance should be expected for the increased weights. Flight characteristics and performance at the increased weights have not been evaluated. Flight Permit for operations of overweight aircraft may be found in Advisory Circular AC21-4B.

NOTE 6. The following serial numbers are manufactured under the name Cessna Aircraft Company: 20608001 thru 20608353.

NOTE 7. Company name change, effective 7/29/15. The following serial numbers are manufactured under the name Textron Aviation Inc.: 20608354 and On.

XIII. Model T206H, 6 PCLM (Normal Category), Approved October 1, 1998

Engine	Lycoming TIO-540-AJ1A, Rated 310 Horsepower
Fuel	100/100LL minimum grade aviation gasoline
Engine Limits	For all operations, 2500 RPM
Propeller and Propeller Limits	1. McCauley Constant Speed Propeller Hub: B3D36C432/80VSA-1 Diameter: Not over 79 in., not under 77.5 in. Pitch settings at 30 in. station: Low pitch: 16.9° High pitch: 33.8° a. Propeller spinner: Cessna 2150151 b. Hydraulic governor: McCauley Governor DC290D1/T25

Airspeed Limits			
	Maneuvering	125 Knots IAS	(123 Knots CAS)
	Max. Structural Cruising	149 Knots IAS	(147 Knots CAS)
	Never Exceed	182 Knots IAS	(179 Knots CAS)
	Flaps Extended	100 Knots IAS	(100 Knots CAS)

CG Range	Normal Category:	
	Aft Limits:	49.7 inches aft of datum at 3600 lb. or less.
	Forward Limits:	Linear variation from 42.5 inches aft of datum at 3600 pounds to 33.0 inches aft of datum at 2500 lb.; 33.0 inches aft of datum at 2500 lb. or less.
MAC	58.8 inches; Leading edge of MAC 25.90 inches aft of datum	
Leveling Means	Left side of Tailcone at 151.85 inches and 180.25 inches aft of datum	

Maximum Weights (See NOTE 4)	Maximum Ramp:	3617 lbs.
	Maximum Takeoff:	3600 lbs.
	Maximum Landing:	3600 lbs.
No. of Seats	Six (Two at 34.0 to 48.0 inches aft of datum; Two at 69.0 to 79.0 inches aft of datum; Two at 98.0 inches aft of datum)	
Maximum Baggage	180 lbs. (at 109.0 to 145.0 inches aft of datum)	
Fuel Capacity	(Units T20608001 thru T20608361) 92 gal. total; 88 gal. usable (Units T20608362 and On) 92 gal. total; 87 gal. usable (Two 46 gal. integral tanks in wings at 46.5 inches aft of datum) See NOTE 1 for data on unusable fuel.	
Oil Capacity	11.0 qts. at 12.8 inches forward of datum; 6 qts. usable	

XIII. Model T206H (cont'd)

Empty Wt. C.G. Range	None	
Reference Datum	Front Face of Firewall (Fuselage Station 0.0)	
Control Surface Movements	Wing Flaps:	Down 40° +1°, -2°
	Elevator Tab:	Up 25° +1°, -0° Down 5° +1°, -0°
	Ailerons:	Up 21° ± 2° Down 14°30' ± 2°
	Elevator:	Up 21° ± 1° Down 17° ± 1° (Relative to stabilizer)
	Rudder:	Right: 24° ± 1° Left: 24° ± 1° (Parallel to 0.00 W.L.) Right: 27°13' ± 1° Left: 27°13' ± 1° (Perpendicular to hinge line)
Serial Nos. Eligible	T20608001 and On	

Data Pertinent to Model T206H:

Certification Basis:
Part 23 of the Federal Aviation Regulations (FAR) effective February 1, 1965, as amended by 23-1 through 23-6, except as follows:

FAR 23.423, 23.611, 23.619, 23.623, 23.689, 23.775, 23.871, 23.1323, 23.1563 as amended by Amendment 23-7. 23.807 and 23.1524 as amended by Amendment 23-10. 23.507, 23.771, 23.853(a)(b)(c), 23.1365 as amended by Amendment 23-14. 23.951 as amended by Amendment 23-15. 23.607, 23.675, 23.685, 23.733, 23.787, 23.1309, 23.1322 as amended by Amendment 23-17. 23.1301 as amended by Amendment 23-20.

