DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

	3A13
	Revision 76
	Textron Aviation
182	182K
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	October 8, 2026

WARNING: Use of alcohol-based fuels can cause serious performance degradation and fuel system component damage and is therefore prohibited on Cessna airplanes.

TYPE CERTIFICATE DATA SHEET NO. 3A13

This data sheet which is part of Type Certificate No. 3A13 prescribes conditions and limitations under which the product for which the type certificate was issued meets the airworthiness requirements of the Federal Aviation Regulations.

Type Certificate Holder	Textron Aviation Inc. One Cessna Boulevard Wichita, Kansas 67215
Type Certificate Holder Record	Cessna Aircraft Company transferred to Textron Aviation Inc. on July 29, 2015

I. Model 182, Skylane, 4 PCLM (Normal Category), Approved March 2, 1956

Engine	Continental O-470-L
*Fuel	80 minimum grade aviation gasoline
*Engine Limits	For all operations, 2600 r.p.m. (230 hp.)

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I. Model 182 (cont'd)

Propeller and Propeller Limits	1. Hartzell constant speed (a) Hub HC82XF-1 or HCA2XF-1 or BHCA2XF-1 with 8433-2 blades Diameter: not over 82 in., not under 80 in. Pitch settings at 30 in. sta.: low 12°, high 24° (b) Cessna spinner 0752006 (c) Woodward governor 210065, 210105, 210155 or 210340 2. McCauley constant speed (a) Hub 2A36C1-U/90M-8, 2A36C18/90M-8 or 2A36C29/90M-8 blades Diameter: not over 82 in., not under 80 in. Pitch settings at 36 in. sta.: low 10.5°, high 22° (b) Cessna spinner 0752004 (c) Woodward governor 210065, 210105, 210155, 210345 or 210452, or McCauley C290D2/T1 or C290D3/T1 3. Hartzell constant speed (a) Hub BHC-C2YF-1 with 8468-2 blades Diameter: not over 82 in., not under 80 in. Pitch settings at 30 in. sta.: low 13°, high 24° (b) Cessna spinner 0752619 (c) Woodward governor 210105AF, 210340 or 210451 4. McCauley constant speed (a) Hub 2A34C50/90A-8 or 2A34C66/90AT-8 blades Diameter: not over 82 in., not under 80 in. Pitch settings at 36 in. sta.: low 10.5°, high 21.5° (b) Cessna spinner 0752004 (c) Woodward governor 210065, 210105, 210155, 210345 or 210452 or McCauley C290D2/T1 or C290D3/T1 5. Aircraft reworked per Cessna Service Kit SK182-121: McCauley constant speed (Threadless) (a) Hub 2A34C203/90DCA-8 blades Diameter: not over 82 in., not under 80.5 in. Pitch settings at 30 in. sta.: low 12.5°, high 25.0° (b) Cessna spinner 0752004 (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1
*Airspeed Limits (CAS)	Maneuvering 122 m.p.h. (106 knots) Maximum structural cruising 160 m.p.h. (139 knots) Never exceed 184 m.p.h. (160 knots) Flaps extended 100 m.p.h. (87 knots)
C.G. Range	(+39.5) to (+45.8) at 2550 lb. (+35.0) to (+45.8) at 2050 lb. or less Straight line variation between points given.
Empty Wt. C.G. Range	None
*Maximum Weight	2550 lb.
No. of Seats	4 (2 at +36, 2 at +70)
Maximum Baggage	120 lb. (+95)
Fuel Capacity	60 gal. (55 gal. usable); two 30 gal. tanks in wings at +48. (See NOTE 1 for data on unusable fuel).

I. Model 182 (cont'd)

Oil Capacity	12 qt. (-15) (6 qt. usable) (See NOTE 1 for data on undrainable oil).				
Control Surface Movements	Wing flaps	Takeoff		Retracted	0°
				1st notch	10°
				2nd notch	20°
		Landing		3rd notch	30°
				4th notch	40°
	Ailerons	Up	20° ± 2°	Down	14° ± 2°
	Adj. stabilizer	Up	1° 50' ± 15	Down	8° 20' ± 15'
	Elevator (With stabilizer full down)	Up	25° ± 1°	Down	22° 50' ± 1°
	Rudder	Right	24° ± 1°	Left	24° ± 1°
Serial Nos. Eligible	613 and 33000 thru 33842 (1956 Model)				

II. Model 182A, Skylane, 4 PCLM (Normal Category), Approved December 7, 1956

Engine: Continental O-470-L

*Fuel: 80 minimum grade aviation gasoline

*Engine Limits: For all operations, 2600 r.p.m. (230 hp.)

Propeller and Propeller Limits:

1. Hartzell constant speed
   (a) Hub HC82XF-1 or HCA2XF-1 or BHCA2XF-1 with 8433-2 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 30 in. sta.:
   low 12°, high 24°
   (b) Cessna spinner 0752006
   (c) Woodward governor 210065, 210105, 210155 or 210340
2. McCauley constant speed
   (a) Hub 2A36C1-U/90M-8, 2A36C18/90M-8 or 2A36C29/90M-8 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 36 in. sta.:
   low 10.5°, high 22°
   (b) Cessna spinner 0752004
   (c) Woodward governor 210065, 210105, 210155 or 210452,
   or McCauley C290D2/T1 or C290D3/T1
3. Hartzell constant speed
   (a) Hub BHC-C2YF-1 with 8468-2 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 30 in. sta.:
   low 13°, high 24°
   (b) Cessna spinner 0752619
   (c) Woodward governor 210105AF, 210340 or 210451
4. McCauley constant speed
   (a) Hub 2A34C50/90A-8 or 2A34C66/90AT-8 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 36 in. sta.:
   low 10.5°, high 21.5°
   (b) Cessna spinner 0752004
   (c) Woodward governor 210065, 210105, 210155, 210345, 210452,
   or McCauley C290D2/T1 or C290D3/T1

II. Model 182A (cont'd)

Propeller and Propeller Limits (cont'd)

5. Aircraft reworked per Cessna Service Kit SK182-121:
   McCauley constant speed (Threadless)
   (a) Hub 2A34C203/90DCA-8 blades
   Diameter: not over 82 in., not under 80.5 in.
   Pitch settings at 30 in. sta.:
   low 12.5°, high 25.0°
   (b) Cessna spinner 0752004
   (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1

*Airspeed Limits (CAS)	Maneuvering	122 m.p.h. (106 knots)
	Maximum structural cruising	160 m.p.h. (139 knots)
	Never exceed	184 m.p.h. (160 knots)
	Flaps extended	100 m.p.h. (87 knots)

C.G. Range: (+40.0) to (+45.8) at 2650 lb.
(+33.5) to (+45.8) at 2100 lb. or less
Straight line variation between points given.

Empty Wt. C.G. Range: None

*Maximum Weight: 2650 lb.

No. of Seats: 4 (2 at +36, 2 at +70)

Maximum Baggage: 120 lb. (+95)

Fuel Capacity: 65 gal. (55 gal. usable); two 32.5 gal. tanks in wings at +48
(See NOTE 1 for data on unusable fuel).

Oil Capacity: 12 qt. (-15) (6 qt. usable)
(See NOTE 1 for data on undrainable oil).

Control Surface Movements	Wing flaps	Takeoff		Retracted	0°
				1st notch	10°
				2nd notch	20°
		Landing		3rd notch	30°
				4th notch	40°
	Ailerons	Up	20° ± 2°	Down	14° ± 2°
	Adj. stabilizer	Up	1° 50' ± 15'	Down	8° 20' ± 15'
	Elevator (With stabilizer full down)	Up	25° ± 1°	Down	22° 50' ± 1°
	Rudder	Right	24° ± 1°	Left	24° ± 1°

Serial Nos. Eligible: 33843 thru 34753 (1957 Model)
34755 thru 34999 and 51001 thru 51556 (1958 Model)

III. Model 182B, Skylane, 4 PCLM (Normal Category), Approved August 22, 1958

Engine: Continental O-470-L

*Fuel: 80 minimum octane aviation gasoline

*Engine Limits: For all operations, 2600 r.p.m. (230 hp.)

III. Model 182B (cont'd)

Propeller and Propeller Limits

1. Hartzell constant speed
   (a) Hub HC82XF-1 or HCA2XF-1 or BHCA2XF-1 with 8433-2 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 30 in. sta.:
   low 12°, high 24°
   (b) Cessna spinner 0752006
   (c) Woodward governor 210065, 210105, 210155, or 210340
2. McCauley constant speed
   (a) Hub 2A36C1-U/90M-8, 2A36C18/90M-8 or 2A36C29/90M-8 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 36 in. sta.:
   low 10.5°, high 22°
   (b) Cessna spinner 0752004
   (c) Woodward governor 210065, 210105, 210155, 210345, 210452, or McCauley C290D2/T1 or C290D3/T1
3. Hartzell constant speed
   (a) Hub BHC-C2YF-1 with 8468-2 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 30 in. sta.:
   low 13°, high 24°
   (b) Cessna spinner 0752619
   (c) Woodward governor 210105AF, 210340, or 210451 or McCauley C290D2/T1 or C290D3/T1
4. McCauley constant speed
   (a) Hub 2A34C50/90A-8 or 2A34C66/90AT-8 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 36 in. sta.:
   low 10.5°, high 21.5°
   (b) Cessna spinner 0752004
   (c) Woodward governor 210065, 210105, 210155, 210345, 210452,
5. Aircraft reworked per Cessna Service Kit SK182-121:
   McCauley constant speed (Threadless)
   (a) Hub 2A34C203/90DCA-8 blades
   Diameter: not over 82 in., not under 80.5 in.
   Pitch settings at 30 in. sta.:
   low 12.5°, high 25.0°
   (b) Cessna spinner 0752004
   (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1

*Airspeed Limits (CAS)	Maneuvering	122 m.p.h. (106 knots)
	Maximum structural cruising	160 m.p.h. (139 knots)
	Never exceed	184 m.p.h. (160 knots)
	Flaps extended	100 m.p.h. (87 knots)

C.G. Range
(+40.0) to (+45.8) at 2650 lb.
(+33.5) to (+45.8) at 2100 lb. or less
Straight line variation between points given.

Empty Wt. C.G. Range
None

*Maximum Weight
2650 lb.

No. of Seats
4 (2 at +36, 2 at +70)

Maximum Baggage
120 lb. (+95)

Fuel Capacity
65 gal. (55 gal. usable); two 32.5 gal. tanks in wings at +48
(See NOTE 1 for data on unusable fuel).

III. Model 182B (cont'd)

Oil Capacity	12 qt. (-15) (6 qt. usable) (See NOTE 1 for data on undrainable oil).					
Control Surface Movements	Wing flaps	Takeoff			Retracted	0°
					1st notch	10°
					2nd notch	20°
		Landing			3rd notch	30°
					4th notch	40°
	Ailerons		Up	20° ±2°	Down	14° ±2°
	Adj. stabilizer		Up	1° 50' ±15'	Down	8° 20' ±15'
	Elevator (With stabilizer full down)		Up	25° ±1°	Down	22° 50' ±1°
	Rudder		Right	24° ±1°	Left	24° ±1°
Serial Nos. Eligible	34754, 51557 thru 52358 except 51623 (1959 Model)					

IV. Model 182C, Skylane, 4 PCLM (Normal Category), Approved July 8, 1959
Model 182D, Skylane, 4 PCLM (Normal Category), Approved June 14, 1960

Engine: Continental O-470-L

*Fuel: 80 minimum octane aviation gasoline

*Engine Limits: For all operations, 2600 r.p.m. (230 hp.)

Propeller and Propeller Limits:

1. Hartzell constant speed
   (a) Hub HC82XF-1 or HCA2XF-1 or BHCA2XF-1 with 8433-2 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 30 in. sta.:
   low 12°, high 24°
   (b) Cessna spinner 0752006
   (c) Woodward governor 210065, 210105, 210155, or 210340
2. McCauley constant speed
   (a) Hub 2A36C1-U/90M-8, 2A36C18/90M-8 or 2A36C29/90M-8 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 36 in. sta.:
   low 10.5°, high 22°
   (b) Cessna spinner 0752004
   (c) Woodward governor 210065, 210105, 210155, 210345, 210452, or McCauley C290D2/T1 or C290D3/T1
3. Hartzell constant speed
   (a) Hub BHC-C2YF-1 with 8468-2 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 30 in. sta.:
   low 13°, high 24°
   (b) Cessna spinner 0752619
   (c) Woodward governor 210105AF, 210340, or 210451
4. McCauley constant speed
   (a) Hub 2A34C50/90A-8 or 2A34C66/90AT-8 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 36 in. sta.:
   low 10.5°, high 21.5°
   (b) Cessna spinner 0752004
   (c) Woodward governor 210065, 210105, 210155, 210345, 210452, or McCauley C290D2/T1 or C290D3/T1

IV. Model 182C, Model 182D (cont'd)

Propeller and Propeller Limits (cont'd)	5. Aircraft reworked per Cessna Service Kit SK182-121: McCauley constant speed (Threadless) (a) Hub 2A34C203/90DCA-8 blades Diameter: not over 82 in., not under 80.5 in. Pitch settings at 30 in. sta.: low 12.5°, high 25.0° (b) Cessna spinner 0752004 (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1
*Airspeed Limits (CAS)	Maneuvering 122 m.p.h. (106 knots) Maximum structural cruising 160 m.p.h. (139 knots) Never exceed 184 m.p.h. (160 knots) Flaps extended 100 m.p.h. (87 knots)
C.G. Range	(+40.0) to (+45.8) at 2650 lb. (+33.5) to (+45.8) at 2100 lb. or less Straight line variation between points given.
Empty Wt. C.G. Range	None
*Maximum Weight	2650 lb.
No. of Seats	4 (2 at +36, 2 at +70)
Maximum Baggage	120 lb. (+95)
Fuel Capacity	65 gal. (55 gal. usable); two 32.5 gal. tanks in wings at +48 See NOTE 1 for data on unusable fuel
Oil Capacity	12 qt. (-15) (6 qt. usable) See NOTE 1 for data on undrainable oil
Control Surface Movements	Wing flaps Takeoff 0°, 10°, 20° Landing 30°, 40° Ailerons Up 20° ±2° Down 14° ±2° Adj. stabilizer Up 0° 45' ±15' Down 8° 45' ±15' Elevator Up 25° ±1° Down 22° 50' ±1° (With stabilizer full down) Rudder Right 24° ±1° Left 24° ±1° (Measured parallel to 0.0.W.L.)
Serial Nos. Eligible	Model 182C: 631, 52359 thru 53007 (1960 Model) Model 182D: 51623, 18253008 thru 18253598 (1961 Model)

V. Model 182E, Skylane, 4 PCLM (Normal Category), Approved June 27, 1961
Model 182F, Skylane, 4 PCLM (Normal Category), Approved August 1, 1962
Model 182G, Skylane, 4 PCLM (Normal Category), Approved July 19, 1963

Engine	Continental O-470-L or 0-470-R
*Fuel	80/87 minimum grade aviation gasoline
*Engine Limits	For all operations, 2600 r.p.m. (230 hp.)