FAR 23.1353 and 23.1559 as amended by Amendment 23-21. 23.603, 23.605, 23.613, 23.1329, 23.1545 as amended by Amendment 23-23. 23.441 and 23.1549 as amended by Amendment 23-28. 23.779 and 23.781 as amended by Amendment 23-33. 23.1, 23.51, 23.561 as amended by Amendment 23-34. 23.301, 23.331, 23.351, 23.427, 23.677, 23.701, 23.735, 23.831 as amended by Amendment 23-42. 23.961, 23.1093, 23.1107(b), 23.1143(g), 23.1147(b), 23.1303, 23.1357, 23.1361, 23.1385 as amended by Amendment 23-43. 23.562(a)(b)(2)(c)(1)(2)(3)(4) as amended by Amendment 23-44. 23.33, 23.53, 23.305, 23.321, 23.485, 23.621, 23.655, 23.731 as amended by Amendment 23-45.

FAR 36 dated December 1, 1969, as amended by Amendments 36-1 through 36-21.

Additions for the Garmin G1000 Integrated Cockpit System (ICS) Only:
FAR 23.303, 23.307, 23.601, 23.1163(a)(1)(2), 23.1367, 23.1381 as amended by Amendment 23-N/C. 23.1589 as amended by Amendment 23.13. 23.771(a) as amended by Amendment 23.14. 23.607 and (Electrical System) 23.1309(a)(1)(2)(c) as amended by Amendment 23-17. 23.1301, 23.1327, 23.1547(e) as amended by Amendment 23-20. 23.1501 and 23.1541(a)(1)(2)(b)(1)(2) as amended by Amendment 23-21. 23.603 and 23.605 as amended by Amendment 23-23. 23.1529 as amended by Amendment 23-26. 23.561(e), 23.1523, 23.1581(a)(2), 23.1583(a)(1)(2)(b)(h), 23.1585(a)(b)(d) as amended by Amendment 23-34. 23.301 as amended by Amendment 23-42. 23.1322, 23.1331, 23.1357(a)(b)(c)(d) as amended by Amendment 23-43. 23.305, 23.773(a)(1)(2), 23.1525, 23.1549 as amended by Amendment 23-45. 23.1303(a)(b)(c)(f), 23.1309(a)(1)(i)(ii)(2)(b)(1)(2)(i)(ii)(b)(3)(b)(4)(i)(ii)(iii)(iv)(c)(1)(2)(iii)(3)(d)(e)(f)(1), 23.1311, 23.1321(a)(c)(d)(e), 23.1323(a)(b)(1)(2)(c), 23.1329(g)(h), 23.1351(a)(1)(2)(i)(b)(1)(iii)(2)(3)(c)(4)(d)(1), 23.1353(a)(b)(c)(d)(e), 23.1359(c), 23.1361, 23.1365(a)(b)(d)(e)(f), 23.1431(a)(b)(d)(e) as amended by Amendment 23-49. 23.1325(a)(b)(1)(2)(i)(3)(c)(d)(e), 23.1543(b)(c), 23.1545(a)(b)(1)(2)(3)(4), 23.1553, 23.1555(a)(b), 23.1563(a), 23.1567(a) as amended by Amendment 23.50. 23.777(a)(b), 23.955(a)(2), 23.1337(a)(1)(2)(b)(1)(c) as amended by Amendment 23.51. 23.1305(a)(1)(2)(3)(b)(2)(3)(i)(4)(i)(5)(6)(i) as amended by Amendment 23-52. 23.901(a)(b) as amended by Amendment 23-53.

Additions for the Garmin GFC-700 Automatic Flight Control System (AFCS) Only:
FAR 23.1335 as amended by Amendment 23-20, 14 CFR 23.1329 (a)(c)(d)(e)(f) as amended by Amendment 23-49.