V. Models 182E, 182F, 182G (cont'd)

Propeller and Propeller Limits

1. Hartzell constant speed
   (a) Hub HC82XF-1 or HCA2XF-1 or BHCA2XF-1 with 8433-2 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 30 in. sta.:
   low 12°, high 24°
   (b) Cessna spinner 0752006
   (c) Woodward governor 210065, 210105, 210155, or 210340
   (Not eligible on O-470-R engine installation)
2. McCauley constant speed
   (a) Hub 2A36C1-U/90M-8, 2A36C18/90M-8 or 2A36C29/90M-8 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 36 in. sta.:
   low 10.5°, high 22°
   (b) Cessna spinner 0752004
   (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or McCauley C290D2/T1 or C290D3/T1
3. Hartzell constant speed
   (a) Hub BHC-C2YF-1 with 8468-2 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 30 in. sta.:
   low 13°, high 24°
   (b) Cessna spinner 0752619
   (c) Woodward governor 210105AF, 210340, or 210451
4. McCauley constant speed
   (a) Hub 2A34C50/90A-8 or 2A34C66/90AT-8 blades
   Diameter: not over 82 in., not under 80 in.
   Pitch settings at 36 in. sta.:
   low 10.5°, high 21.5°
   (b) Cessna spinner 0752004
   (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1
5. Aircraft reworked per Cessna Service Kit SK182-121:
   McCauley constant speed (Threadless)
   (a) Hub 2A34C203/90DCA-8 blades
   Diameter: not over 82 in., not under 80.5 in.
   Pitch settings at 30 in. sta.:
   low 12.5°, high 25.0°
   (b) Cessna spinner 0752004
   (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1

*Airspeed Limits (CAS)	Maneuvering	128 m.p.h. (111 knots)
	Maximum structural cruising	160 m.p.h. (139 knots)
	Never exceed	193 m.p.h. (168 knots)
	Flaps extended	110 m.p.h. (96 knots)

C.G. Range
(+38.4) to (+47.4) at 2800 lb.
(+33.0) to (+47.4) at 2250 lb. or less
Straight line variation between points given.

Empty Wt. C.G. Range
None

*Maximum Weight
2800 lb.

No. of Seats
4 (2 at +36, 2 at +71)

Maximum Baggage
120 lb. (+97)

V. Models 182E, 182F, 182G (cont'd)

Fuel Capacity	65 gal. (60 gal. usable); two 32.5 gal. tanks in wings at +48 (See NOTE 1 for data on unusable fuel).				
Oil Capacity	12 qt. (-15) (6 qt. usable) (See NOTE 1 for data on undrainable oil).				
Control Surface Movements	Wing flaps				40° +1°, -2°
	Elevator tab	Up	25° ±2°	Down	15° ±1°
	Ailerons	Up	20° ±2°	Down	15° ±2°
	Elevator (relative to stabilizer)	Up	26° ±1°	Down	17° ±1°
	Rudder	Right	24° ±1°	Left	24° ±1°
Serial Nos. Eligible	Model 182E: 18253599 thru 18254423 (1962 Model) Model 182F: 18254424 thru 18255058 (1963 Model) Model 182G: 18255059 thru 18255844 (1964 Model)				

VI. Model 182H, Skylane, 4 PCLM (Normal Category), Approved September 17, 1964
Model 182J, Skylane, 4 PCLM (Normal Category), Approved October 20, 1965
Model 182K, Skylane, 4 PCLM (Normal Category), Approved August 3, 1966
Model 182L, Skylane, 4 PCLM (Normal Category), Approved July 28, 1967

Engine	Continental O-470-R
*Fuel	80/87 minimum grade aviation gasoline
*Engine Limits	For all operations, 2600 r.p.m. (230 hp.)
Propeller and Propeller Limits	1. McCauley constant speed (a) Hub 2A34C66/90AT-8 blades Diameter: not over 82 in., not under 80 in. Pitch settings at 36 in. sta.: low 10.5°, high 22° (b) Cessna spinner 0752637 (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1 2. Aircraft reworked per Cessna Service Kit SK182-121: McCauley constant speed (Threadless) (a) Hub 2A34C203/90DCA-8 blades Diameter: not over 82 in., not under 80.5 in. Pitch settings at 30 in. sta.: low 12.5°, high 25.0° (b) Cessna spinner 0752637 (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1

*Airspeed Limits (CAS)	Maneuvering	128 m.p.h. (111 knots)
	Maximum structural cruising	160 m.p.h. (139 knots)
	Never exceed	193 m.p.h. (168 knots)
	Flaps extended	110 m.p.h. (96 knots)

C.G. Range	(+38.4) to (+47.4) at 2800 lb. (+33.0) to (+47.4) at 2250 lb. or less Straight line variation between points given.
Empty Wt. C.G. Range	None
*Maximum Weight	2800 lb.

VI. Models 182H, 182J, 182K, 182L (cont'd)

No. of Seats	4 (2 at +36, 2 at +71)				
Maximum Baggage	120 lb. (+97)				
Fuel Capacity	65 gal. (60 gal. usable); two 32.5 gal. tanks in wings at +48 See NOTE 1 for data on unusable fuel				
Oil Capacity	12 qt. (-15) (6 qt. usable) See NOTE 1 for data on undrainable oil				
Control Surface Movements	Wing flaps				40° +1°, -2°
	Elevator tab	Up	25° ±2°	Down	15° ±1°
	Ailerons	Up	20° ±2°	Down	15° ±2°
	Elevator (relative to stabilizer)	Up	26° ±1°	Down	17° ±1°
	Rudder	Right	24° ±1°	Left	24° ±1°
Serial Nos. Eligible	Model 182H: 634, 18255846 thru 18256684 (1965 Model) Model 182J: 18256685 thru 18257625 (1966 Model) Model 182K: 18255845, 18257626 thru 18257698, 18257700 thru 18258505 (1967 Model) Model 182L: 18258506 thru 18259305 (1968 Model)				

VII. Model 182M, Skylane, 4 PCLM (Normal Category), Approved September 19, 1968

Engine	Continental O-470-R
*Fuel	80/87 minimum grade aviation gasoline
*Engine Limits	For all operations, 2600 r.p.m. (230 hp.)
Propeller and Propeller Limits	1. McCauley constant speed (a) Hub 2A34C66/90AT-8 blades Diameter: not over 82 in., not under 80 in. Pitch settings at 36 in. sta.: low 10.5°, high 22° (b) Cessna spinner 0752637 (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1 2. McCauley constant speed (a) Hub 2A34C201/90DA-8 blades Diameter: not over 82 in., not under 80 in. Pitch settings at 30 in. sta.: low 13°, high 24.5° (b) Cessna spinner 0752637 (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1 3. McCauley constant speed (a) Hub 2A34C203/90DCA-8 blades Diameter: not over 82 in., not under 80.5 in. Pitch settings at 30 in. sta.: low 12.5°, high 25° (b) Cessna spinner 0752637 (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1

VII. Model 182M (cont'd)

*Airspeed Limits (CAS)	Maneuvering		128 m.p.h. (111 knots)		
	Maximum structural cruising		160 m.p.h. (139 knots)		
	Never exceed		193 m.p.h. (168 knots)		
	Flaps extended		110 m.p.h. (96 knots)		
C.G. Range	(+38.4) to (+47.4) at 2800 lb. (+33.0) to (+47.4) at 2250 lb. or less Straight line variation between points given				
Empty Wt. C.G. Range	None				
*Maximum Weight	2800 lb.				
No. of Seats	4 (2 at +36, 2 at +71)				
Maximum Baggage	120 lb. (+97)				
Fuel Capacity	65 gal. (60 gal. usable); two 32.5 gal. tanks in wings at +48 (See NOTE 1 for data on unusable fuel).				
Oil Capacity	12 qt. (-15) (6 qt. usable) (See NOTE 1 for data on undrainable oil).				
Control Surface Movements	Wing flaps				40° +1°, -2°
	Elevator tab	Up	25° ±2°	Down	15° ±1°
	Ailerons	Up	20° ±2°	Down	15° ±2°
	Elevator(relative to stabilizer)	Up	26° ±1°	Down	17° ±1°
	Rudder	Right	24° ±1°	Left	24° ±1°
Serial Nos. Eligible	18257699, 18259306 thru 18260055 (1969 Model)				

VIII. Model 182N, Skylane, 4 PCLM (Normal Category), Approved September 17, 1969

Engine	Continental O-470-R Continental O-470-S (See NOTE 4)
*Fuel	80/87 minimum grade aviation gasoline
*Engine Limits	For all operations, 2600 r.p.m. (230 hp.)
Propeller and Propeller Limits	1. McCauley constant speed (a) Hub 2A34C201/90DA-8 blades Diameter: not over 82 in., not under 80 in. Pitch settings at 30 in. sta.: low 13°, high 24.5° (b) Cessna spinner 0752637 (c) Woodward governor 210065, 210105, 210155, 210345, or A210452, or Garwin 34-828-01-2A, or McCauley C290D2/T1 or C290D3/T1 2. McCauley constant speed (a) Hub 2A34C66/90AT-8 blades Diameter: not over 82 in., not under 80 in. Pitch settings at 36 in. sta.: low 10.5°, high 22° (b) Cessna spinner 0752637 (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1

VIII. Model 182N (cont'd)

Propeller and Propeller Limits (cont'd)

3. McCauley constant speed
   (a) Hub 2A34C203/90DCA-8 blades
   Diameter: not over 82 in., not under 80.5 in.
   Pitch settings at 30 in. sta.:
   low 12.5°, high 25°
   (b) Cessna spinner 0752637
   (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1

*Airspeed Limits (CAS)	Maneuvering	131 m.p.h. (114 knots)
	Maximum structural cruising	160 m.p.h. (139 knots)
	Never exceed	198 m.p.h. (172 knots)
	Flaps extended	110 m.p.h. (96 knots)

C.G. Range
(+39.9) to (+47.4) at 2950 lb.
(+38.4) to (+47.4) at 2800 lb.
(+33.0) to (+47.4) at 2250 lb. or less
Straight line variation between points given

Empty Wt. C.G. Range: None

*Maximum Weight: 2950 lb. takeoff only, 2800 lb. landing

No. of Seats
4 Front standard (2 at +36 to +49)
Optional (2 at +32 to +44)
Rear (2 at +74)

Maximum Baggage
120 lb. (+97) (S/N 18260056 thru 18260445)
120 lb. (+97) and 80 lb. (+117) (S/N 18260446 and On)

Fuel Capacity
65 gal. (60 gal. usable); two 32.5 gal. tanks in wings at +48
(See NOTE 1 for data on unusable fuel).

Oil Capacity
12 qt. (-15) (6 qt. usable)
(See NOTE 1 for data on undrainable oil).

Control Surface Movements	Wing flaps			Down	40° +1°, -2°
	Elevator tab	Up	25° ±2°	Down	15° ±1°
	Ailerons	Up	20° ±2°	Down	15° ±2°
	Elevator (rel. to stabilizer)	Up	26° ±1°	Down	17° ±1°
	Rudder (parallel to 0.00 W.L.)	Right	24° ±1°	Left	24° ±1°
	(Perpendicular to hinge line)	Right	27° 13' ±1°	Left	27° 13' ±1°

Serial Nos. Eligible: 18260056 thru 18260445 (1970 Model) 18260446 thru 18260825 (1971 Model)

IX. Model 182P, Skylane, 4 PCLM (Normal Category), Approved October 8, 1971

Engine	Continental O-470-R, Aircraft S/N 18260826 through 18263475 Continental O-470-S, Aircraft S/N 18260826 and up (See NOTE 4)
*Fuel	80/87 minimum grade aviation gasoline
*Engine Limits	For all operations, 2600 r.p.m. (230 hp.)
Propeller and Propeller Limits	1. McCauley constant speed (a) Hub 2A34C201/90DA-8 blades Diameter: not over 82 in., not under 80 in. Pitch settings at 30 in. sta.: low 13°, high 24.5° (b) Cessna spinner 0752637 (c) Woodward governor 210065, 210105, 210155, 210345, or A210452, or Garwin 34-828-01-2A, or McCauley C290D2/T1 or C290D3/T1 2. McCauley constant speed (a) Hub 2A34C66/90AT-8 blades Diameter: not over 82 in., not under 80 in. Pitch settings at 36 in. sta.: low 10.5°, high 22° (b) Cessna spinner 0752637 (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1 3. McCauley constant speed (a) Hub 2A34C203/90DCA-8 blades Diameter: not over 82 in., not under 80.5 in. Pitch settings at 30 in. sta.: low 12.5°, high 25° (b) Cessna spinner 0752637 (c) Woodward governor 210065, 210105, 210155, 210345, or 210452, or Garwin 34-828-01, or McCauley C290D2/T1 or C290D3/T1
*Airspeed Limits (CAS)	(S/N 675, 18260826 thru 18264295) Maneuvering 126 m.p.h. (109 knots) Maximum structural cruising 160 m.p.h. (139 knots) Never exceed 198 m.p.h. (172 knots) Flaps extended 110 m.p.h. (96 knots)
*Airspeed Limits (IAS) (See NOTE 5 on use of IAS)	(S/N 18264296 thru 18265175) Maneuvering 110 knots Maximum structural cruising 141 knots Never exceed 176 knots Flaps extended 95 knots
C.G. Range	(+39.5) to (+48.5) at 2950 lb. (+33.0) to (+48.5) at 2250 lb. or less Straight line variation between points given
Empty Wt. C.G. Range	None
*Maximum Weight	2950 lb.
No. of Seats	4 (2 front at +32.0 to +50.0) (2 rear at +74)

IX. Model 182P (cont'd)

Maximum Baggage

S/N 18260826 thru 18263475
200 lb. (120 lb. at + 82.0 to +108.0)
(80 lb. at +108.0 to +124.0)
S/N 675 and 18263476 thru 18265175
200 lb. (120 lb. at + 82.0 to +108.0)
(80 lb. at +108.0 to +136.0)

Fuel Capacity

(S/N 675, 18260826 thru 18262250)
Standard Range Tanks:
65 gal. (60 gal. usable); two 32.5 gal. tanks in wings at +48
Long Range Tanks:
84 gal. (79 gal. usable); two 42.0 gal. tanks in wings at +48
(S/N 18262251 thru 18265175)
Standard Range Tanks:
61 gal. (56 gal. usable); two 29 gal. tanks in wings at +48
Long Range Tanks:
80 gal. (75 gal. usable); two 37 gal. tanks in wings at +48
(See NOTE 1 for data on unusable fuel).

Oil Capacity

12 qt. (-15) (6 qt. usable)
(See NOTE 1 for data on undrainable oil).

Control Surface Movements

Wing flaps				Down	40° +1°, -2°
Elevator tab	Up	25° ±2°		Down	15° ±1°
Ailerons	Up	20° ±2°		Down	15° ±2°
Elevator (rel. to stabilizer)	Up	26° ±1°		Down	17° ±1°
Rudder(parallel to 0.00 W.L.)	Right	24° ±1°		Left	24° ±1°
(perpendicular to hinge line)	Right	27° 13' ±1°		Left	27° 13' ±1°

Serial Nos. Eligible

18260826 thru 18261425 (1972 Model)
18261426 thru 18262465 (1973 Model)
18262466 thru 18263475 (1974 Model)
675, 18263476 thru 18264295 except 18263479 (1975 Model)
18264296 thru 18265175 (1976 Model)

X. Model 182Q, Skylane, 4 PCLM (Normal Category), Approved July 28, 1976

Engine

Continental O-470-U

*Fuel

100/130 minimum aviation grade gasoline (S/N 18265176 through 18265965)
100LL/100 aviation grade gasoline (S/N 18265966 through 18267715)

*Engine Limits

For all operations, 2400 r.p.m. (230 hp.)

Propeller and Propeller Limits

McCauley constant speed
(a) Hub C2A34C204/90DCB-8 blades
Diameter: not over 82 in., not under 80.5 in.
Pitch settings at 30 in. sta.:
low 15°, high 29.4°
(b) Cessna spinner 0752637
(c) McCauley governor C290D3/T14

*Airspeed Limits (IAS) (See NOTE 5 on use of IAS)

Maneuvering	111 knots
Maximum structural cruising	143 knots
Never exceed	179 knots
Flaps extended	95 knots

X. Model 182Q (cont'd)

C.G. Range	(+39.5) to (+48.5) at 2950 lb. (+33.0) to (+48.5) at 2250 lb. or less Straight line variation between points given.
Empty Wt. C.G. Range	None
*Maximum Weight	2950 lb.
No. of Seats	4 (2 front at +32.0 to +50.0) (2 rear at +74)
Maximum Baggage	200 lb. (120 lb. at +82.0 to +108.0) (80 lb. at +108.0 to +136.0)
Fuel Capacity	Standard Range Tanks: 61 gal. (56 gal. usable); two 30.5 gal. tanks in wings at +48 (S/N 18263479, 18265176 thru 18266590) Long Range Tanks: 80 gal. (75 gal. usable); two 40.0 gal. tanks in wings at +48 (S/N 18263479, 18265176 thru 18266590) 92 gal. (88 gal. usable); two 46.0 gal. integral tanks in wings at +46.5 (S/N 18266591 thru 18267715) (See NOTE 1 for data on unusable fuel).
Oil Capacity	12 qt. (-15.0) (6 qt. usable) (See NOTE 1 for data on undrainable oil).

Control Surface Movements					
Wing flaps			Down	40° +1°, -2°
Elevator tab	Up	25° ±2°	Down	15° ±1°
Ailerons	Up	20° ±2°	Down	15° ±2°
Elevator (rel. to stabilizer)	Up	26° ±1°	Down	17° ±1°
Rudder (parallel to 0.00 W.L.)	Right	24° ±1°	Left	24° ±1°
(perpendicular to hinge line)	Right	27° 13' ±1°	Left	27° 13' ±1°

Serial Nos. Eligible	18265176 thru 18265965 (1977 Model) 18263479, 18265966 thru 18266590 (1978 Model) 18266591 thru 18267300 (1979 Model) 18267301 thru 18267715, except 18267302 (1980 Model)

XI. Model R182, Skylane RG, 4 PCLM (Normal Category), Approved July 7, 1977
Model TR182, Turbo Skylane RG, 4 PCLM (Normal Category), Approved September 12, 1978

Model R182

Engine	Lycoming O-540-J3C5D, rated at 235 hp.
*Fuel	100LL/100 aviation grade gasoline
*Engine Limits	Full throttle for all operations, 2400 r.p.m.

XI. Models R182, TR182 (cont'd)

Model R182 (cont'd)

Propeller and Propeller Limits

1. McCauley constant speed (S/N R18200002 thru R18201313)
   (a) Hub B2D34C214/90DHB-8 blades
   Diameter: not over 82 in., not under 80.5 in.
   Pitch settings at 30 in. sta.:
   low 15.8°, high 29.4°
   (b) Cessna prop & spinner installation 2250003
   Cessna spinner installation 1750050
   (c) McCauley governor C290D3/T16
   (d) McCauley governor DC290D1/T4

2. McCauley constant speed (S/N R18201314 and On)
   (a) Hub B2D34C218/90DHB-8 blades
   Diameter: not over 82 in., not under 80.5 in.
   Pitch settings at 30 in. sta.:
   low 15.8°, high 29.4°
   (b) Cessna prop & spinner installation 2250124
   Cessna spinner installation 2250123
   (c) McCauley governor C290D3/T22
   (d) McCauley governor DC290D1/T8
3. McCauley constant speed (S/N R18201629 thru R18202041 and aircraft reworked per SK182-71)
   (a) Hub B3D32C407/82NDA-3 blades
   Diameter: not over 79 in., not under 78 in.
   Pitch settings at 30 in. sta.:
   low 16.0°, high 31.7°
   (b) Cessna prop & spinner installation 2252076
   Cessna spinner installation 2252074
   (c) McCauley governor C290D3/T22
   (d) McCauley governor DC290D1/T8

Model TR182

Engine: Lycoming O-540-L3C5D, rated at 235 hp.
(Turbocharged in accordance with Cessna Drawing No. 2250065)

*Fuel: 100LL/100 aviation grade gasoline

*Engine Limits: For all operations, 2400 r.p.m., 31 in. hg. mp.

Propeller and Propeller Limits

1. McCauley constant speed (S/N R18200001, R18200584 through R18201313)
   (a) Hub B2D34C217/90DHB-8 blades
   Diameter: not over 82 in., not under 80.5 in.
   Pitch settings at 30 in. sta.:
   low 15.8°, high 31.9°
   (b) Cessna prop & spinner installation 2250003
   Cessna spinner installation 1750050
   (c) McCauley governor C290D3/T21
   (d) McCauley governor DC290D1/T7

XI. Models R182, TR182 (cont'd)

Model TR182 (cont'd)

Propeller and Propeller Limits (cont'd)

2. McCauley constant speed (S/N R18201314 and on)
   (a) Hub B2D34C219/90DHB-8 blades
   Diameter: not over 82 in., not under 80.5 in.
   Pitch settings at 30 in. sta.:
   low 15.8°, high 31.9°
   (b) Cessna prop & spinner installation 2250124
   Cessna spinner installation 2250123
   (c) McCauley governor C290D3/T22
   (d) McCauley governor DC290D1/T8
3. McCauley constant speed (S/N R18201315, R18201629 and on and aircraft reworked per SK182-71 or SK182-72)
   (a) Hub B3D32C407/82NDA-3 blades
   Diameter: not over 79 in., not under 78 in.
   Pitch settings at 30 in. sta.:
   low 16.0°, high 31.7°
   (b) Cessna prop & spinner installation 2252076
   Cessna spinner installation 2252074
   (c) McCauley governor C290D3/T22
   (d) McCauley governor DC290D1/T8

*Airspeed Limits (IAS) (See NOTE 5 on use of IAS)

1978 Model R182	Maneuvering	112 knots
	Maximum structural cruising	143 knots
	Never exceed	182 knots
	Flaps extended	95 knots
	Landing gear extension	140 knots
1979 Model R182	Maneuvering	112 knots
	Maximum structural cruising	160 knots
	Never exceed	182 knots
	Flaps extended	95 knots
	Landing gear extension	140 knots
Model TR182	Maneuvering	112 knots
	Maximum structural cruising	157 knots
	Never exceed	179 knots
	Flaps extended	95 knots
	Landing gear extension	140 knots
1980 and up Model R182	Maneuvering	112 knots
	Maximum structural cruising	159 knots
	Never exceed	181 knots
	Flaps extended	95 knots
	Landing gear extension	140 knots
Model TR182	Maneuvering	112 knots
	Maximum structural cruising	157 knots
	Never exceed	178 knots
	Flaps extended	95 knots
	Landing gear extension	140 knots

C.G. Range

(a) S/N R18200001 through R18201628 except R18200975 & R18201315
(+40.9) to (+47.0) at 3100 lb.
(+35.5) to (+47.0) at 2700 lb.
(+33.0) to (+47.0) at 2250 lb. or less
Straight line variation between points given
Moment change due to retracting gear (+3052 in.-lb.)

(b) S/N R18200975, R18201315, R18201629 through R18202041
(+40.9) to (+46.0) at 3100 lb.
(+35.5) to (+46.0) at 2700 lb.
(+33.0) to (+46.0) at 2250 lb. or less
Straight line variation between points given
Moment change due to retracting gear (+3052 in.-lb.)

XI. Models R182, TR182 (cont'd)

Model TR182 (cont'd)

Empty Wt. C.G. Range: None

*Maximum Weight: 3100 lb.

No. of Seats: 4 (2 front at +32.0 to +50.0)
(2 rear at +74.0)

Maximum Baggage: 200 lb. (120 lb. at +82.0 to +110.0)
(80 lb. at +110.0 to +134.0)

Fuel Capacity:

(a) S/N R18200002 thru R18200583
Standard Range Tanks:
61 gal. (56 gal. usable); two 30.5 gal. tanks in wings at +48
Long Range Tanks:
80 gal. (75 gal. usable); two 40.0 gal. tanks in wings at +48

(b) S/N R18200001, R18200584 thru R18202041
92 gal. (88 gal. usable); two 46.0 gal. integral tanks
in wings at +46.5

(See NOTE 1 for data on unusable fuel).

Oil Capacity: 9 qt. (-14.8)
(See NOTE 1 for data on oil).

Control Surface Movements:

(a) S/N R18200001 thru R18201628 except R18200975 & R18201315

Wing flaps			Down	40° +1°, -2°
Elevator tab	Up	25° ±2°	Down	15° ±1°
Ailerons	Up	20° ±2°	Down	15° ±2°
Elevator (rel. to stabilizer)	Up	28° ±1°	Down	17° ±1°
Rudder (parallel to 0.00 W.L.)	Right	24° ±1°	Left	24° ±1°
(Perpendicular to hinge line)	Right	27° 13' ±1°	Left	27° 13' ±1°

(b) S/N R18200975, R18201629 thru R18201798

Wing flaps			Down	40° +1°, -2°
Elevator tab	Up	24° ±2°	Down	15° ±1°
Ailerons	Up	20° ±2°	Down	15° ±2°
Elevator (rel. to stabilizer)	Up	28° ±1°	Down	21° ±1°
Rudder (parallel to 0.00 W.L.)	Right	24° +0°, -1°	Left	24° +0°, -1°
(Perpendicular to hinge line)	Right	27° 13' +0°, -1°	Left	27° 13' +0°, -1°

(c) S/N R18201315, R18201799 thru R18202041

Wing flaps			Down	38° +0°, -1°
Elevator tab	Up	24° ±2°	Down	15° ±1°
Ailerons	Up	20° ±1°	Down	15° ±2°
Elevator (rel. to stabilizer)	Up	28° ±1°	Down	21° ±1°
Rudder (parallel to 0.00 W.L.)	Right	24° +0°, -1°	Left	24° +0°, -1°
(Perpendicular to hinge line)	Right	27° 13' +0°, -1°	Left	27° 13' +0°, -1°

Serial Nos. Eligible:

Model	R182:	R18200002 thru R18200583	(1978 Model)
Model	R182/TR182:	R18200001, R18200584 thru R18201313	(1979 Model)
Model	R182/TR182:	R18201314 thru R18201628 except R18201315	(1980 Model)
Model	R182/TR182:	R18201629 thru R18201798	(1981 Model)
Model	R182/TR182:	R18201799 thru R18201928	(1982 Model)
Model	R182/TR182:	R18201929 thru R18201973	(1983 Model)
Model	R182/TR182:	R18201974 thru R18201999	(1984 Model)
Model	R182/TR182:	R18201315, R18202000 thru R18202031	(1985 Model)
Model	R182/TR182:	R18202032 thru R18202041	(1986 Model)