Data Pertinent to Model T206H: (cont'd)
Certification Basis (cont'd)

Additions for the Garmin GI 275 Electronic Flight Instrument Only:
14 CFR 23.1327 as amended by Amendment 23-20. 23.1501 as amended by Amendment 23-21. 23.1529 as amended by Amendment 23-26. 23.1523(b) and 23.1581(a)(1)(2) as amended by Amendment 23-34. 23.1322 and 23.1331 as amended by Amendment 23-43. 23.1525 as amended by Amendment 23-45. 23.1303(a)(b)(f), 23.1309(a)(1)(2)(b)(c)(1)(2)(iii)(3)(d)(e)(f), 23.1311(a)(b), 23.1321(a)(c)(d)(e), 23.1323(a)(c), 23.1351(a)(1)(2)(i), 23.1359(c), 23.1365(a)(d)(e), 23.1431(a)(b) as amended by Amendment 23-49. 23.1325(a)(b)(1)(i)(ii)(iii)(2)(i), 23.1543(b)(c), 23.1545(a)(b)(1)(2)(3)(4), 23.1555(a)(b) as amended by Amendment 23-50. 23.777(a)(b) as amended by Amendment 23-51. 23.1308(a)(b)(c) as amended by Amendment 23-57. 23.1306(a)(b) as amended by Amendment 23-61. 23.2010 and 23.2510 as amended by Amendment 23-64 (see NOTE 8). Effective serial numbers T20609639 and On.

Additions for the 14 CFR Part 23, Amendment 64:
14 CFR 23.2000, 23.2005, 23.2010 as amended by Amendment 23-64 (see NOTE 9). Effective S/N T20609639 and On.

Additions for the Garmin G1000 NXi Integrated Cockpit Systems Only:
14 CFR 23.683, 23.867 as amended by Amendment 23-7. 23.613(a)(b) as amended by Amendment 23-45. 23.301(a), 23.337, 23.561(a)(b)(3)(e), 23.607(a)(b), 23.611 as amended by Amendment 23-48. 23.1329(b), 23.1351(g), 23.1353(h), 23.1359(a)(b) as amended by Amendment 23-49. 23.143(c), 23.1555(e)(2), 23.1581(a)(b)(1)(3)(c)(d)(f), 23.1583(h)(m), 23.1585(j) as amended by Amendment 23-50. 23.1337(b)(c) as amended by Amendment 23-51. 23.2515, 23.2520 as amended by Amendment 23-64 (See NOTE 10). Effective S/N T20609786 and On.

Additions for the Dual Electronic Ignition System Only:
14 CFR 23.1309(a)(1)(2), 23.1353(h), 23.1431(b) as amended by Amendment 23-49. 23.1581(a)(2) as amended by Amendment 23-50. 23.901(a)(1)(2)(b)(1)(2) as amended by Amendment 23-53. Effective for S/N T20608319 and T20609859 and On.

Additions for new or changed non-rechargeable lithium battery installations:
14 CFR Part 23 regulations as amended by Amendment 23-64 (see NOTE 11): 14 CFR 23.2010 and 23.2325(g). Effective for S/N T20608319 and T20609859 and On.

Additions for new or changed installations affecting aeronautical security (cybersecurity):
14 CFR Part 23 regulations as amended by Amendment 23-64 (see NOTE 12): 14 CFR 23.2010, 23.2500, 23.2505 and 23.2510. Effective for S/N T20608319 and T20609859 and On.

Equivalent Safety Items:

1. Throttle Control	23.1143(g)	FAA letter No. 97-4, dated October 1, 1998.
2. Mixture Control	23.1147(b)	FAA letter No. 97-4, dated October 1, 1998.
3. Fuel Tank Sump	23.971	FAA letter No. ACE-02-03, dated January 3, 2002. (Units T20608362 and On)
4. Anti-collision Lights	23.1401(d)	FAA letter No. ACE-02-02, dated January 3, 2002. (Units T20608362 and On)
5. Aviation White Color Requirement	23.1397(c)	FAA letter No. ACE-07-12, dated November 29, 2007.

Date of Application for Amended Type Certificate was October 30, 1996.
Type Certificate No. A4CE was amended October 1, 1998.

Special Conditions as follows:

No. 23-150-SC, Special Conditions: Cessna Aircraft Company, Model T206H, Installation of Electronic Flight Instrument System and the Protection of the System from High Intensity Radiated Fields (HIRF).

Production Basis (Model T206H)
Production Certificate No. PC-4 issued November 25, 1998. Applies to airplane S/N T20608001 and On. Cessna is authorized to issue airworthiness certificates under the delegation provisions of Delegation Option Authorization No. CE-1 in accordance with Part 21 of the Federal Aviation Regulations.