XII. Model 182R, 4 PCLM (Normal Category), Approved August 29, 1980
Model T182, 4 PCLM (Normal Category), Approved August 15, 1980

Model 182R

Engine	Continental O-470-U
*Fuel	100LL/100 aviation grade gasoline
*Engine Limits	For all operations, 2400 r.p.m. (230 hp.)
Propeller and Propeller Limits	McCauley constant speed (a) Hub C2A34C204/90DCB-8 Diameter: not over 82 in., not under 80.5 in. Pitch settings at 30 in. sta.: low 15°, high 29.4° (b) Cessna spinner 0752637 (c) McCauley governor C290D3/T14

Model T182

Engine	Lycoming 0-540-L3C5D, rated at 235 hp. (Turbocharged in accordance with Cessna Drawing No. 2250065)
*Fuel	100LL/100 aviation grade gasoline
*Engine Limits	For all operations, 2400 r.p.m., 31 in. Hg. mp.
Propeller and Propeller Limits	1. McCauley constant speed (a) Hub B2D34C219/90DHB-8 Diameter: not over 82 in., not under 80.5 in. Pitch settings at 30 in. sta.: low 15.8°, high 31.9° (b) Cessna spinner 2250124 (c) McCauley governor C290D3/T22 (d) McCauley governor DC290D1/T8 2. McCauley constant speed (a) Hub B3D32C407/82NDA-3 Diameter: not over 79 in., not under 78 in. Pitch settings at 30 in. sta.: low 16.0°, high 31.7° (b) Cessna spinner 2252076 (c) McCauley governor C290D3/T22 (d) McCauley governor DC290D1/T8

*Airspeed Limits (IAS) (See NOTE 5 on use of IAS)	Model 182R	Maneuvering	111 knots
		Maximum structural cruising	143 knots
		Never exceed	179 knots
		Flaps extended	95 knots
	Model T182	Maneuvering	111 knots
		Maximum structural cruising	140 knots
		Never exceed	178 knots
		Flaps extended	95 knots

C.G. Range	Model 182R	(+40.9) to (+46.0) at 3100 lb. (+33.0) to (+46.0) at 2250 lb. or less Straight line variation between points given
	Model T182	(+40.9) to (+46.0) at 3100 lb. (+35.5) to (+46.0) at 2700 lb. (+33.0) to (+46.0) at 2250 lb. or less Straight line variation between points given
Empty Wt. C.G. Range	None	

XII. Models 182R, T182 (cont'd)

*Maximum Weight
3100 lb. takeoff/flight
2950 lb. landing

No. of Seats
4 (2 front at +32.0 to +50.0)
(2 rear at +74.0)

Maximum Baggage
200 lb. (120 lb. at +92.0 to +108.0)
(80 lb. at +108.0 to +136.0)

Fuel Capacity
92 gal. (88 gal. usable); two 46 gal. integral tanks in wings at +46.5
(See NOTE 1 for data on unusable fuel).

Oil Capacity

Model 182R	Model T182
12 qt. (-15.0) (6 qt. usable) (thru S/N 18268055)	9 qt (-14.8) (6 qt. usable)
12 qt. (-14.1) (6 qt. usable) (S/N 18268056 and On)	

(See NOTE 1 for data on oil)

Control Surface Movements

(a) S/N 18267716 thru 18268055

Wing flaps			Down	40° +1°, -2°
Elevator tab	Up	24° ±2°	Down	15° ±1°
Ailerons	Up	20° ±2°	Down	15° ±2°
Elevator (rel. to stabilizer)	Up	28° ±1°	Down	21° ±1°
Rudder (parallel to 0.00 W.L.)	Right	24° +0°, -1°	Left	24° +1°, -0°
(Perpendicular to hinge line)	Right	27° 13' +0°, -1°	Left	27° 13' +0°, -1°

(b) S/N 18268056 thru 18268586

Wing flaps			Down	38° +0°, -1°
Elevator tab	Up	24° ±2°	Down	15° ±1°
Ailerons	Up	20° ±2°	Down	15° ±2°
Elevator (rel. to stabilizer)	Up	28° ±1°	Down	21° ±1°
Rudder (parallel to 0.00 W.L.)	Right	24° +0°, -1°	Left	24° +0°, -1°
(Perpendicular to hinge line)	Right	27° 13' +0°, -1°	Left	27° 13' +0°, -1°

Serial Nos. Eligible

Model 182R/T182:	18267302, 18267716 through 18268055		(1981 Model)
Model 182R/T182:	18268056	through 18268293	(1982 Model)
Model 182R/T182:	18268294	through 18268368	(1983 Model)
Model 182R/T182:	18268369	through 18268434	(1984 Model)
Model 182R/T182:	18268435	through 18268541	(1985 Model)
Model 182R:	18268542	through 18268586	(1986 Model)

Data Pertinent to Model Items I through XII

Datum	Front face of firewall
Leveling Means	Upper door sill. Top surface centerline of tailcone (S/N 18253599 thru 18265965) Jig located nutplates and screws on left of tailcone (S/N 18263479, 18265966 thru 18268586) (S/N R18200001 thru 18202041)

Certification Basis:

Model 182 Series

Part 3 of the Civil Air Regulations (CAR), dated November 1, 1949, as amended by 3-1 through 3-12 and Paragraph 3.112 as amended October 1, 1959, for Model 182E and On. In addition, effective S/N 18266591 through 18268586, FAR 23.1559 effective March 1, 1978. FAR Part 36 dated December 1, 1969, plus Amendments 36-1 through 36-6 for Model 182Q and On. In addition, effective S/N 18268435 through 18268586, 23.1545(a) Amendment 23-23 dated December 1, 1978.

Model T182

CAR Part 3 dated November 1, 1949, as amended by 3-1 through 3-12 and Paragraph 3.112 as amended October 1, 1959. FAR Part 23 dated February 1, 1965, 23.901, 23.909, 23.1041, 23.1043, 23.1143, 23.1305 as amended February 14, 1975; 23.1559 effective March 1, 1978; FAR Part 36 dated December 1, 1969, plus Amendments 36-1 through 36-10. In addition, effective S/N 18268435 through 18268541, 23.1545(a) Amendment 23-23 dated December 1, 1978.

Model R182

CAR Part 3 dated November 1, 1949, as amended by 3-1 through 3-12 and Paragraph 3.112 as amended October 1, 1959. FAR Part 23 dated February 1, 1965, 23.729, 23.777(e), 23.781, 23.1555(e)(1)(2), 23.1563 as amended February 14, 1975. In addition, effective S/N R18200001, R18200584 and On; 23.1559 effective March 1, 1978. FAR Part 36 dated December 1, 1969, plus Amendments 36-1 through 36-6. In addition, effective at S/N R18202000 through R18202041, 23.1545(a) Amendment 23-23 dated December 1, 1978.

Model TR182

CAR Part 3 dated November 1, 1949, as amended by 3-1 through 3-12 and Paragraph 3.112 as amended October 1, 1969. FAR Part 23 dated February 1, 1965, 23.729, 23.777(e), 23.781, 23.901, 23.909, 23.1041, 23.1043, 23.1143, 23.1305, 23.1555(e)(1)(2), 23.1563 as amended February 14, 1975; 23.1559 effective March 1, 1978; FAR Part 36 dated December 1, 1969, plus Amendments 36-1 through 36-9. In addition, effective at S/N R18202000 through R18202041, 23.1545(a) Amendment 23-23 dated December 1, 1978.

Application for Type Certificate dated July 11, 1955.

Type Certificate No. 3A13 issued March 2, 1956, obtained by the manufacturer under delegation option procedures.

Equivalent Safety Items:

S/N 18263479, 18264296 through 18267715

Airspeed Indicator	CAR 3.757 (See NOTE 5 on use of IAS)
Operating Limitations	CAR 3.778(a)

S/N 18267716 through 18268586

Airspeed Indicator	CAR 3.757 (See NOTE 5 on use of IAS) (S/N 18267716 through 18268434)
Operating Limitations	CAR 3.778(a)
Fuel System	CAR 3.430

S/N R18200001 through R18202041

Airspeed Indicator	CAR 3.757 (See NOTE 5 on use of IAS) (S/N R18200001 through R18201999)
Operating Limitations	CAR 3.778(a)
Fuel System	CAR 3.430

Data Pertinent to Model Items I through XII (cont'd)

Production Basis:
Production Certificate No. 4. Delegation Option Manufacturer No. CE-1 authorized to issue airworthiness certificates under delegation option provisions of Part 21 of the Federal Aviation Regulations.

Equipment:
The basic required equipment as prescribed in the applicable airworthiness requirements (see Certification Basis) must be installed in the aircraft for certification. This equipment must include a current AFM, effective at S/N 18266591 thru 18268586 and R18200584 thru R18202041.

In addition, the following item of equipment is required: Stall warning indicator, Cessna Dwg. S1672-5.

The equipment portion of Aircraft Specification 3A13, Revision 15, or Cessna Publication TS3000-13 should be used for equipment references on all aircraft prior to the Model 182G. Refer to the applicable Equipment List for the Model 182G and subsequent models.

NOTE 1: Current weight and balance report including list of equipment included in certificated empty weight, and loading instructions when necessary, must be provided for each aircraft at the time of original certification.

Serial Numbers: 613 and 33000 thru 34999
631 and 51001 thru 53007
18253008 thru 18264295 except 18263479

The certificated empty weight and corresponding center of gravity location must include unusable fuel of 30 lb. (+46) on Models 182, 182E, 182F, 182G, 182H, 182J, 182K, 182L, 182M, 182N, 182P through S/N 18264295 and 60 lb. (+46) on Models 182A, 182B, 182C, 182D and undrainable oil of (0) lb.

Serial Numbers: 18263479, 18264296 thru 18266590
The certificated empty weight and corresponding center of gravity location must include unusable fuel of 30 lb. (+46) and full oil of 22.5 lb. at (-15.0).

Serial Numbers: 18266591 thru 18268055
The certificated empty weight and corresponding center of gravity location must include unusable fuel of 24 lb. at (+48) and full oil of 22.5 lb. at (-15.0) for Models 182Q, 182R, and include oil of 16.9 lb. at (-14.8) for Model T182.

Serial Numbers: 18268056 thru 18268586
The certificated empty weight and corresponding center of gravity location must include unusable fuel of 24 lb. at (+48) and full oil of 24.4 lb. at (-14.1) for Model 182R and include oil of 16.9 lb. at (-14.8) for Model T182.

Serial Numbers: R18200002 thru R18200583
The certificated empty weight and corresponding center of gravity location must include unusable fuel of 30 lb. (+46) and include oil of 16.9 lb. (-15.7).

Serial Numbers: R18200001, R18200584 thru R18202041
The certificated empty weight and corresponding center of gravity location must include unusable fuel of 24 lb. (+48) and include oil of 16.9 lb. (-14.8).

NOTE 2: The following placards must be displayed in locations as indicated:

A. Applicable to Model 182 only:
(1) In full view of the pilot:
(a) "This airplane must be operated as a normal category airplane in compliance with operating limitations stated in the form of placards, markings and manuals. No acrobatic maneuvers including spins approved.
Flight Maneuvering Load Factors
Flaps Up +3.8 -1.52
Flaps Down +3.5
Maximum design weight 2550 lb.
Reference weight and balance data for loading instructions."
(b) "Both tanks on for takeoff and landing."
(c) "Flaps - Pull to extend

Takeoff	Retracted	0°
	1st Notch	10°
	2nd Notch	20°
Landing	3rd Notch	30°
	4th Notch	40°

(2) In baggage compartment
"Maximum baggage 120 lb. For additional loading instructions see weight and balance data."

B. Applicable to Models 182A, 182B, 182C, 182D
(1) In full view of the pilot:
(a) "This airplane must be operated as a normal category airplane in compliance with operating limitations stated in the form of placards, markings and manuals. No acrobatic maneuvers including spins approved.
Flight Maneuvering Load Factors
Flaps Up +3.8 -1.52
Flaps Down +3.5
Maximum design weight 2650 lb.
Reference weight and balance data for loading instructions."
(b) "Both tanks on for takeoff and landing."
(c) "Flaps - Pull to extend

Takeoff	Retracted	0°
	1st Notch	10°
	2nd Notch	20°
Landing	3rd Notch	30°
	4th Notch	40°"

(2) In baggage compartment
"Maximum baggage 120 lb. For additional loading instructions see weight and balance data."

C. Applicable to Models 182E, 182F, 182G, 182H, 182J, 182K, 182L, 182M
(1) In full view of the pilot:
(a) "This airplane must be operated as a normal category airplane in compliance with operating limitations stated in the form of placards, markings and manuals. No acrobatic maneuvers including spins approved.
Flight Maneuvering Load Factors
Flaps Up +3.8 -1.52
Flaps Down +3.5
Maximum design weight 2800 lb.
Reference weight and balance data for loading instructions."
(2) On the fuel selector valve plate:
"Both off. Left tank level flight only 31 gal. Both on for landing and takeoff all flight attitudes 60 gal. Right tank level flight only 31 gal."
(3) On the control lock:
"Control lock - Remove before starting engine."
(4) On the baggage door:
"120 lb. maximum baggage and/or auxiliary seat passengers. For additional loading instructions, see weight and balance data.

Data Pertinent to Model Items I through XII (cont'd)

D. Applicable to Models 182N:

(1) In full view of the pilot:

(a) S/N 18260056 thru 18260445

"This airplane must be operated as a normal category airplane in compliance with the operating limitations as stated in the form of placards, markings, and manuals.
No acrobatic maneuvers approved, including spins.

	Maximums	
Design weight	2950 lb. takeoff 2800 lb. landing	Alt. loss in stall recovery-160 ft. Flight Maneuvering Load Factors
Maneuvering speed	131 m.p.h.-CAS	Flaps up +3.8, -1.52, Flaps down +3.5

Reference weight and balance data for loading instructions"

(b) S/N 182670446 thru 18260825

"This airplane must be operated as a normal category airplane in compliance with the operating limitations as stated in the form of placards, markings and manuals.