Equipment
The basic required equipment as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the airplane for certification.

Data Pertinent to Model T206H (cont'd)

NOTE 1. Weight and Balance:
S/N T20608001 thru T20608361
The certificated basic empty weight and corresponding center of gravity location must include unusable fuel of 24 lbs. at 48 inches aft of datum, and full oil of 20.6 lb. at 12.8 inches forward of datum.

S/N T20608362 and On
The certificated basis empty weight and corresponding center of gravity location must include unusable fuel of 30lbs. at 48 inches aft of datum, and full oil of 20.6 lb. at 12.8 inches forward of datum

NOTE 2. FAA Approved AFM or POH/AFM: Part number T206HPHUS00 or later FAA approved revisions are applicable to the Model T206H. The Airplane must be operated according to the appropriate AFM or POH/AFM. Required placards are included in the AFM or POH/AFM, the applicable operating rules, or the certification basis, and must be installed as specified via the parts list for 1200281, Airplane Assembly. A useful placarding reference is the Textron Aviation Illustrated Parts Catalogue (IPC). Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

FAA Approved AFM: Part Number T206HPHAUS-00 (or later FAA approved revisions) are applicable to the Model T206H equipped with Garmin G1000 Integrated Cockpit System. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis, and must be installed as specified via the parts list for 1200281, Airplane Assembly. A useful placarding reference is the Textron Aviation IPC. Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

FAA Approved AFM: Part Number T206HPHBUS-00 (or later FAA approved revisions) are applicable to the Model T206H equipped with Garmin G1000 Integrated Cockpit System and Garmin GFC-700 AFCS. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis, and must be installed as specified via the parts list for 1200281, Airplane Assembly. A useful placarding reference is the Textron Aviation IPC. Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

FAA Approved AFM: Part Number T206HPHCUS-00 (or later FAA approved revisions) are applicable to the Model T206H equipped with Garmin G1000 NXi Integrated Cockpit System. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis, and must be installed as specified via the parts list for 1200281, Airplane Assembly. A useful placarding reference is the Textron Aviation IPC. Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

NOTE 3. The CHT probe must be installed on Head No. 5.

NOTE 4. Model T206H airplanes, S/N T20608101 thru T20608158, may differ structurally and are, therefore, not eligible for any weight increases above the approved maximum takeoff weight limit of 3,600 pounds unless compliance with Cessna Document SSP00-57-01 (latest revision) has been accomplished and documented with an appropriate logbook entry. Any exceptions must first be coordinated with the Central Certification Branch.

NOTE 5. Special Ferry Flight Authorization. FSDO are authorized to issue Special Overweight Ferry Flight Authorizations. This airplane is structurally satisfactory for ferry flight if maintained within the following limits:

1. Takeoff weight must not exceed 130% of the maximum weight for Normal Category.
2. The Never Exceed Airspeed (V_{NE}) and Maximum Structural Cruising Speed (V_C) must be reduced by 30%.
3. Forward and aft center of gravity limits may not be exceeded.
4. Structural load factors of 2.5 g. to -1.0 g. may not be exceeded.

Requirements for any additional oil should be established in accordance with Advisory Circular AC23.1011-1. Increased stall speeds and reduced climb performance should be expected for the increased weights. Flight characteristics and performance at the increased weights have not been evaluated. Flight Permit for operations of overweight aircraft may be found in Advisory Circular AC21-4B.

NOTE 6. The following S/N are manufactured under the name Cessna Aircraft Company: T20608001 thru T20609184.

NOTE 7. Company name change, effective 7/29/2015. The following serial numbers are manufactured under the name Textron Aviation Inc.: T20609185 and On.

NOTE 8. Means of Compliance for 23.2510 [23-64] is Draft AC 20-184A.

NOTE 9. 14 CFR 23.2005 Certification Level and Performance Level: Level 2, Low Speed.

NOTE 10. Means of Compliance for 23.2515 [23-64] and 23.2520 [23-64] is FAA Policy PS-ACE-23-10.

NOTE 11. Means of Compliance for 23.2010 [23-64] and 23.2325(g) [23-64] is CPP-23.2525-1 (signed 2023.10.19).

NOTE 12. Means of Compliance for 23.2010, 23.2500, 23.2505, and 23.2510 [23-64] is ASTM F3532-22.

-------- END ---------