	Maximums
Maneuvering speed	131 m.p.h. CAS (114 knots)
Gross weight	Takeoff 2950 lb. Landing 2800 lb.
Flight load factor	Flaps up +3.8, -1.52 Flaps down +3.5

No acrobatic maneuvers approved, including spins. Altitude loss in a stall recovery 160 ft. Known icing conditions to be avoided. This airplane is certified for the following flight operations as of date of original airworthiness certificate:

DAY - NIGHT - VFR - IFR" (as applicable)

(2) On the fuel selector valve plate:
"Both off. Left tank level flight only 31 gal. Both on for landing and takeoff all flight attitudes, 60 gal. Right tank level flight only 31 gal."

(3) On the control lock:
"Control lock - Remove before starting engine."

(4) On the baggage door:

(a) "120 lb. maximum baggage and/or auxiliary seat passengers. For additional loading instructions, see weight and balance data."
Applicable to Model 182N, S/N 18260056 thru 18260445.

(b) "120 lb. maximum baggage and/or auxiliary passenger forward of baggage door latch, and 80 pounds maximum baggage aft of baggage door latch. Maximum 200 lb. combined. For additional loading instructions see weight and balance data."
Applicable to Model 182N, S/N 18260446 and On.

(5) On flap control indicator:

(a) "0° to 20° - T.O."

(b) "10° - 20° - Full.

(Indices at these positions with blue color code and 160 m.p.h. callout, and white color code with 110 m.p.h. callout; mechanical detent at 10° and 20°)"

Data Pertinent to Model Items I through XII (cont'd)

E. Applicable to Model 182P:

(1) In full view of the pilot:
(S/N 675, 18260826 through 18264295)

(a) "This airplane must be operated as a normal category airplane in compliance with the operating limitations as stated in the form of placards, markings and manuals.

	Maximums
Maneuvering speed	126 m.p.h. CAS (109 knots)
Gross weight	2950 lb.
Flight load factor	Flaps up +3.8, -1.52 Flaps down +2.0

No acrobatic maneuvers, including spins, approved. Altitude loss in a stall recovery 160 ft. Known icing conditions to be avoided. This airplane is certified for the following flight operations as of date of original airworthiness certificate: DAY-NIGHT-VFR-IFR." (as applicable) (S/N 18264296 through 18265175)

NOTE 2. (cont'd)

E. Applicable to Model 182P: (cont'd)

(b) "This airplane must be operated as a normal category airplane in compliance with the operating limitations as stated in the form of placards, markings, and manuals.

	Maximums
Maneuvering speed (IAS)	110 knots
Gross weight	2950 lb.
Flight load factor	Flaps up +3.8, -1.52 Flaps down +2.0

No acrobatic maneuvers approved, including spins. Altitude loss in a stall recovery 160 ft. Flight into known icing conditions prohibited. This airplane is certified for the following flight operations as of date of original airworthiness certificate:

DAY - NIGHT - VFR - IFR" (as applicable)

(2) On the fuel selector valve plate: (S/N 675, 18260826 thru 18262250)
Standard range tanks: "Off. Left tank level flight only 31 gal. Both on for landing and takeoff all flight attitudes, 60 gal. Right tank level flight only 31 gal."
Long range tanks: "Off. Left tank level flight only 39 gal. Both on for landing and takeoff all flight attitudes, 79 gal. Right tank level flight only 39 gal."
On the fuel selector valve plate: (S/N 182622251 thru 18265175)
Standard range tanks: "Off. Left tank level flight only 29 gal. Both on for landing and takeoff all flight attitudes, 56 gal. Right tank level flight only 29 gal."
Long range tanks: "Off. Left tank level flight only 37 gal. Both on for landing and takeoff all flight attitudes, 75 gal. Right tank level flight only 37 gal."

(3) On the control lock: "Control lock - remove before starting engine."

(4) On the baggage door: (S/N 18260826 thru 18263475)
"120 lb. maximum baggage and/or auxiliary passenger forward of baggage door latch, and 80 lb. maximum baggage aft of baggage door latch. Maximum 200 lb. combined. For additional loading instructions, see weight and balance data."

On the baggage door: (S/N 675, 18263476 through 18265175)
"Forward of baggage door latch, 120 lb. maximum baggage and/or auxiliary passenger.
Aft of baggage door latch, 80 lb. maximum baggage including 25 lb. maximum in baggage wall hat shelf. Maximum 200 lb. combined. For additional loading instructions see weight and balance data."

(5) On flap control indicator: (S/N 675, 18260826 thru 18264295)
"(a) 0° to 10° - (Blue color code and 160 m.p.h. callout;
also, mechanical detent at 10°)
(b) 10° to 20°- Full (Indices at these positions with white color code and

110 m.p.h. callout; also, mechanical detent at 10° and 20°)"

On flap control indicator (S/N 18264296 thru 18265175)

"(a) 0° to 10° - (Blue color code and 140 KTS callout; also, mechanical detent at 10°)

(b) 10° to 20°- Full (Indices at these positions with white color code and 95 KTS callout; also, mechanical detent at 10° and 20°)"

Data Pertinent to Model Items I through XII (cont'd)
NOTE 2. (cont'd)

(6) Forward of the filler cap on the wing surface: (S/N 675, 18260826 through 18262250)
Standard range tanks: "Service this airplane with 80/87 minimum aviation grade gasoline. Capacity 32.5 gal."
Long range tanks: "Service this airplane with 80/87 minimum aviation grade gasoline. Capacity 42.0 gal."
Forward of the filler cap on the wing surface: (S/N 18262251 through 18265175)
Standard range tanks: "Service this airplane with 80/87 minimum aviation grade gasoline. Capacity 30.5 gal."
Long range tanks: "Service this airplane with 80/87 minimum aviation grade gasoline. Capacity 40.0 gal."

(7) On aft panel of baggage compartment:
"Oxygen refill." (All models with oxygen)

(8) Adjacent to overvoltage light:
"High voltage."

(9) Above the left fuel gauge:
"Do not turn off alternator in flight except in emergency."
(Model 182P, S/N 18260826 thru 18261425)

F. Applicable to Model 182Q:
(1) In full view of the pilot:
(a) S/N 18263479, 18265176 thru 18266590
"This airplane must be operated as a normal category airplane in compliance with the operating limitations as stated in the form of placards, markings and manuals.

	Maximums
Maneuvering speed (IAS)	111 knots
Gross weight	2950 lb.
Flight load factor	Flaps up +3.8, -1.52
	Flaps down +2.0

No acrobatic maneuvers, including spins, approved. Altitude loss in a stall recovery 160 ft. Flight into known icing conditions prohibited. This airplane is certified for the following flight operations as of date of original airworthiness certificate:

DAY - NIGHT - VFR - IFR." (as applicable)

S/N 18266591 thru 18267715
"The markings and placards installed in this airplane contain operating limitations which must be complied with when operating this airplane in the Normal Category. Other operating limitations which must be complied with when operating this airplane in this category are contained in the POH/AFM.

No acrobatic maneuvers, including spins, approved. Flight into known icing conditions prohibited. This airplane is certified for the following flight operations as of date of original airworthiness certificate:

DAY - NIGHT - VFR - IFR." (as applicable)

(b) Near airspeed indicator:
S/N 18266591 thru 18267715
"Maneuver Speed
111 KIAS"

Data Pertinent to Model Items I through XII (cont'd)
NOTE 2. (cont'd)

(2) On the fuel selector valve plate:
S/N 18263479, 18265176 through 18266590
Standard range tanks: "Off.
Left - 29 gal. Level flight only.
Both - 56 gal. All flight attitudes.
Both on for takeoff and landing.
Right - 29 gal. Level flight only."

Long range tanks: "Off.
Left - 37 gal. Level flight only.
Both - 75 gal. All flight attitudes.
Both on for takeoff and landing.
Right - 37 gal. Level flight only."

S/N 18266591 thru 18267715
"Take Off - Both - Landing,
All Flight - 88.0 Gal. - Attitudes
Left - 44.0 Gal. Level Flight Only
Right - 44.0 Gal. Level Flight Only
Off."

(3) On the control lock: "Control lock - remove before starting engine."

(4) On the baggage door: "Forward of baggage door latch, 120 pounds maximum baggage and/or auxiliary passenger. Aft of baggage door latch, 80 pounds maximum baggage including 25 pounds maximum in baggage wall hat shelf. Maximum 200 pounds combined. For additional loading instructions, see weight and balance data."

(5) On flap control indicator:
"0° to 10° - (Blue color code and 140 KTS callout;
also, mechanical detent at 10°)"
"0° to 20° - Full (Indices at these positions with white color code and 95 KTS callout; also, mechanical detent at 10° and 20°)"

(6) Forward of the filler cap on the wing surface:
S/N 18265176 thru 18265965
Standard range tanks: "Service this airplane with 100/130 minimum aviation grade gasoline.
Capacity 30.5 gal."

Long range tanks: "Service this airplane with 100/130 minimum aviation grade gasoline.
Capacity 40.0 gal."

S/N 18263479, 18265966 thru 18266590
Standard range tanks: "Service this airplane with 100LL/100 aviation grade gasoline.
Capacity 30.5 gal."

Long range tanks: "Service this airplane with 100LL/100 aviation grade gasoline.
Capacity 40.0 gal."

S/N 18266591 thru 18267715
"Fuel 100LL/100 minimum grade aviation gasoline.
Capacity 46 U.S. gal. Capacity 34.5 U.S. gal.
To bottom of filler collar."

Data Pertinent to Model Items I through XII (cont'd)
NOTE 2. (cont'd)

(7) On aft panel of baggage compartment:
"Oxygen refill." (All models with oxygen)

(8) Adjacent to overvoltage light:
S/N 18263479, 18265176 through 18266590
"High Voltage"

S/N 18266591 through 18267715
"Low Voltage"

G. Applicable to Models R182, TR182, S/N R18200001 thru R18201928:
(1) In full view of the pilot:
(a) S/N R18200002 thru R18200583
"This airplane must be operated as a normal category airplane in compliance with the operating limitations as stated in the form of placards, markings, and manuals.

	Maximums
Gross weight	3100 lb.
Flight load factor	Flaps up +3.8, -1.52 Flaps down +2.0

No acrobatic maneuvers, including spins, approved. Altitude loss in a stall recovery 240 ft. Flight into known icing conditions prohibited. This airplane is certified for the following flight operations as of date of original airworthiness certificate:

DAY - NIGHT - VFR - IFR." (as applicable)

(b) S/N R18200001, R18200584 thru R18202041
"The markings and placards installed in this airplane contain operating limitations which must be complied with when operating this airplane in the Normal Category. Other operating limitations which must be complied with when operating this airplane in this category are contained in the Pilot's Operating Handbook and FAA Approved Airplane Flight Manual.

No acrobatic maneuvers, including spins, approved. Flight into known icing conditions prohibited. This airplane is certified for the following flight operations as of date of original airworthiness certificate:

DAY - NIGHT - VFR - IFR." (as applicable)

(c) Near Airspeed Indicator:
"MAX SPEED - KIAS

Maneuver	112
Gear Oper	140
Gear Down	140"

(2) On the fuel selector valve plate:
(a) S/N R18200002 thru R18200583

Standard range tanks:	"Off Left - 29 gal. Level flight only. Both - 56 gal. All flight attitudes. Both on for takeoff and landing. Right - 29 gal. Level flight only."
Long range tanks:	"Off Left - 37 gal. Level flight only. Both - 75 gal. All flight attitudes. Both on for takeoff and landing. Right - 37 gal. Level flight only."

Data Pertinent to Model Items I through XII (cont'd)
NOTE 2. (cont'd)

(b) S/N R18200001, R18200584 thru R18201798
"Take Off - Both - Landing,
All Flight - 88.0 Gal. - Attitudes
Left - 44.0 Gal. Level Flight Only
Right - 44.0 Gal. Level Flight Only
Off."

(c) S/N R18201799 thru R18202041
"Both - 88.0 Gal. - Take Off - Landing - All Flight Attitudes; Left - 44.0 Gal. - Level Flight Only
Right - 44.0 Gal. - Level Flight Only
Off - Off."

(3) On the control lock:
(a) S/N R18200001 thru R18201798
"Control lock - Remove before starting engine."
(b) S/N R18201799 thru R18202041
"Caution! Control Lock - Remove before starting engine."

(4) On the baggage door: "120 Pounds Maximum
Baggage and/or Auxiliary Passenger
Forward of Baggage Door Latch And
80 Pounds Maximum
Baggage Aft of Baggage Door Latch
Maximum 200 Pounds Combined
For Additional Loading Instructions See Weight and Balance Data"

(5) On the flap control indicator:
"0° to 10° - (Blue color code and 140 KTS callout;
also, mechanical detent at 10°)"

"0° to 20° - Full (Indices at these positions with white color code and 95 KTS
callout; also, mechanical detent at 10° and 20°)"

(6) Forward of the filler cap on the wing surface:
(a) S/N R18200002 thru R18200583
Standard range tanks: "Service this airplane with 100LL/100 aviation grade gasoline.

Capacity 30.5 gal."

Long range tanks: "Service this airplane with 100LL/100 aviation grade gasoline.
Capacity 40.0 gal."

(b) S/N R18200001, R18200584 thru R18202041
Fuel 100LL/100 minimum grade aviation gasoline.
Capacity 46 U.S. gal. Capacity 34.5 U.S. gal.
To bottom of filler collar."

(7) Adjacent to overvoltage light:
(a) S/N R18200002 through R18200583
"High Voltage"

(b) S/N R18200001, R18200584 through R18202041
"Low Voltage"

Data Pertinent to Model Items I through XII (cont'd)
NOTE 2. (cont'd)

(8) Near gear hand pump:
"Manual Gear Extension
1. Select Gear Down
2. Pull Handle Fwd.
3. Pump Vertically
CAUTION
Do Not Pump With Gear
Up Selected"

(9) Forward of each fuel filler cap:
"Fuel Cap Forward - Arrow Alignment, Cap Must Not Rotate During Closing."

H. Applicable to Models 182R and T182, S/N 18267302, 18267716 through 18268293:
(1) In full view of the pilot:
(a) "The markings and placards installed in this airplane contain operating limitations which must be complied with when operating this airplane in the Normal Category. Other operating limitations which must be complied with when operating this airplane in this category are contained in the Pilot's Operating Handbook (POH) and FAA Approved Airplane Flight Manual (AFM).

No acrobatic maneuvers, including spins, approved. Flight into known icing conditions prohibited. This airplane is certified for the following flight operations as of date of original airworthiness certificate:

DAY - NIGHT - VFR - IFR." (as applicable)

(b) Near airspeed indicator: "Maneuver Speed 111 KIAS"

(2) On the fuel selector valve plate:
(a) S/N 18267716 thru 18268055
"Take Off - Both - Landing, -
All Flight - 88.0 Gal. - Attitudes
Left - 44.0 Gal. Level Flight Only
Right - 44.0 Gal. Level Flight Only
Off."

(b) S/N 18268056 thru 18268586
"Both - 88.0 Gal. - Takeoff - Landing - All Flight Attitudes
Left - 44.0 Gal. - Level Flight Only
Right - 44.0 Gal. - Level Flight Only
Off - Off."

(3) On the control lock:
(a) S/N 18267716 thru 18268055
"Control Lock - Remove before starting engine."

(b) S/N 18268056 thru 18268586
"Caution! Control Lock - Remove before starting engine."

(4) On baggage door:
"120 Pounds Maximum
Baggage and/or Auxiliary Passenger
Forward of Baggage Door Latch and 80 Pounds Maximum
Baggage Aft of Baggage Door Latch
Maximum 200 Pounds Combined

For Additional Loading Instructions see Weight and Balance Data"

Data Pertinent to Model Items I through XII (cont'd)
NOTE 2. (cont'd)

(5) On flap control indicator:
"0° to 10° - (Blue color code and 140 KTS callout;
also, mechanical detent at 10°)"
"0° to 20° - Full (Indices at these positions with white color code and 95 KTS
callout; also mechanical detent at 10° and 20°)"

(6) Forward of the filler cap on the wing surface:
"Fuel 100LL/100 minimum grade aviation gasoline. Capacity 46 U.S. gal.
Capacity 34.5 U.S. gal. To bottom of filler collar."

(7) Forward of each fuel filler cap:
"Fuel cap fwd - arrow alignment, cap must not rotate during closing."

(8) Adjacent to overvoltage light:
"Low Voltage"

I. Applicable to Models R182 and TR182, S/N R18201929 thru R18202041:
All placards required in the POH/AFM must be installed in the appropriate locations.

J. Applicable to Models 182R and T182, S/N 18268294 thru 18268586:
All placards required in the POH/AFM must be installed in the appropriate locations.

NOTE 3. The cylinder head thermistors must be installed as follows:

Model	Engine and Cylinder Head Number				
	O-470-R	O-470-S	O-470-U	O-540-J	O-540-L
182N (1970 and 1971 Model)	3	3	N/A	N/A	N/A
182P (1972 and 1973 Model)	2	3	N/A	N/A	N/A
182P (1974 Model) 1	3	N/A	N/A	N/A	
182P (1975 and 1976 Model)	N/A	3	N/A	N/A	N/A
182Q (1977 through 1980 Model)	N/A	N/A	3	N/A	N/A
182R (1981 Model through 18268160)	N/A	N/A	5	N/A	N/A
182R (18268161 through 18268586)	N/A	N/A	3	N/A	N/A
T182 (1981 Model through 1985 Model)	N/A	N/A	N/A	N/A	1
R182 (1978 and 1979 Model)	N/A	N/A	N/A	5	N/A
R182 (1980 Model through 1986 Model)	N/A	N/A	N/A	4	N/A
TR182 (1979 Model)	N/A	N/A	N/A	N/A	3
TR182 (1980 Model through 1986 Model)	N/A	N/A	N/A	N/A	5

NOTE 4. The installation of the 0-470-S engine in Model 182N, 182P (1970 through 1974) will require a change of the oil temperature gauge. Reference Cessna Service Letter SE75-2 for information and instructions for this change.

NOTE 5. The marking of the airspeed indicator with IAS provides an equivalent level of safety to CAR 3.757 when the approved airspeed calibration data presented in Section V of the POH listed below is available to the pilot:

182P	Cessna P/N D1062-13	S/N 18264296 thru 18265175
182Q	Cessna P/N D1087-13	S/N 18265176 thru 18265965
182Q	Cessna P/N D1114-13	S/N 18263479, 18265966 thru 18266590
182Q	Cessna P/N D1141-13PH	S/N 18266591 thru 18267300
182Q	Cessna P/N D1176-13PH	S/N 18267301 thru 18267715

Data Pertinent to Model Items I through XII (cont'd)
NOTE 5. (cont'd)

182R	Cessna P/N D1196-13PH	S/N 18267716 thru 18268055
182R	Cessna P/N D1215-13PH	S/N 18268056 thru 18268293
182R	Cessna P/N D1233-13PH	S/N 18268294 thru 18268368
182R	Cessna P/N D1254-13PH	S/N 18268369 thru 18268434
T182	Cessna P/N D1197-13PH	S/N 18267302, 18267716 thru 18268055
T182	Cessna P/N D1216-13PH	S/N 18268056 thru 18268293
T182	Cessna P/N D1234-13PH	S/N 18268294 thru 18268368
T182	Cessna P/N D1234R1-13PH	(Special) S/N 18268365
T182	Cessna P/N D1255-13PH	S/N 18268369 thru 18268434
R182	Cessna P/N D1115-13	S/N R18200002 thru R18200583
R182	Cessna P/N D1142-13PH	S/N R18200584 thru R18201313
R182	Cessna P/N D1177-13PH	S/N R18201314 thru R18201628
R182	Cessna P/N D1198-13PH	S/N R18201629 thru R18201798
R182	Cessna P/N D1217-13PH	S/N R18201799 thru R18201928
R182	Cessna P/N D1235-13PH	S/N R18201929 thru R18201973
R182	Cessna P/N D1256-13PH	S/N R18201974 thru R18201999
R182	Cessna P/N D1277-13PH	S/N R18202000 thru R18202031
R182	Cessna P/N D1299-13PH	S/N R18202032 thru R18202041
TR182	Cessna P/N D1143-13PH	S/N R18200001, R18200584 thru R18201313 except R18200975
TR182	Cessna P/N D1143-2-13PH	(Special) S/N R18200975
TR182	Cessna P/N D1178-13PH	S/N R18201314 thru R18201628 except R18201315
TR182	Cessna P/N D1199-13PH	S/N R18201629 thru R18201798
TR182	Cessna P/N D1218-13PH	S/N R18201799 thru R18201928
TR182	Cessna P/N D1236-13PH	S/N R18201929 thru R18201973
TR182	Cessna P/N D1257-13PH	S/N R18201974 thru R18201999
TR182	Cessna P/N D1278-13PH	S/N R18201315, R18202000 thru R18202031
TR182	Cessna P/N D1300-13PH	S/N R18202032 thru R18202041

NOTE 6. 14-volt electrical system
(182 series through S/N 18265965 except 18263479)
28-volt electrical system
(182 series S/N 18263479, 18265966 thru 18268586)
(R182 and TR182 series S/N R18200001 thru R18202041)

NOTE 7. Special Ferry Flight Authorization. Flight Standards District Offices are authorized to issue Special overweight ferry flight authorizations. These airplanes are structurally satisfactory for ferry flight if maintained within the following limits:

1. Takeoff weight must not exceed 130% of the maximum weight for Normal Category.
2. The Never Exceed Airspeed (VNE) and Maximum Structural Cruising Speed (VC) must be reduced by 30%.
3. Forward and aft center of gravity limits may not be exceeded.
4. Structural load factors of +2.5 g. to -1.0 g. may not be exceeded.

Requirements for any additional engine oil should be established in accordance with AC 23.1011-1. Increased stall speeds and reduced climb performance should be expected for the increased weights. Flight characteristics and performance at the increased weights have not been evaluated. Procedures for issuing a Flight Permit for operations of overweight aircraft may be found in AC 21-4B.

In addition to the above specified placards, the prescribed operating limitations indicated by an asterisk (*) under Sections I through XII must also be displayed by permanent markings.

XIII. Model 182S, Skylane, 4 PCLM (Normal Category), Approved 03 October 1996.
Model 182T, Skylane, 4 PCLM (Normal Category), Approved 23 February 2001.

Engine	Lycoming IO-540-AB1A5. Rated 230 Horsepower
Fuel	100/100LL minimum grade aviation gasoline
Engine Limits	For all operations, 2400 RPM
Propeller	(1) McCauley Constant Speed (182S) (a) Hub B2D34C235/90DKB-8 (2 blades) Diameter: not over 82 in., not under 80.5 in. Pitch settings at 30 in. sta.: Low 17.0º, High 31.8º (b) McCauley Spinner: D-7267-2 (c) McCauley Governor: DC290D1/T8 (2) McCauley Constant Speed (182S, 182T) (a) Hub B3D36C431/80VSA-1 (3 blades) Diameter: not over 79 in., not under 77.5 in. Pitch settings at 30 in. sta.: Low 14.9º, High 31.7º (b) McCauley Spinner: D-7261-2 (c) McCauley Governor: DC290D1/T8
Propeller Limits	Static RPM at full throttle: Not over 2400; Not under 2300.

Airspeed Limits (182S)			
	Maneuvering	110 Knots IAS	(108 Knots CAS)
	Max Structural Cruising	140 Knots IAS	(138 Knots CAS)
	Never Exceed	175 Knots IAS	(170 Knots CAS)
	Flaps Extended	100 Knots IAS	(99 Knots CAS)
Airspeed Limits (182T)			
	Maneuvering	110 Knots IAS	(108 Knots CAS)
	Max Structural Cruising	140 Knots IAS	(136 Knots CAS)
	Never Exceed	175 Knots IAS	(171 Knots CAS)
	Flaps Extended	100 Knots IAS	(99 Knots CAS)

C.G. Range (182S)	Normal Category	
	(1) Aft Limits:	46.0 inches aft of datum at 3100 lbs. or less.
	(2) Forward Limits:	Linear variation from 40.9 inches aft of datum at 3100 pounds to 33.0 inches aft of datum at 2250 lbs.; 33.0 inches aft of datum at 2250 lbs. or less.
C.G. Range (182T)	Normal Category	
	(1) Aft Limits	46.0 inches aft of datum at 3,100 pounds or less.
	(2) Forward Limits	Linear variation from 40.9 inches aft of datum at 3,100 pounds, to 35.5 inches aft of datum at 2,700 pounds, to 33.0 inches aft of datum at 2,250 pounds; 33.0 inches aft of datum at 2,250 pounds or less.
Empty Wt. C.G. Range	None	
Reference Datum	Lower portion of front face of firewall	
MAC	58.8 inches; Leading edge of MAC 25.98 inches aft of datum	
Leveling Means	Left side of Tailcone at 139.65 inches and 171.65 inches aft of datum.	

Maximum Weights (see Note 5)	Normal Category	
	Maximum Ramp	3,110 pounds
	Maximum Takeoff	3,100 pounds
	Maximum Landing	2,950 pounds

XIII. Model 182S, Model 182T (cont'd)

No. of Seats	4 (2 at 32.0 to 50.0 inches aft of datum; 2 at 74.0 inches aft of datum)
Maximum Baggage	120 pounds at 82.0 to 109.0 inches aft of datum 80 pounds at 109.0 to 134.0 inches aft of datum (Max. combined weight capacity for baggage areas is 200 pounds)
Fuel Capacity (Gal.)	182S: 92 gallons total: 88 gallons usable 182T: 92 gallons total; 87 gallons usable (Two 46 gallon tanks in wings at 46.5 inches aft of datum) (See NOTE 1 for data on usable fuel).
Oil Capacity (Gal.)	9.0 quarts at 14.8 inches forward of datum 5.0 quarts usable

Control Surface Movements

Wing flaps			Down	38° +0°, -1°
Elevator tab	Up	24° ± 2°	Down	15° ± 1°
Ailerons	Up	20° ± 2°	Down	15° ± 2°
Elevator (Relative to stabilizer)	Up	28° ± 1°	Down	21° ± 1°

Rudder: Right: 24° +0°, -1° Left: 24° +0°,-1°
(Parallel to 0.00 W.L.)
Right: 27°13' +0°, -1° Left: 27°13' +0°,-1°
(Perpendicular to hinge line)

Serial Numbers Eligible	182S: 18280001 thru 18280944 182T: 18280945 and On

Data Pertinent to Models 182S and 182T

Certification Basis:

FAR Part 23, effective February 1, 1965, as amended by 23-1 through 23-6, except as follows:
FAR 23.423; 23.611; 23.619; 23.623; 23.689; 23.775; 23.871; 23.1323; and 23.1563 as amended by Amendment 23-7. 23.807 and 23.1524 as amended by Amendment 23-10. 23.507; 23.771; 23.853(a)(b)(c); and 23.1365 as amended by Amendment 23-14. 23.951 as amended by Amendment 23-15. 23.607; 23.675; 23.685; 23.733; 23.787; 23.1309 and 23.1322 as amended by Amendment 23-17. 23.1301 as amended by Amendment 23-20. 23.1353; and 23.1559 as amended by Amendment 23-21. FAR 23.603; 23.605; 23.613; 23.1329 and 23.1545 as amended by Amendment 23-23. 23.441 and 23.1549 as amended by Amendment 23-28. 23.779 and 23.781 as amended by Amendment 23-33. 23.1; 23.51 and 23.561 as amended by Amendment 23-34. 23.301; 23.331; 23.351; 23.427; 23.677; 23.701; 23.735; and 23.831 as amended by Amendment 23-42. 23.961; 23.1093; 23.1143(g); 23.1147(b); 23.1303; 23.1357; 23.1361 and 23.1385 as amended by Amendment 23-43. 23.562(a)(b)(2)(c)(1)(2)(3)(4) as amended by Amendment 23-44. 23.33; 23.53; 23.305; 23.321; 23.485; 23.621; 23.655 and 23.731 as amended by Amendment 23-45.

FAR 36, dated December 1, 1969, as amended by Amendments 36-1 through 36-21.

Equivalent Safety Items, Model 182S:

1. Introduction System Icing Protection	23.1093.
2. Throttle Control	23.1143(g)
3. Mixture Control	23.1147(b)

Date of Application for Amended Type Certificate was January 22, 1996.
Type Certificate No. 3A13 was amended October 3, 1996.

Data Pertinent to Model 182S and 182T (cont'd)

Equivalent Safety Items, 182T:

1. Induction System Icing Protection	23.1093, refer to FAA letter dated 12/19/2000
2. Throttle Control	23.1143(g), refer to FAA letter dated 12/19/2000
3. Mixture Control	23.1147(b), refer to FAA letter dated 12/19/2000
4. Anti-collision Lights	23.1401(d), refer to FAA letter dated 2/20/2001
5. Aviation White Color Requirement	23.1397(c), refer to ACE-07-11, FAA letter dated 11/29/2007

Additions for the Garmin G1000 Integrated Cockpit System (ICS) Only:
14 CFR 23.303, 23.307, 23.601, 23.1163(a)(1)(2), 23.1367, 23.1381 as amended by Amendment 23-N/C. 23.1589 as amended by Amendment 23-13. 23.771(a) as amended by Amendment 23-14. 23.607 and (Electrical System) 23.1309(a)(1)(2)(c) as amended by Amendment 23-17. 23.1301, 23.1327, 23.1547(e) as amended by Amendment 23-20. 23.1501, 23.1541(a)(1)(2)(b)(1)(2) as amended by Amendment 23-21. 23.603, 23.605 as amended by Amendment 23-23. 23.1529 as amended by Amendment 23-26. 23.561(e), 23.1523; 23.1581(a)(2), 23.1583(a)(1)(2)(b)(h), 23.1585(a)(b)(d) as amended by Amendment 23-34. 23.301 as amended by Amendment 23-42. 23.1322, 23.1331, 23.1357(a)(b)(c)(d) as amended by Amendment 23-43. 23.305, 23.773(a)(1)(2), 23.1525, 23.1549 as amended by Amendment 23-45. 23.1303(a)(b)(c)(f), 23.1309(a)(1)(i)(ii)(2)(b)(1)(2)(i)(ii)(3)(4)(i)(ii)(iii)(iv)(c)(1)(2)(iii)(3)(d)(e)(f)(1), 23.1311, 23.1321(a)(c)(d)(e), 23.1323(a)(b)(1)(2)(c); 23.1329(g)(h), 23.1351(a)(1)(2)(i)(b)(1)(iii)(2)(3)(c)(4)(d)(1), 23.1353(a)(b)(c)(d)(e), 23.1359(c), 23.1361, 23.1365(a)(b)(d)(e)(f), 23.1431(a)(b)(d)(e) as amended by Amendment 23-49. 23.1325(a)(b)(1)(2)(i)(3)(c)(d)(e), 23.1543(b)(c), 23.1545(a)(b)(1)(2)(3)(4), 23.1553, 23.1555(a)(b), 23.1563(a), 23.1567(a) as amended by Amendment 23-50. 23.777(a)(b), 23.955(a)(2), 23.1337(a)(1)(2)(b)(1)(c) as amended by Amendment 23-51. 23.1305(a)(1)(2)(3)(b)(2)(3)(i)(4)(i)(5)(6)(i) as amended by Amendment 23-52. 23.901(a)(b) as amended by Amendment 23-53.

Additions for the Garmin GFC-700 Automatic Flight Control System (AFCS) Only:
14 CFR 23.1335 as amended by Amendment 23-20. 23.1329 (a)(c)(d)(e)(f) as amended by Amendment 23-49.

Additions for the Garmin GI 275 Electronic Flight Instrument Only:
14 CFR 23.1327 as amended by Amendment 23-20. 23.1501 as amended by Amendment 23-21. 23.1529 as amended by Amendment 23-26. 23.1523(b), 23.1581(a)(1)(2) as amended by Amendment 23-34. 23.1322, 23.1331 as amended by Amendment 23-43. 23.1525 as amended by Amendment 23-45. 23.1303(a)(b)(f), 23.1309(a)(1)(2)(b)(c)(1)(2)(iii)(3)(d)(e)(f), 23.1311(a)(b), 23.1321(a)(c)(d)(e), 23.1323(a)(c), 23.1351(a)(1)(2)(i), 23.1359(c), 23.1365(a)(d)(e), 23.1431(a)(b) as amended by Amendment 23-49. 23.1325(a)(b)(1)(i)(ii)(iii)(2)(i), 23.1543(b)(c), 23.1545(a)(b)(1)(2)(3)(4), 23.1555(a)(b) as amended by Amendment 23-50. 23.777(a)(b) as amended by Amendment 23-51. 23.1308(a)(b)(c) as amended by Amendment 23-57; 23.1306(a)(b) as amended by Amendment 23-61. 23.2010, 23.2510 as amended by Amendment 23-64 (see NOTE 8). Effective at S/N 18283135 and On.

Additions for 14 CFR Part 23 Amendment 64:
14 CFR 23.2000, 23.2005, 23.2010 as amended by Amendment 23-64 (see NOTE 9). Effective at S/N 18283135 and On.

Additions for the Garmin G1000 NXi Integrated Cockpit System Only:
14 CFR 23.683, 23.867 as amended by Amendment 23-7. 23.613(a)(b) as amended by Amendment 23-45. 23.301(a), 23.337, 23.561(a)(b)(3)(e), 23.607(a)(b), 23.611 as amended by Amendment 23-48. 23.1329(b), 23.1351(g), 23.1353(h), 23.1359(a)(b), as amended by Amendment 23-49. 23.143(c), 23.1555(e)(2), 23.1581(a)(b)(1)(3)(c)(d)(f), 23.1583(h)(m), 23.1585(j) as amended by Amendment 23-50. 23.1337(b)(c) as amended by Amendment 23-51. 23.2515, 23.2520 as amended by Amendment 23-64 (see NOTE 10). Effective at S/N 18283280 and On.

Additions for the Dual Electronic Ignition System Only:
14 CFR 23.1309(a)(1)(2), 23.1353(h), 23.1431(b) as amended by Amendment 23-49. 23.1581(a)(2) as amended by Amendment 23-50. 23.901(a)(1)(2)(b)(1)(2) as amended by Amendment 23-53. Effective at S/N 18283359 and On.

Additions for new or changed non-rechargeable lithium battery installations:
14 CFR Part 23 regulations as amended by Amendment 23-64 (see NOTE 11): 14 CFR 23.2010 and 23.2325(g). Effective at S/N 18283359 and On.

Additions for new or changed installations affecting aeronautical security (cybersecurity):
14 CFR Part 23 regulations as amended by Amendment 23-64 (see NOTE 12): 14 CFR 23.2010, 23.2500, 23.2505 and 23.2510. Effective at S/N 18283359 and On.

Special Conditions as follows:
No. 23-146-SC, "Cessna Aircraft Company, Cessna Model 182T/T182T Airplane: Installation of Electronic Flight Instrument System and the Protection of the System from High Intensity Radiated Fields (HIRF).

Data Pertinent to Model 182S and 182T (cont'd)

Production Basis (Model 182S)

Production Certificate No. PC-4 issued June 30, 1997. Applies to S/N 18280013, 18280016, 18280017, 18280019 and On. Airplane serial numbers not listed were produced under Type Certificate only. Cessna is authorized to issue airworthiness certificates under the delegation provisions of Delegation Option Authorization No. CE-1 in accordance with Part 21 of the FAR.

Production Basis (Model 182T)

Production Certificate No. 4 issued March 8, 2001. Applies to S/N 18280945 and On. Cessna is authorized to issue airworthiness certificates under the delegation provisions of Delegation Option Authorization No. DOA-100129-CE in accordance with Part 21 of the FAR.

Equipment

The basic required equipment as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the airplane for certification.

NOTE 1. Weight and Balance:

Model 182S, S/N 18280001 thru 18280944
The certificated basic empty weight and corresponding center of gravity location must include unusable fuel of 24 lbs. at 48 inches aft of datum, and full oil of 16.2 lb. at 14.8 inches forward of datum.

Model 182T, S/N 18280945 and On
The certificated basic empty weight and corresponding center of gravity location must include unusable fuel of 30 lbs. at 48 inches aft of datum, and full oil of 16.2 lb. at 14.8 inches forward of datum.

NOTE 2. FAA Approved AFM: P/N 182SPHUS00, or later FAA approved revision, are applicable to Model 182S. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 070810, Airplane Assembly. A useful placarding reference is the Textron Aviation Illustrated Parts Catalogue (IPC). Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

FAA Approved AFM: P/N 182TPHUS00, or later FAA approved revision, is applicable to the Model 182T. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 070810, Airplane Assembly. A useful placarding reference is the Textron Aviation IPC. Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

FAA Approved AFM: P/N 182TPHAUS-00, or later FAA approved revision, are applicable to the Model 182T equipped with Garmin G1000 Integrated Cockpit System. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 070810, Airplane Assembly. A useful placarding reference is the Textron Aviation IPC. Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

FAA Approved AFM: P/N 182TPHBUS-00, or later FAA approved revision, are applicable to the Model 182T equipped with Garmin G1000 Integrated Cockpit System and Garmin GFC-700 AFCS. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 070810, Airplane Assembly. A useful placarding reference is the Textron Aviation IPC. Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

Data Pertinent to Model 182S and 182T (cont'd)
NOTE 2. (cont'd)

FAA Approved AFM, P/N 182TPHCUS-00, or later FAA approved revision, are applicable to the Model T182T equipped with Garmin G1000 NXi Integrated Cockpit System. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 070810, Airplane Assembly. A useful placarding reference is the Textron Aviation IPC. Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

NOTE 3. The CHT probe must be installed on Head No. 1 (182S) or No. 3 (182T).

NOTE 4. Special Ferry Flight Authorization. Flight Standards District Offices are authorized to issue Special overweight ferry flight authorizations. These airplanes are structurally satisfactory for ferry flight if maintained within the following limits:
1. Takeoff weight must not exceed 130% of the maximum weight for Normal Category.
2. The Never Exceed Airspeed (VNE) and Maximum Structural Cruising Speed (VC) must be reduced by 30%.
3. Forward and aft center of gravity limits may not be exceeded.
4. Structural load factors of +2.5 g. to -1.0 g. may not be exceeded.

Requirements for any additional engine oil should be established in accordance with AC 23.1011-1. Increased stall speeds and reduced climb performance should be expected for the increased weights. Flight characteristics and performance at the increased weights have not been evaluated. Procedures for issuing a Flight Permit for operations of overweight aircraft may be found in AC 21-4B.

NOTE 5. Model 182S S/N 18280617 thru 18280670 may differ structurally and are, therefore, not eligible for any weight increases above the approved maximum takeoff weight limit of 3,100 pounds. Any exceptions must first be coordinated with the Wichita ACO Branch. Exceptions to this limitation have been inspected and found to comply with type data for Model 182S and includes the following S/N: 18280620.

NOTE 6. The following serials numbers are manufactured under the name Cessna Aircraft Company:
Model 182S: 18280001 thru 18280944; Model 182T: 18280945 thru 18282368.

NOTE 7. Company name change effective 7/29/15. The following serials numbers are manufactured under the name Textron Aviation Inc.: Model 182T: 18282369 and On.

NOTE 8. Means of Compliance for 23.2510 [23-64] is draft AC 20-184A.

NOTE 9. 14 CFR 23.2005 Certification Level and Performance Level: Level 2, Low Speed

NOTE 10. Means of Compliance for 23.2515 [23-64] and 23.2520 [23-64] is FAA Policy PS-ACE-23-10

NOTE 11. Means of Compliance for 23.2010 [23-64] and 23.2325(g) [23-64] is CPP-23.2525-1 (signed 2023.10.19).

NOTE 12. Means of Compliance for 23.2010, 23.2500, 23.2505, and 23.2510 as amended by Amendment 23-64 is ASTM F3532-22.

XIV. Model T182T, Skylane, 4 PCLM (Normal Category), Approved 23 February 2001

Engine	Lycoming TIO-540-AK1A. Rated 235 Horsepower
Fuel	100/100LL minimum grade aviation gasoline
Engine Limits	For all operations, 2,400 RPM
Propeller	McCauley Constant Speed (a) Hub B3D36C442/80VSB-1 Diameter: not over 79 inches; not under 77.5 inches Pitch settings at 30 in. sta.: Low 15.1º, High 35.4º Note: A low pitch setting of 15.3º is also acceptable for this model number prop. (b) McCauley Spinner: D-7261-2 (c) McCauley Governor: DC290D1/T8

XIV. Model T182T (cont'd)

Propeller Limits	Static RPM at full throttle: Not over 2400; Not Under 2300			
Airspeed Limits	Maneuvering	110 Knots IAS	(110 Knots CAS)	
	Max Structural Cruising	140 Knots IAS	(137 Knots CAS)	
	Never Exceed	175 Knots IAS	(170 Knots CAS)	
	Flaps Extended	100 Knots IAS	(100 Knots CAS)	
C.G. Range	Normal Category			
	(1) Aft Limits	46.0 inches aft of datum at 3,100 pounds or less.		
	(2) Forward Limits	Linear variation from 40.9 inches aft of datum at 3,100 pounds, to 35.5 inches aft of datum at 2,700 pounds, to 33.0 inches aft of datum at 2,250 pounds; 33.0 inches aft of datum at 2,250 pounds or less.		
Empty Wt. C.G. Range	None			
Reference Datum	Lower portion of front face of firewall			
MAC	58.8 inches; Leading edge of MAC 25.98 inches aft of datum			
Leveling Means	Left side of Tailcone at 139.65 inches and 171.65 inches aft of datum			
Maximum Weights	Normal Category			
	Maximum Ramp	3,110 pounds		
	Maximum Takeoff	3,100 pounds		
	Maximum Landing	2,950 pounds		
No. of Seats	4 (2 at 32.0 to 50.0 inches aft of datum; 2 at 74.0 inches aft of datum)			
Maximum Baggage	120 pounds at 82.0 to 109.0 inches aft of datum 80 pounds at 109.0 to 134.0 inches aft of datum (Max. combined weight capacity for baggage areas is 200 pounds)			
Fuel Capacity (Gal.)	92 gallons total; 87 gallons usable (Two 46 gallon tanks in wings at 46.5 inches aft of datum) See NOTE 1 for data on usable fuel.			
Oil Capacity (Qts.)	9.0 quarts at 14.8 inches forward of datum 5.0 quarts usable			
Control Surface Movements	Wing flaps		Down	38° +0°, -1°
	Elevator tab	Up 24° ± 2°	Down	15° ± 1°
	Ailerons	Up 20° ± 2°	Down	15° ± 2°
	Elevator	Up 28° ± 1°	Down	21° ± 1°
	(Relative to stabilizer)			
	Rudder:	Right: 24° +0°, -1° (Parallel to 0.00 W.L.)	Left:	24° +0°, -1°
		Right: 27°13' +0°, -1° (Perpendicular to hinge line)	Left:	27°13' +0°, -1°
Serial Numbers Eligible	T18208001 and On			

Data Pertinent to Model T182T

Certification Basis

FAR Part 23, effective February 1, 1965, as amended by 23-1 through 23-6, except as follows:
14 CFR 23.423, 23.611, 23.619, 23.623, 23.689, 23.775, 23.871, 23.1323, 23.1563 as amended by Amendment 23-7. 23.807, 23.1524 as amended by Amendment 23-10. 23.507, 23.771, 23.853(a)(b)(c), 23.1365 as amended by Amendment 23-14. 23.951 as amended by Amendment 23-15. 23.607, 23.675, 23.685, 23.733, 23.787, 23.1309, 23.1322 as amended by Amendment 23-17. 23.1301 as amended by Amendment 23-20. 23.1353, 23.1559 as amended by Amendment 23-21. 23.603, 23.605, 23.613, 23.1329, 23.1545 as amended by Amendment 23-23. 23.441, 23.1549 as amended by Amendment 23-28. 23.779, 23.781 as amended by Amendment 23-33. 23.1, 23.51, 23.561 as amended by Amendment 23-34. 23.301, 23.331, 23.351, 23.427, 23.677, 23.701, 23.735, 23.831 as amended by Amendment 23-42. 23.961, 23.1093, 23.1143(g), 23.1147(b), 23.1303, 23.1357, 23.1361, 23.1385 as amended by Amendment 23-43. 23.562(a)(b)(2)(c)(1)(c)(2)(3)(4) as amended by Amendment 23-44. 23.33, 23.53, 23.305, 23.321, 23.485, 23.621, 23.655, 23.731 as amended by Amendment 23-45.

FAR Part 36, dated December 1, 1969, as amended by Amendments 36-1 through 36-22.

Equivalent Level of Safety Items:

1. Throttle Control	23.1143(g), refer to FAA letter dated 12/19/2000
2. Mixture Control	23.1147(b), refer to FAA letter dated 12/19/2000
3. Anti-collision Lights	23.1401(d), refer to FAA letter dated 02/20/2001
4. Aviation White Color Requirement	23.1397(c), refer to ACE-07-11, FAA letter dated 11/29/2007

Additions for the Garmin G1000 Integrated Cockpit System (ICS) Only:
14 CFR 23.303, 23.307, 23.601, 23.1163(a)(1)(2), 23.1367, 23.1381 as amended by Amendment 23-N/C. 23.1589 as amended by Amendment 23-13. 23.771(a) as amended by Amendment 23-14. 23.607 and (Electrical System) 23.1309(a)(1)(2)(c) as amended by Amendment 23-17. 23.1301, 23.1327, 23.1547(e) as amended by Amendment 23-20. 23.1501, 23.1541(a)(1)(2)(b)(1)(2) as amended by Amendment 23-21. 23.603, 23.605 as amended by Amendment 23-23. 23.1529 as amended by Amendment 23-26. 23.561(e), 23.1523, 23.1581(a)(2), 23.1583(a)(1)(2)(b)(h), 23.1585(a)(b)(d) as amended by Amendment 23-34. 23.301 as amended by Amendment 23-42. 23.1322, 23.1331, 23.1357(a)(b)(c)(d) as amended by Amendment 23-43. 23.305, 23.773(a)(1)(2), 23.1525, 23.1549 as amended by Amendment 23-45. 23.1303(a)(b)(c)(f), 23.1309(a)(1)(i)(ii)(2)(b)(1)(2)(i)(ii)(3)(4)(i)(ii)(iii)(iv)(c)(1)(2)(iii)(3)(d)(e)(f)(1), 23.1311, 23.1321(a)(c)(d)(e), 23.1323(a)(b)(1)(2)(c), 23.1329(g)(h), 23.1351(a)(1)(2)(i)(b)(1)(iii)(2)(3)(c)(4)(d)(1), 23.1353(a)(b)(c)(d)(e), 23.1359(c), 23.1361, 23.1365(a)(b)(d)(e)(f) and 23.1431(a)(b)(d)(e) as amended by Amendment 23-49. 23.1325(a)(b)(1)(2)(i)(3)(c)(d)(e); 23.1543(b)(c), 23.1545(a)(b)(1)(2)(3)(4), 23.1553, 23.1555(a)(b), 23.1563(a), 23.1567(a) as amended by Amendment 23-50. 23.777(a)(b), 23.955(a)(2), 23.1337(a)(1)(2)(b)(1)(c) as amended by Amendment 23-51. 23.1305(a)(1)(2)(3)(b)(2)(3)(i)(4)(i)(5)(6)(i) as amended by Amendment 23-52. 23.901(a)(b) as amended by Amendment 23-53.

Additions for the Garmin GFC-700 Automatic Flight Control System (AFCS) Only:
14 CFR 23.1335 as amended by Amendment 23-20. 23.1329(a)(c)(d)(e)(f) as amended by Amendment 23-49.

Additions for the Garmin GI 275 Electronic Instrument Only:
14 CFR 23.1327 as amended by Amendment 23-20. 23.1501 as amended by Amendment 23-21. 23.1529 as amended by Amendment 23-26. 23.1523(b), 23.1581(a)(1)(2) as amended by Amendment 23-34. 23.1322, 23.1331 as amended by Amendment 23-43. 23.1525 as amended by Amendment 23-45. 23.1303(a)(b)(f); 23.1309(a)(1)(2)(b)(c)(1)(2)(iii)(3)(d)(e)(f), 23.1311(a)(b), 23.1321(a)(c)(d)(e), 23.1323(a)(c), 23.1351(a)(1)(2)(i), 23.1359(c), 23.1365(a)(d)(e), 23.1431(a)(b) as amended by Amendment 23-49. 14 CFR 23.1325(b)(1)(i)(ii)(iii)(2)(i), 23.1543(b)(c), 23.1545(a)(b)(1)(2)(3)(4), 23.1555(a)(b) as amended by Amendment 23-50. 23.777(a)(b) as amended by Amendment 23-51. 23.1308(a)(b)(c) as amended by Amendment 23-57. 23.1306(a)(b) as amended by Amendment 23-61. 23.2010, 23.2325(g)(1)(2), 23.2510 as amended by Amendment 23-64 (see NOTE 7). Effective at S/N T18209101 and On.

Additions for 14 CFR Part 23 Amendment 64:
14 CFR 23.2000, 23.2005, 23.2010 as amended by Amendment 23-64 (see NOTE 8). Effective at S/N T18209101 and On.

Data Pertinent to Model T182T (cont'd)
Certification Basis (cont'd)

Additions for the Garmin G1000 NXi Integrated Cockpit System Only:
14 CFR 23.683; 23.867 as amended by Amendment 23-7; 23.613(a)(b) as amended by Amendment 23-45; 23.301(a); 23.337; 23.561(a)(b)(3)(e); 23.607(a)(b); 23.611 as amended by Amendment 23-48; 23.1329(b); 23.1351(g); 23.1353(h); 23.1359(a)(b) as amended by Amendment 23-49; 23.143(c); 23.1555(e)(2); 23.1581(a)(b)(1)(3)(c)(d)(f); 23.1583(h)(m); 23.1585(j) as amended by Amendment 23-50; 23.1337(b)(c) as amended by Amendment 23-51; 23.2515; 23.2520 as amended by Amendment 23-64 (see NOTE 9). Effective at S/N T18209159 and On.

Additions for the Dual Electronic Ignition System Only:
14 CFR 23.1309(a)(1)(2), 23.1353(h), 23.1431(b) as amended by Amendment 23-49. 23.1581(a)(2) as amended by Amendment 23-50. 23.901(a)(1)(2)(b)(1)(2) as amended by Amendment 23-53. Effective for S/N's T18208204, T18208714, and T18209210 and On.

Additions for new or changed non-rechargeable lithium battery installations:
14 CFR Part 23 regulations as amended by Amendment 23-64 (see NOTE 10): 14 CFR 23.2010 and 23.2325(g). Effective for S/N's T18208204, T18208714, and T18209210 and On.

Additions for new or changed installations affecting aeronautical security (cybersecurity):
14 CFR Part 23 regulations as amended by Amendment 23-64 (see NOTE 11): 14 CFR 23.2010, 23.2500, 23.2505 and 23.2510. Effective for S/N's T18208204, T18208714, and T18209210 and On.

Special Conditions as follows:

No. 23-146-SC, "Cessna Aircraft Company, Cessna Model 182T and T182T: Installation of Electronic Flight Instrument System and the Protection of the System from High Intensity Radiated Fields (HIRF).

Production Basis (Model T182T)
Production Certificate No. 4 issued March 8, 2001. Applies to airplane S/N T18208001 and On. Cessna is authorized to issue airworthiness certificates under the delegation provisions of Delegation Option Authorization No. DOA-100129-CE in accordance with Part 21 of the FAR.

Equipment
The basic required equipment as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the airplane for certification.

NOTE 1. Weight and Balance:

S/N T18208001 and On (Model T182T)
The certificated empty weight and corresponding center of gravity location must include unusable fuel of 30 lb. at 48 inches aft of datum, and full oil of 16.2 lb. at 14.8 inches forward of datum.

NOTE 2. The POH/AFM, P/N T182TPHUS00, or later FAA approved revision, is applicable to Model T182T. The airplane must be operated according to the appropriate POH/AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 070811, Airplane Assembly. A useful placarding reference is the Textron Aviation IPC. Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

FAA Approved AFM, P/N T182TPHAUS-00, or later FAA approved revision, are applicable to the Model 182T equipped with Garmin G1000 Integrated Cockpit System. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 070811, Airplane Assembly. A useful placarding reference is the Textron Aviation IPC. Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

FAA Approved AFM, P/N T182TPHBUS-00, or later FAA approved revision, are applicable to the Model T182T equipped with Garmin G1000 Integrated Cockpit System and Garmin GFC-700 AFCS. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 070811, Airplane Assembly. A useful placarding reference is the Textron Aviation IPC. Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

Data Pertinent to Model T182T (cont'd)
NOTE 2. (cont'd)

FAA Approved AFM, P/N T182TPHCUS-00, or later FAA approved revision, are applicable to the Model T182T equipped with Garmin G1000 NXi Integrated Cockpit System. The airplane must be operated according to the appropriate AFM. Required placards are included in the AFM, the applicable operating rules, or the certification basis must be installed as specified via the parts list for 070811, Airplane Assembly. A useful placarding reference is the Textron Aviation IPC. Any discrepancies identified between the IPC and an aircraft under inspection need to be reconciled using the previously stated parts list.

NOTE 3. The CHT probe must be installed on Head No. 4.

NOTE 4. Special Ferry Flight Authorization. Flight Standards District Offices (FSDO) are authorized to issue Special overweight ferry flight authorizations. This airplane is structurally satisfactory for ferry flight if maintained within the following limits:

1. Takeoff weight must not exceed 130% of the maximum weight for Normal Category.
2. The Never Exceed Airspeed (VNE) and Maximum Structural Cruising Speed (VC) must be reduced by 30%.
3. Forward and aft center of gravity limits may not be exceeded.
4. Structural load factors of +2.5 g. to -1.0 g. may not be exceeded.

Requirements for any additional engine oil should be established in accordance with AC 23.1011-1. Increased stall speeds and reduced climb performance should be expected for the increased weights. Flight characteristics and performance at the increased weights have not been evaluated. Procedures for issuing a Flight Permit for operations of overweight aircraft may be found in AC 21-4B.

NOTE 5. The following serials numbers are manufactured under the name Cessna Aircraft Company: T18208001 thru T18209100.

NOTE 6. Company name change effective 7/29/15. The following serials numbers are manufactured under the name Textron Aviation Inc.: T18209101 and On.

NOTE 7. Means of Compliance for 23.2510 [23-64] is draft AC 20-184A.

NOTE 8. 14 CFR 23.2005 Certification Level and Performance Level: Level 2, Low Speed.

NOTE 9. Means of Compliance for 23.2515 [23-64] and 23.2520 [23-64] is FAA Policy PS-ACE-23-10.

NOTE 10. Means of Compliance for 23.2010 [23-64] and 23.2325(g) [23-64] is CPP-23.2525-1 (signed 2023.10.19).

NOTE 11. Means of Compliance for 23.2010, 23.2500, 23.2505, and 23.2510 as amended by Amendment 23-64 is ASTM F3532-22.

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